

Glass modeling in LS-DYNA

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Content

- Common approach to model laminated safety glass
- material model *MAT_GLASS (*MAT_280)
 - Theory
 - Keyword *MAT_280
- PVB-layer
- application

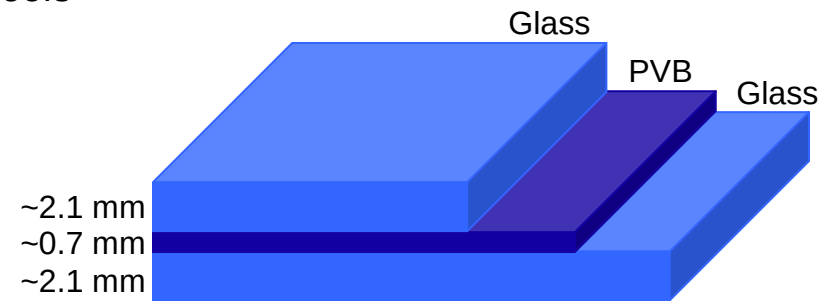
Laminated safety glass

■ Two types of safety glass commonly used:

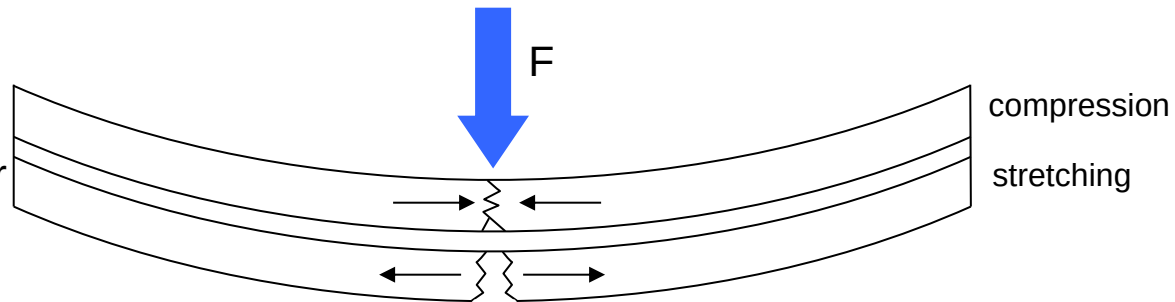
- Tempered glass → side windows
- Laminated glass → windshields, panorama sunroofs

■ Laminated glass consists of 3 layers

- 2x glass
- PVB interlayer



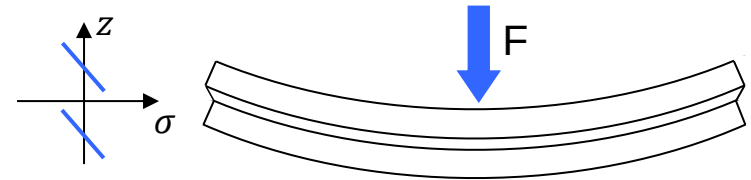
■ Glass fragments are bonded → Difficult mechanical behavior



Modelling of laminated safety glass

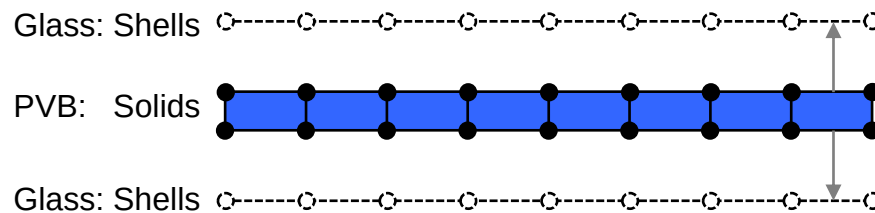
■ Discretization:

- PVB: transverse shear deformation important
→ **solid elements**
- Glass: **shell elements**



■ Contact between layers: **Shared nodes**

■ Offset the glass layers by **NLOC** parameter



*SECTION SHELL								
\$	SECID	ELFORM	SHRF	NIP	PROPT	QR/IRID	ICOMP	SETYP
\$	103	2	1.0	7	3.0	0.0		
\$	T1	T2	T3	T4	NLOC	MAREA		
	2.1	2.1	2.1	2.1	-1.00			

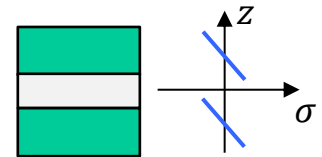
Modelling of laminated safety glass

■ Discretization PVB:

- solid elements
- 1 solid element over thickness
- elform 1 (under-integrated),
elform -1 or -2 (for poor aspect ratio) or elform 2 (fully integrated)
- hourglass stabilization HGID=6 and qh=0.5-0.8

■ Discretization glass layers

- shell elements
- e.g. elform = 2 (under-integrated) ... for *MAT_GLASS
- hourglass stabilization HGID=4
- sufficient number of integration points over thickness



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Material model *MAT_GLASS (*MAT_280)

- Smeared fixed crack material model
- Stress based **failure criteria with tension – compression asymmetry**
- **Treatment of failure without deleting elements**
- **Crack closure and opening effects**

- Further: Easy to use, low computation times, usable in crash simulations
- For shell and thick shells elements (tshell, elform 1,2,6) and explicit analysis

Detection of failure – failure criteria

- Failure criteria: stress based: Rankine, Mohr-Coulomb, Drucker-Prager

- Failure:

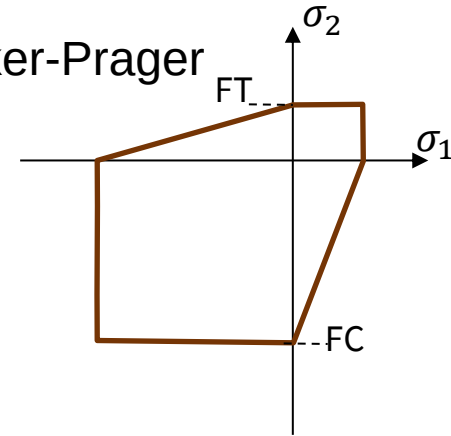
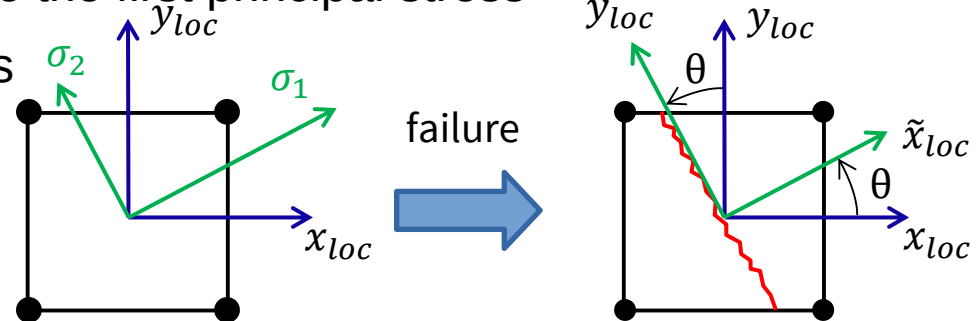
- Compressive failure: Material is 'crumbled'

- Tensile failure: Single crack occurs

- Tensile Failure:

- Crack direction perpendicular to the first principal stress

- Local crack coordinate systems



- Second crack can occur, orthogonal to the first crack

- Cracks can open and close independently

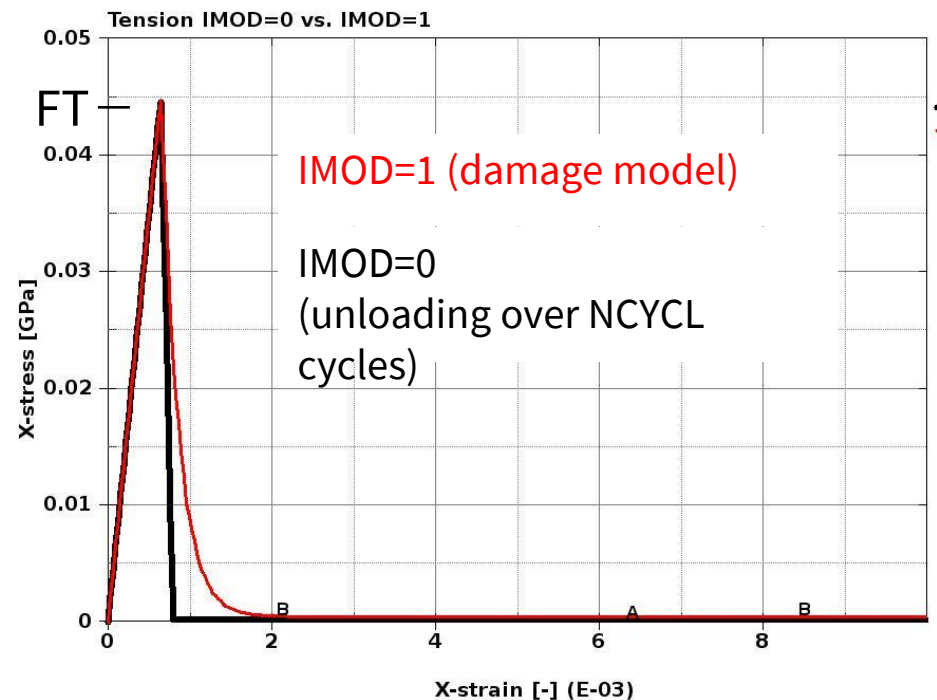
Softening

After a tensile crack is detected:

- Orthotropic material behavior in crack coordinate system
- Stress components which can no longer be carried are lowered in **NCYCR cycles** or with **damage model**

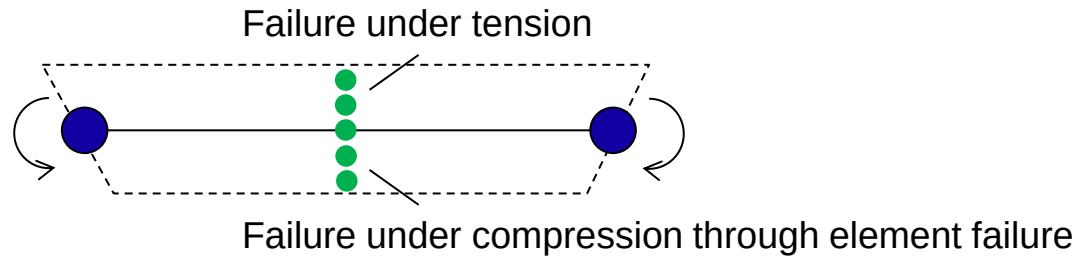
After compressive failure:

- All stress components are lowered in **NCYCR cycles** or with **damage model**



Element failure

- If a certain number of IP in one element failed, whole element will fail
- The crack direction in the IP which failed first sets the direction for the whole element
- Crack strain is computed individually in each IP



*MAT_GLASS

Card 1	1	2	3	4	5	6	7	8
Variable	MID	RO	E	PR			IMOD	ILAW
Type	A8	F	F	F			F	F

Card 2	1	2	3	4	5	6	7	8
Variable	FMOD	FT	FC	AT	BT	AC	BC	FTSCL
Type	F	F	F	F	F	F	F	F

Card 3	1	2	3	4	5	6	7	8
Variable	SFSTI	SFSTR	CRIN	ECRCL	NCYCR	NIPF		
Type	F	F	F	F	F	F		

Optional card.

Card 4	1	2	3	4	5	6	7	8
Variable	EPSCR	ENGCR	RADCR	RATENL	RFILTF			
Type	F	F	F					

parameter from R9.0
parameter from R10.0
parameter from R9.2
parameter from R11.0

*MAT_GLASS (Parameters from R9.0)

- FMOD: Failure model (0.: Rankine (default), 1.: Mohr-Coulomb, 2.: Drucker-Prager)
- FT: Tensile strength
- FC: Compressive strength
- SFSTI: Stiffness scale factor in case of failure
SFSTI = 0.1 means: stiffness is reduced to 10% of elastic stiffness at failure
- SFSTR: Stress scale factor in case of failure
- NCYCR: Number of cycles in which the stress is lowered to SFSTR * failure stress
- CRIN: Flag for crack strain initialization:
0.: initial crack strain is strain when crack occurs (default)
1.: initial crack strain is 0.0
- ECRCL: Crack strain from which on the crack is considered as completely closed
- NIPF: Number of failed integration points to fail all through thickness integration points

*MAT_GLASS (Parameters from R10.0)

stress reduction with damage model

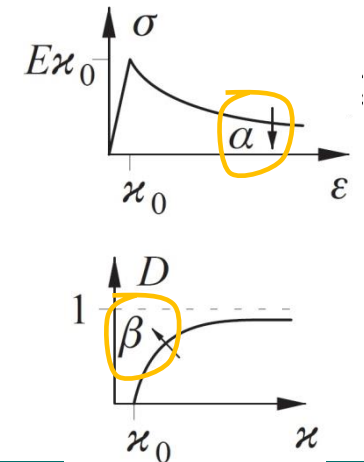
- IMOD: Flag to choose unloading procedure, when critical stress is reached:
EQ.0.0: Softening in NCYCR load steps (default),
EQ.1.0: Damage model for softening
- ILAW: EQ.0.0: same damage evolution for tensile and compressive failure (default):

$$D = 1 - \frac{\kappa^0}{\kappa} (1 - \alpha_{t,c} + \alpha_{t,c} e^{-\beta_{t,c} (\kappa - \kappa^0)})$$

EQ.1.0: damage evolution for tensile failure: $D = 1 - \frac{\kappa^0}{\kappa} (1 - \alpha_t + \alpha_t e^{-\beta_t (\kappa - \kappa^0)})$

damage evolution compressive failure: $D = 1 - \frac{\kappa^0}{\kappa} (1 - \alpha_c) - \alpha_c e^{-\beta_c (\kappa - \kappa^0)}$

- AT: tensile damage evolution parameter α_t . $0 \leq AT \leq 1.0$
- BT: tensile damage evolution parameter β_t . $BT \geq 0.0$
- AC: compressive damage evolution parameter α_c . $0 \leq AC \leq 1.0$
- BC: compressive damage evolution parameter β_c . $BC \geq 0.0$



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Increasing of tensile strength for first tensile crack (from R9.2):

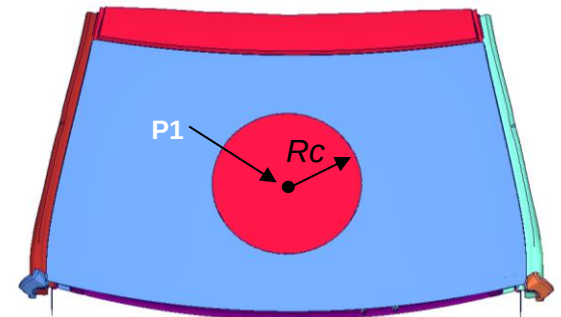
- FTSC: If RATENL=0.:
Factor, that scales tensile strength FT until the first tensile crack is detected.
Thereafter, the tensile strength is set to FT.

Element deletion (from R11.0)

- EPSCR: >0: maximum effective strain strain $\varepsilon_{eff} = \sqrt{\frac{2}{3} \varepsilon_{ij} \varepsilon_{ij}}$ for element deletion
<0: |EPSCR| is critical crack opening displacement (R11.1 needed)

Nonlocal failure criterion (from R11.0)

- ENGCR: Critical energy for nonlocal failure criterion
- RADCR: Critical radius for nonlocal failure criterion



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Nonlocal strain rate dependent tensile strength (from R11.0.0):

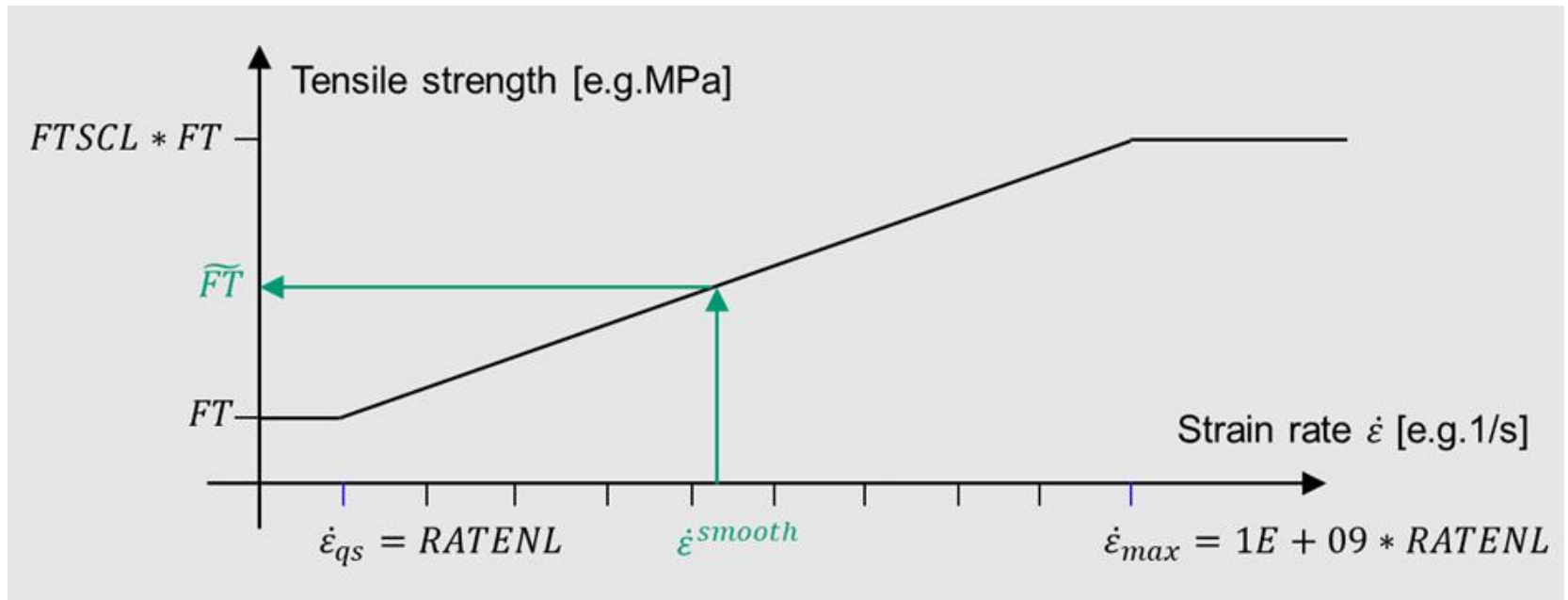
- **RATENL:** quasi-static strain rate threshold value.
If RATENL≠0, the nonlocal strain rate dependent tensile strength adaption is activated.
- **FTSCL:** If RATENL≠0.:
Factor, that scales tensile strength FT until the information of a neighbor element is transmitted, that the neighbor-element is cracked
- **RFILTF:** If RATENL ≠0.:
smoothing of effective strain rate for evaluation of current tensile strength:
$$\dot{\epsilon}^{smooth} = RFILTF * \dot{\epsilon}^{smooth,n} + (1.-RFILTF) * \dot{\epsilon}^{n+1}$$

RFILTF=0.: no smoothing
RFILTF=0.99: strong smoothing

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Nonlocal strain rate dependent tensile strength (from R11.0.0):

- Initialization of all Integration points with tensile strength: $FT_{initial} = FT_{max} = FT_{SCL} * FT$



At the moment: $\dot{\epsilon}_{max}$ is not a material parameter

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Nonlocal strain rate dependent tensile strength (from R11.0.0):

- If one integration point in an element recognizes failure, the tensile strength in the neighbor-elements is adapted to

$$\widetilde{FT} = \begin{cases} FT_{min} = FT & \text{if } \dot{\epsilon} \leq RATENL \\ FT + (FTSCL * FT - FT) * \frac{\log(\dot{\epsilon}/\dot{\epsilon}_{qs})}{\log(\dot{\epsilon}_{max}/\dot{\epsilon}_{qs})} & \text{if } RATENL \leq \dot{\epsilon} \leq \dot{\epsilon}_{max} \\ FTSCL * FT & \text{if } \dot{\epsilon} \geq \dot{\epsilon}_{max} \end{cases}$$

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History variables – post-processing:

- history variable 1: Crack Flag
 - 1 = one tensile crack in element (failure under tension)
 - 2 = two tensile cracks in element (failure under tension)
 - -1 = failure under compression
- History variable 2:
1st crack: direction of maximum principle stress (angle in rad) related to element direction
(direction of first tensile crack is perpendicular to this direction)
- History variable 3:
2nd crack: direction of 2nd principle stress (angle in rad) related to element direction
(direction of second tensile crack is perpendicular to this direction)

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History variables – post-processing:

- history variable 4:
failure criterion value
- History variable 7: (from R11.0.0)
currently used tensile strength if RATENL≠0.
- History variable 8: (from R11.0.0)
current strain rate

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Remarks to PVB behavior

- strongly rate dependent nonlinear elastic
- Hysteresis
- Temperature dependent
- ***MAT_SIMPLIFIED_RUBBER (*MAT_181)**
take care about really smooth curves and smooth derivative in LC/TBID
for nonlinear-elastic rate dependent stress-strain behavior
- ***MAT_HYPERELASTIC_RUBBER (*MAT_077)**

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