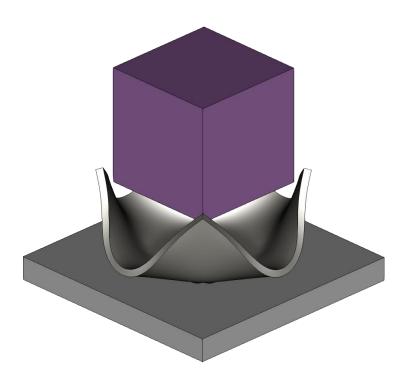


# Guideline for User Defined Interfaces in Ansys LS-DYNA Software



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## / Abstract

In this document, some of the possibilities for user-defined features in Ansys LS-DYNA software will be presented. The objective is to provide a basic foundation for users with previous experience of the Ansys LS-DYNA software that are interested in starting to develop customized functionality. Both Fortran code examples of user subroutines and accompanying simulation models (keyword files) are provided

This document is under continuous development, and future improved revisions will be released.

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### 1 Introduction

The multi-physics solver software Ansys LS-DYNA [1][3] has many pre-defined building blocks (traditional finite elements and other spatial discretization options, over 250 material models [2], many possible contact interactions etc.) for a wide range of different analysis types. For these pre-defined building blocks, the user can input parameter values, for example the hardening curve for a material model, while the behavior based on the given parameter values will be determined by the software.

In addition to these pre-defined modelling capabilities, Ansys LS-DYNA software also offers many possibilities for the user to define fully customized building blocks, like material models, elements, friction models and loadings, see Ref. [20] for an overview. These customer defined features plug into LS-DYNA via user interfaces, involving (amongst others) writing Fortran code. However, most analysis demands are met by corresponding built-in keywords, and it shall be stressed that writing subroutines in Fortran is rarely required for standard analysis.

The wide range of user interfaces offers great possibilities for researchers, either academic or company research, to implement their own developments as building-blocks within the already available LS-DYNA solver environment (rather than developing a complete FE-solver from scratch for research purposes). It is possible to develop a more-or-less fully customized FE-solver in a stepwise fashion: for example, starting with a user defined material model in combination with pre-defined elements, then writing user defined element routines, in the next step combining with user defined friction, and in the end adding even a user defined linear equation solver for implicit analysis.

The purpose of the present document is to provide an overview of some of the possibilities for user-defined features. Traditionally, the user defined features have been seen as a very advanced topic, mostly used by senior researchers and highly experienced specialists. Hopefully, this guideline can shed some light on the development of user defined features from a more applied viewpoint, opening possibilities for experienced LS-DYNA users to also work with user defined features. The everyday user should be able to gain insight into the possibilities that user defined features offer, should the predefined building blocks of LS-DYNA seem insufficient for solving a specific task.

It is assumed that the reader is familiar with common engineering terms and has knowledge of continuum mechanics and finite element theory. In addition, some years of experience of the Ansys LS-DYNA software is assumed. Basic keywords or FE modelling will not be discussed.

For most user-defined features treated in this Guideline, examples will be given both in the form of Fortran code and keyword files. Please note that the provided examples are intended for demonstrational purposes only. They should not be used in any type of daily production analysis. The user should review all provided examples with critical eyes.

This Guideline assumes that version R11 or later of the Ansys LS-DYNA software is used. Note that user defined features may be changed, removed, or added in later versions. Versions prior to R11 partially had a different lay-out of the source code for the user defined features. This will not be discussed in any detail in the present document.

This Guideline is currently focused on developing user features in a Linux environment, but some details regarding user feature development in Windows will also be mentioned.



For general support, see <u>Isdyna.ansys.com/knowledge-base/</u>. Useful publications from LS-DYNA users and developers may be found on <u>Isdyna.ansys.com/conference-papers/</u>. Example keyword files can be found at <u>Isdyna.ansys.com/</u>. For further questions, or if errors are found in this document, please contact your local Ansys LS-DYNA supplier.

The present document is based on the course notes [4][5] developed by Dr. Thomas Borrvall, Dr. Jesper Karlsson, and others. Course notes [6][7] developed by Dr. Tobias Erhart have also been of great help. The present document was developed by Dr. Anders Jonsson.

#### 1.1 Co-simulation and interaction with other software

There are potentially many applications for running Ansys LS-DYNA software together with some other software, as a part of co-simulation, for example for applying hydraulic pressure from a system-level simulation tool to a detailed structure model [32]. Even though it may be possible to interact directly with other software via user-defined interfaces [33] the preferable approach is to use the Functional Mock-up Interface (see <a href="https://fmi-standard.org/">https://fmi-standard.org/</a>) standard for exchanging data and synchronization of the solutions. This is available in Ansys LS-DYNA since R12, via the FMU Manager (see Ref. [1] for details and download instructions) and the built-in \*COSIM [30][31] keywords. It offers many possibilities for transferring loads, pressures, displacements etc. both ways between LS-DYNA and other software (Python, Matlab/Simulink, Adams, etc.). In that sense, FMI can provide functionality similar to user defined loadings. For a general introduction to the co-simulation capabilities of the Ansys LS-DYNA software, see the <a href="Webinar">Webinar</a> available from the Ansys Training Center.

## 2 Overview

See Table 1 for a quick guide to the user features, corresponding subroutine, which Fortran source code file it can be found in, and the section of the Guideline where it is discussed.

In Section 3, some prerequisites required to get started are discussed, including some instructions regarding the LS-DYNA usermat package, and some hints to Fortran programming.

In Section 4, user defined material models are discussed. The user defined materials are accessed via the keyword \*MAT\_USER\_DEFINED\_MATERIAL\_MODELS, and the related subroutines umatXX are found in the Fortran files dyn21umats.f and dyn21umatv.f of the usermat package.

In Section 5, user defined friction models are presented. The related keyword is \*USER\_INTERFACE\_FRICTION, and the subroutines usrfrc and mortar\_usrfrc are found in the Fortran file dyn21cnt.f of the usermat package.

In Section 6, user defined weld tie conditions are described. The related keyword is \*CONTACT\_AUTOMATIC\_SURFACE\_TO\_SURFACE\_MORTAR\_TIED\_WELD\_THERMAL, and the subroutines mortar usrtie are found in the Fortran file dyn21cnt.f of the usermat package.

In Section7 the (almost) inverse, namely user defined tiebreak conditions for Mortar contact are described. The related keyword is \*CONTACT\_AUTOMATIC\_SURFACE\_TO\_SURFACE\_TIEBREAK\_USER\_MORTAR, and the subroutines mortar usrtbrk are found in the Fortran file dyn21cnt.f of the usermat package.

Section 8 briefly treats user defined loadings. The related keyword is \*USER LOADING {SET}, and the subroutines loadud and loadsetud are found in the Fortran file dyn21.f of the usermat package.

Table 1. Overview of the user defined features

User defined feature	User defined feature Subroutine		Section
Material models	umat <i>XX</i> , utan <i>XX</i>	dyn21umats.f,	4
		dyn21utan.f	
	umat <i>XX</i> v	dyn21umatv.f	
Friction	usrfrc,	dyn21cnt.f	5
	mortar_usrfrc		
Weld tie	mortar_usrtie	dyn21cnt.f	6
Tiebreak	mortar_usrtbrk	dyn21cnt.f	7
Loading	loadud, loadsetud	dyn21.f	8

Provided examples (keyword files, Fortran code) were tested with mpp/LS-DYNA R13.1, R14.1.0 and R15.0.2 double precision, sse2 (also avx2 for examples not involving contacts) under Linux, with acceptable results in all cases. All examples except the user defined friction for non-Mortar contacts (usrfrc) were also tested with smp/LS-DYNA R14.1.0 and R15.0.2, with acceptable results.

## 3 Prerequisites

This Section describes what is required to get started with creating user defined features for Ansys LS-DYNA software. The first step is to download a programming environment (the usermat package) from your LS-DYNA provider, see Section 3.1. Since all interaction with the user defined features will require Fortran programming, a Fortran compiler is a fundamental requirement for getting started. This is not included in the LS-DYNA usermat package. An overview of recommended compilers is presented in Section 3.2, and a very brief introduction to Fortran programming is given in Section 3.6. How to build a user defined module is described in Section 3.3, and how to integrate it in a simulation model using the appropriate LS-DYNA keywords is described in Section 3.4.

It is mainly assumed that programming and user feature development is done in a Linux environment, but some special considerations when working in Windows environment are mentioned in Section 3.5.

Appendix A of Ref. [1] also contains a general overview of how to get started working with user defined features.

Do not hesitate to contact your local Ansys LS-DYNA provider for questions regarding download of required files, or setup of compilers for different environments, or other issues related to the user defined features.

## 3.1 Download the Ansys LS-DYNA usermat package

It is possible to integrate the user defined features into LS-DYNA using either static or dynamic<sup>1</sup> linking. Static linking means that a special version of LS-DYNA is built, which contains the user defined features. This was the traditional way of working with user-defined features. It is straight-forward but offers little flexibility. It is hard to integrate more features from different sources, for example 3<sup>rd</sup> party material routines with in-house development friction models.

Dynamic linking, see also Figure 1, means that a shared object (a file with extension .so in Linux) is built which then can be dynamically linked to a sharelib – version of LS-DYNA using the \*MODULE - keywords, see Section 3.4. The dynamic linking approach offers more flexibility since many shared objects from different sources can be linked to the same (standard) LS-DYNA main binary. Building a shared object may also be less resource intensive in terms of compiler and linking time. Currently², the dynamic linking approach is only supported by LS-DYNA under Linux.

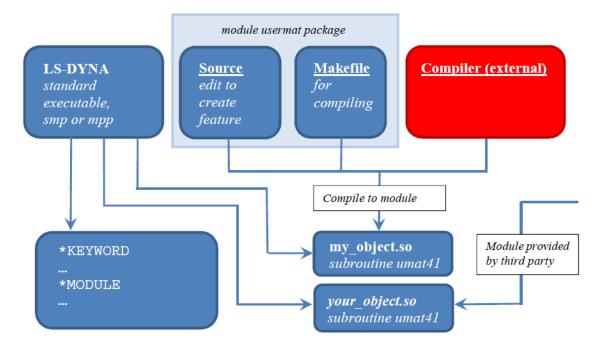


Figure 1. Integrating the user defined features into LS-DYNA using the shared object approach. Image from Ref. [7].

The first step to get started with the development of user-defined features for LS-DYNA is to decide which approach to use, either static or dynamic linking, and then download the required files from your local Ansys LS-DYNA provider. The Linux-version package for dynamic linking will typically be named

ls-dyna\_mpp....sharelib.usermat.tar.gz **or** ls-dyna\_mpp....sharelib.usermat.tar.gz\_extractor.sh and for static linking

<sup>&</sup>lt;sup>2</sup>By official LS-DYNA versions available up to 2024-12-15.



Ansys LS-DYNA user-defined //

<sup>&</sup>lt;sup>1</sup> From R9. Currently only supported for LS-DYNA under Linux.

```
ls-dyna_....usermat.tar.gz or ls-dyna_....usermat.tar.gz_extractor.sh
```

The files will be packaged in a compressed archive format, so after downloading unpacking will be required.

Just as for "standard" LS-DYNA, it is important to distinguish between SMP/MPP/hybrid and single/double precision. This means that the user defined features must be compiled and linked for the right "flavor" and version of LS-DYNA; for example, a shared object compiled for single precision LS-DYNA can in general not be plugged into double-precision LS-DYNA executable, and vice versa. Also, the sse2/avx2/avx512 extensions must be taken into consideration, so that the shared object is built for the corresponding extension - a shared object file based on sse2 will not work properly when used with LS-DYNA with the avx2 - extension.

When working with static linking, it is obvious that re-compilation and new linking of the user-defined features will be required in order to use the user defined features with a new LS-DYNA version. It will also be required for the dynamic linking approach: the shared object file must be built with the corresponding environment for a specific LS-DYNA version, since for example interfaces and arguments to subroutines may have changed between versions. This means that also when working with dynamic linking, the user features must be re-compiled in order to be used with a new LS-DYNA version.

The Fortran files of the usermat package come with some example material models and related subroutines, which may be used as templates for further developments.

## 3.2 Fortran compilers

Intel's Fortran compiles are in general recommended for both Linux and Windows. More specifically [4], for

- Linux Redhat or CentOS, use
  - Intel Fortran Compiler [10] 2013 for LS-DYNA R9,
  - Intel Fortran Compiler 2016 for LS-DYNA R11 and R12,
  - Intel Fortran Compiler 2019 for LS-DYNA R13, R14 and R15.
- Linux Suse, use PGI Fortran Compiler 16.5 (10.5 for LS-DYNA R9) for LS-DYNA R11 and R12,
- Windows x64, see Section 3.5.

Some special setup of the system environment will most likely be required. For example, when working with the Intel Fortran compiler under Linux, the command

```
compilervars.sh -arch intel64 -platform linux
found in the compiler installation directory, under (for Intel Fortran 2016)
```

compilers and libraries 2016/linux/bin/ must be issued in order to apply appropriate settings before compilation can be performed. For compiling mpp/LS-DYNA, or shared objects, also the appropriate mpi-Fortran wrapper (for example mpiifort for Intel MPI) must be configured. For Intel MPI under linux, the command

compilervars.sh -arch intel64 -platform linux



found in the MPI installation directory, under (for Intel MPI 2019)

compilers and libraries 2019/linux/bin/ must be issued in order to apply appropriate settings before compilation for mpp, using mpiifort, can be performed.

For further details on this, consult the documentation of the compiler in question, or contact your local Ansys LS-DYNA provider for support.

## 3.3 Compiling the subroutines for using Ansys LS-DYNA with user defined features

Once the LS-DYNA usermat package has been downloaded and unpacked, the next step is to set up the compiler environment and test-compile the files as-is, without any modifications. In Linux, the Makefile contains the instructions for compiling and linking that are required to build either a shared object file (libmppdyna ... .so) or a statically linked customized LS-DYNA-version (the result will in the latter case be a monolithic executable called mppdyna or lsdyna). In Linux, open the text file Makefile and edit the Fortran compiler command to the appropriate for the present installation (see Section 3.2). The next step is to build the desired shared object, which is obtained by executing the Linux make command. In some cases, it may be required to re-compile all objects, this can be achieved by first issuing make clean and then the make command.

Finally, run a small test model, for example tension of some solid elements (see for example, Section 4.6.1), without any user defined features active in the model but still including the shared object (see Section 3.4 for details) to verify that that LS-DYNA runs as expected.

The purpose of this initial testing stage is to establish a basis for further development of user defined features for LS-DYNA. It is good practice to have sorted out any problems directly related to compiling and linking the shared object files, or problems related to loading them into the LS-DYNA simulation model, before starting to work with the development of advanced user-defined features. Then, if errors should occur at later stages, troubleshooting can be more efficiently focused directly on the likely cause.

Once this testing stage is completed, the development of user defined features can commence, preferably in an incremental and iterative way, as outlined in Section 3.6.5.

## 3.4 Plugging a user-defined shared object into Ansys LS-DYNA

The shared object file must be built with the corresponding environment (usermat package) for a specific LS-DYNA version. This means that also when working with dynamic linking, the user features must be re-compiled in order to be used with a new LS-DYNA version. Note also that specific double / single precision versions of the .so - files must be built. Also, the hardware acceleration extension (sse2, avx2 or avx512) must match between the usermat package and the LS-DYNA version. For example, let us assume that a user-defined material model has been developed for LS-DYNA R11.1.0 sse 2, and this has been compiled to the shared object file libmppdyna R111.so. In order to use this userdefined material model with LS-DYNA R12.2.2 avx2, a new shared object file (which can be named for example libmppdyna R121 avx2.so) must be built using the corresponding usermat package. This

means adapting the user contributed source code to the dyn21-files in the new package. The dyn21-files should **never** be copied between versions.

The LS-DYNA keywords required to dynamically link a user-defined shared object to the main LS-DYNA binary all begin with \*MODULE [1]. The path to a shared object can be specified by the keyword \*MODULE\_PATH. A shared object is loaded by the keyword \*MODULE\_LOAD. In order to map the user subroutines loaded in shared objects to the model, the keyword \*MODULE\_USE is applied, for example in situations where user defined materials for different 3<sup>rd</sup> party suppliers should be combined in the same simulation model. If only a single shared object is used, it can also be linked using the environment variable LD\_LIBRARY\_PATH or the command line argument "module=", instead of \*MODULE."

An example of the use of these keywords is also presented in Section 4.1.

## 3.5 User-defined feature development in a Windows environment

In this section, some special details regarding user defined feature development for LS-DYNA running in a Windows environment will be mentioned. For the Windows versions of LS-DYNA, the only option currently available is to work with static linking. This means that a special version of LS-DYNA is built, including the user defined features. The Windows-version package for static linking will typically be named <code>ls-dyna\_...\_winx64\_...\_lib.zip</code>, or <code>ls-dyna\_...\_winx64\_...\_lib\_installer.exe</code>, see also Figure 2.

A brief instruction on how to build a LS-DYNA executable is given in the readme.txt-file provided in the Windows usermat package.

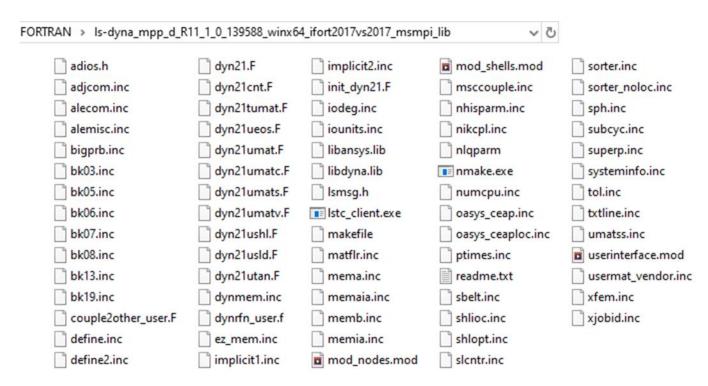


Figure 2. Example of contents of the usermat package for Windows.

The recommended Fortran compiler for LS-DYNA R11 and R12 is

Intel Parallel Studio XE 2017



and the corresponding Microsoft application environment, required for linking and access to standard libraries, is

Microsoft Visual C++ 2017 x64 Cross Tools.

For LS-DYNA R13, R14 and R15, the recommended Fortran compiler is

Intel Parallel Studio XE 2019 (Update 6, Composer Edition)

and the corresponding Microsoft application environment, required for linking and access to standard libraries, is

Microsoft Visual C++ 2019 x64 Cross Tools.

Note that the recommended compilers and tools may change for coming LS-DYNA versions, please see the information provided in the readme.txt-file for updated information.

In addition, the MPI for mpp/LS-DYNA under Windows is required. Note that it is also required to install the MS MPI Software Development Kit (SDK). Recommended are Microsoft MPI v8.1 for R11, and Microsoft MPI v10 for R12 of LS-DYNA, which are available for free download from www.microsoft.com.

The command nmake.exe [26] is used for building the LS-DYNA executable, based on the information in the makefile text file. Note that it may be required to update the search paths in the makefile text file depending on the local installation.

In order to run the customized LS-DYNA version, the executables ansyscl.exe and lstc client.exe and the library libiomp5md.dll (provided with the Intel Fortran compiler installation) need to be in the same folder as the LS-DYNA executable. In some cases, Windows security settings or antivirus programs may prevent execution of a customized LS-DYNA version. How this may be remedied is discussed further in Appendix A.

## 3.6 A very brief introduction to Fortran programming

All interaction with the user defined features of LS-DYNA will involve some amount of Fortran programming. This section is taken from the course material [4] and the purpose is to give a very brief introduction to programming in Fortran. Other sources are for example, from Intel. Fortran's official home page is https://fortran-lang.org/.

Another good starting point seems to be https://www.tutorialspoint.com/fortran/fortran\_useful\_resources.htm

#### 3.6.1 Input format

Fortran is an imperative programming language. A typical program consists of a set of statements, which are executed sequentially. In pseudo-code, a typical lay-out looks something like this:

PROGRAM name declarations executable statements END



The input is case-insensitive, and recommended practice is to always use lower case (upper case only use for emphasis here). Often fixed-format input is used, which means that each Fortran statement is written in positions 7 – 72 on a line. Position 6 is reserved as a continuation marker; in order to continue a statement on the next line, put a character in position 6 of the continued line. If a 'C' or 'c' is found in position 1 of a line, that line is taken as a comment. This is a convenient feature, making it easy to write explanations about what certain parts of the code are supposed to do, or simply de-activating certain lines of code. Position 1 – 5 are reserved for statement numbers or labels.

#### 3.6.2 Variables and arrays

Variables can be of type INTEGER, REAL (floating point numbers) or REAL\*8 (double precision floating point numbers), CHAR (characters or strings) or LOGICAL (Boolean). Variable names can be up to 31 characters long. For example:

```
INTEGER i, j,k
REAL a,b,c
REAL*8 d,e,f
```

If the variable declarations are omitted, the compiler will make certain assumptions regarding types of variables, for example, that variables beginning with the letters I, J, K, L, M, N are INTEGER. The lack of specification often leads to program errors, and it is strongly recommended that variable types are always declared. The statement IMPLICIT NONE means that all variables must be declared, and use of an undeclared variable will cause a compilation error.

Arrays are fields of variables, and are declared as for example

```
INTEGER ii(10), ndx(100)
REAL*8 sigma(6), avec(100)
```

Variables can then be assigned values by the = statement and be used for arithmetic operations (summation by +, multiplication by \*, division by / and subtraction by -), for example:

```
j=20
b=2.49
avec(j) = b
```

The Fortran 90 standard offers convenient input for operations on arrays, where a sequence of elements can be accessed in one line, reducing the need for DO - loops, for example:

```
s(1:3) = sig(1:3) - sum(sig(1:3))/3.0
```

#### 3.6.3 Subroutines

Subroutines are used to simplify coding and should be used for code that must be executed many times.

```
SUBROUTINE subname (parameters)
IMPLICIT NONE
declarations
statements
RETURN
END
```

In the main program, or from other subroutines, the subroutine is called by



```
CALL subname(parameters)
```

The list of parameters in the call and the subroutine declaration must match. In subroutines, arrays in the parameter list can be declared with assumed size, by use of an asterisk / star within parenthesis to declare its size as flexible, for example

```
SUBROUTINE setzero(a, la)
IMPLICIT NONE
REAL*8 a(*)
INTEGER la
...
statements
RETURN
END
```

In Fortran, all parameters to a subroutine are passed by reference. This means that if a parameter is changed (assigned a new value) in the subroutine, it will also be changed in the context where it was called from. This also means that special care must be taken when programming, so that only variables that are intended as output from a subroutine are updated.

#### 3.6.4 Some basic Fortran statements

To repeat statements iteratively a specified number of times, the DO / ENDDO construct can be used, for example

```
DO var=first, last
statements
ENDDO
```

The variable var will take values first, first + 1, ..., last. In order to break a DO – loop, the statement EXIT can be used, for example

```
DO iter=1, maxiter
statements
IF(residual.LE.tol)THEN
EXIT
ENDIF
ENDDO
```

To control the program flow based on logical conditions, the  $\mbox{if}$  /  $\mbox{then}$  /  $\mbox{else}$  construct can be used, for example

```
IF ABS(s) .GT. 1.E-15 THEN
    si = 1./s
ELSE
    si = 1.E16
ENDIF
```

The GOTO statement can be used to make the execution continue on the line with a specified statement number, for example

```
GOTO 10
```

jumps to statement number 10.

#### 3.6.5 Programming for user-defined features

In this Section, some tips specific for programing user-defined features in Ansys LS-DYNA follow.



When developing the user-defined feature, it is recommended to add some debug printouts in order to make sure that the user-defined feature actually is active and called from the main LS-DYNA binary. The user can get access to LS-DYNA's message files (messag, mes0\*) and highspeed-printout file (d3hsp) via the standard Fortran write – command. This is simplified by including the file iouinits.inc in the subroutine, which contains the unit number for the message files (mes0\*) in the variable iomsg and the d3hsp – file in the variable iohsp. An example follows:

```
INCLUDE 'iounits.inc'
.
write(iomsg, *) 'user defined feature writes to the message file'
.
RETURN
END
```

It is recommended to work with the development in a stepwise fashion, making small modifications and checking that each modification has the intended effect. Also starting with small (a couple of hundred elements for example) FE-models is recommended. It is good practice to verify the results from the user defined feature by comparing to handbook solutions or built-in LS-DYNA functions.

Once the developed feature is considered ready for use in a larger scale context, it is recommended to disable or limit debug printouts, since these may slow down performance significantly and create big output files.

## 4 Material models interface

The over 250 built-in material models in LS-DYNA [2] cover many applications, from linear elasticity to orthotropic plasticity models, foams, and composites. A web-based material model selection guide is available from <a href="https://lsdyna.ansys.com/dynamat/">https://lsdyna.ansys.com/dynamat/</a> [24]. Still, there may be cases where specific customer demands cannot be perfectly matched by the existing material models, and perhaps the most common customization of Ansys LS-DYNA is the development of a specialized material model. Third-party companies, like MatFEM [16] and E-Extreme / Digimat [17], deliver material models for specific purposes, like advanced failure modelling and analysis of composites. From R13 of LS-DYNA, also the option to add plasticity, viscoelasticity, creep etc. to already existing material models is available by the keyword \*MAT\_ADD\_INELASTICITY [29] in a quite general manner. Fairly advanced damage and failure models can be added to many of the built-in material models, by the keywords [2] \*MAT\_ADD\_DAMAGE\_{GISSMO/DIEM} see for example Refs. [14][15] for examples and background.

This section will not focus specifically on material modelling as such, since it is a very wide field, with many specializations for metals, composites etc. A vast amount of published research is available, and it is currently a very active research field. For a background to material modelling and continuum mechanics, the reader is referred to for example Refs. [11][12][13].

In the following, it is assumed that an existing valid material model is to be implemented as a user defined material model in LS-DYNA. The user-defined material subroutine  $\mathtt{umat}XX$  (where  $41 \le XX \le 50$ ) is called in the solution sequence, with the strain rate (or deformation gradient) as main input, and its objective is to update the (Cauchy) stress  $\sigma$  (and history variables, if required) for the next time step, in order that nodal forces can be computed, see Figure 3. All input quantities are passed to the user

subroutine in the local element coordinate system, so no additional transformations are required by the user. In the case of an anisotropic material model, the same options for specifying material directions as for the built-in LS-DYNA materials are available (see the remarks related to \*MAT 2 of Ref. [2]). Some care must be taken when implementing the material model for shells and beam elements, since it is then required that the material model will deliver stresses that are consistent with the assumptions related to structural elements (for example  $\sigma_{33}$  = 0 for shells). In the implicit solution sequence, also the material stiffness matrix is required, and the corresponding subroutine utanXX is called when assembling the global stiffness matrix.

It is also possible to create user defined thermal material models, see Appendix H of Ref. [1], but this option will not be discussed further in the present release of this Guideline.

The keyword interface for passing parameter values etc. to the user defined material model is described in Section 4.1. Post-processing the results from user subroutines is outlined in Section 4.2. The Fortran interface to the user subroutines is described in Section 4.3. Some useful pre-defined subroutines for common operations, such as push-forward, are described in Section 4.4. Finally, in Section 4.5, some examples of material model implementation are presented.

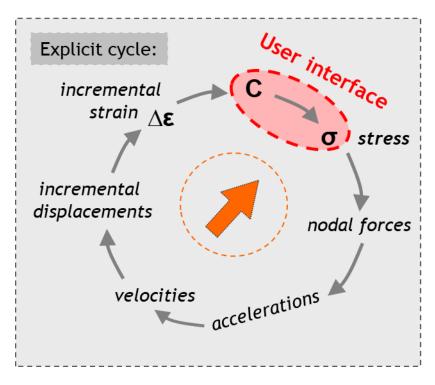


Figure 3. In the explicit solution sequence, the objective of the user-defined material interface is to output the updated stress based on provided incremental strain and current stress. Image from Ref. [7].

See also Appendix A of the Keyword manual [1] for more details on user defined material models. A dedicated course in User Defined Material Models is available from the Ansys Training Center.

## 4.1 Keyword interface to the user defined material models

The keyword interface to the user defined material models is given by

\*MAT USER DEFINED MATERIAL MODELS. By this, a user defined material can be assigned a Material ID (MID) that in turn can be referenced by a \*PART - definition. One example of the keyword syntax follows:

*MAT	USER D	EFINED MATE	ERIAL MODEL	S TITLE				
User	_define	d J2 plast:	icity UMAT4	1				
\$#1	MID	RO	MT	LMC	NHV	IORTHO	IBULK	IG
	3	2.700E-9	41	8	1	0	3	4
\$#2	IVECT	IFAIL	ITHERM	IHYPER	IEOS	LMCA	UNUSED	UNUSED
\$#	P1	P2	Р3	P4	P5	Р6	P7	P8
	70.E3	0.31	61404.0	26718.	103.			

The variables of this keyword are:

- MID: The Material ID.
- RO: Mass density. This, in combination with the bulk and shear moduli, is required input for LS-DYNA to be able to compute the critical time step size for explicit analysis, and penalty factors for contacts and joints.
- MT: The user material type. In this case, MT = 41 means that the subroutine umat 41 (and also utan41 in the case of an implicit analysis) will be called.
- LMC: The number of material parameters to be input on the keyword and passed to the subroutine. In this case 8, meaning that the variables P1 - P8 will be read during initialization and passed to the subroutine umat 41 in the cm array. Fields left blank in the keyword input will be passed as zero.
- NHV: The number of history variables to be stored on integration-point level (max. 200). In this case, one history variable is defined.
- IORTHO: Set to 1 if material is orthotropic (necessary for definition of material axes). The default is an isotropic material model.
- IBULK, IG: The addresses of the bulk and shear moduli, respectively, in the parameters array. In this case, IBULK = 3 means that LS-DYNA can find the bulk modulus as the variable P3, and IG = 4 means that LS-DYNA can find the shear modulus as the variable P4.
- IVECT: If set to 1 the vectorized version of the material routine is called. Default is that the scalar version is called, once for each integration point. The vectorized vs. scalar version of a user subroutine is discussed mode in Section 4.3.
- IFAIL: Set to 1 if the material routine should be able to control element erosion of solids and shells.
- ITHERM. Set to 1 if the integration point temperature shall be computed and passed to the user subroutine.
- IHYPER: Flag for hyperelastic materials. By setting IHYPER = 1, the deformation gradient will be passed to the user subroutine. The implementation of a hyperelastic material is presented as an example in Section 4.5.1.
- IEOS: Flag for equation of state.
- LMCA: Length of additional material constants array (unlimited). •
- P1 ... Pn: Material parameters passed to the user defined material subroutine in the array cm. In this case, material parameters for aluminum are defined. P2 is the Poisson's ratio in this case, and P5 refers to a Load curve ID (103 in this case), which specifies the yield stress as a piecewise linear function of the accumulated effective plastic strain.



Should more than 10 different user defined material models be required, the \*MODULE - keywords provide a handy solution. The following example illustrates this: first, the keywords \*MODULE PATH and \*MODULE LOAD read in two different shared objects,

```
*MODULE PATH
... path to modules ...
*MODULE LOAD
$#1
               MDLID
                                                                          TITLE
                    1 Library 1
$#2 FILENAME
sharedobje01.so
*MODULE LOAD
$#1
               MDLID
                                                                          TITLE
                    2 Library 2
$#2 FILENAME
sharedobje02.so
```

By this, the shared object sharedobje01.so is loaded and assigned the module ID 1, and the shared object sharedobje02.so is loaded and assigned the module ID 2. By the \*MODULE USE keyword, the user material types (typically 41 ≤ MT ≤ 50) can be re-mapped to new ID numbers:

```
*MODULE USE
$#1
                MDLID
                     1
$#2
                 TYPE
                                      PARAM1
                                                                       PARAM2
UMAT
                                          41
                                                                          1001
*MODULE USE
$#1
                MDLID
                     2
$#2
                 TYPE
                                      PARAM1
                                                                       PARAM2
UMAT
                                          41
                                                                          1002
```

This means that the umat41 of sharedobj01.so can be referred to as material type 1001, and the umat41 of sharedobj02.so can be referred to as material type 1002 when creating a user defined material model:

```
*MAT USER DEFINED MATERIAL MODELS
$#1
     MID
                RO
           7.85e-9
       1
                       1001
*MAT USER DEFINED MATERIAL MODELS
$#1 MID
                RO
                        MT
        2
           2.70e-9
                       1002
```

Then these material models can be assigned to \*PARTS, as any built-in LS-DYNA material model:

```
*PART
first part
      PID
               SECID
                           MID
       1
                             1
*PART
second part
$#
      PID
               SECID
                           MID
```

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2 2 2

By a similar procedure, the limitation to 10 user defined material models can be overcome by separating the material models and placing 1 – 10 in the sharedobje01.se, and then 11 – 20 in the sharedobje02.so, and so on.

Note that one user defined material type, for example MT = 41, may be referenced by an unlimited number of \*MATERIALs (with different Material IDs). However, in that case only the parameters RO, and P1...Pn may be changed on each \*MATERIAL.

## 4.2 Post processing user defined material models

The number of additional material history variables to be output to the binary 3D databases d3plot, d3part and d3drlf is controlled by the NEIPH (output for solid elements) and NEIPS (for shell elements) variables of the keyword \*DATABASE\_EXTENT\_BINARY. For example, in order to post-process 5 material history variables for a user defined material model, set NEIPH = NEIPS = 5. This will store the first 5 history variables (hisv(1:5), see Section 4.3) in the binary databases. This means that the history variables of interest for post-processing should appear in the beginning of the hisv array. Interactive post-processing of the binary 3D databases d3plot, d3part and d3drlf, including extra history results, should be possible using LS-PrePost [22], META [23] or other third-party post-processors.

The number of additional material history variables to be output to the elout file (for history / 2D curve plotting) is controlled by the OPTION1 (for solid elements) and OPTION2 (for shell elements) variables of the keyword \*DATABASE\_ELOUT. For example, in order to post-process 5 material history variables for a user defined material model, set OPTION1 = OPTION2 = 5.

For the history output, note that also \*DATABASE\_HISTORY\_... keywords are required, in order to specify for which elements the data should be output.

One motivation for writing a user material subroutine may be that is makes it possible to output additional history results from a "standard" material model, for example back stress terms, or the maximum effective stress during a simulation.

The user defined subroutines can also write text output to the message (mes0\*) and d3hsp – files. This can be very useful, for example in case some initial parameter fitting to a given test curve is done, some quality measure indicating the validity of the fit can be output. It is at least during the development phase warmly recommended to add output of some text message to confirm which subroutine(s) that are called during the solution of a specific FE model.

# 4.3 Interface to the user-defined material models in the subroutine umat

There are two<sup>3</sup> basic categories of subroutines for user defined material models:

- If the subroutine should be called once for each element and integration point, it is denoted as a scalar (or serial) material subroutine. This corresponds to IVECT = 0 on the

  \*MAT\_USER\_DEFINED\_MATERIAL\_MODELS keyword. The subroutines are named umatXX, where 41 ≤

  XX ≤ 50, and can be found in the dyn21umats.f file.
- If the subroutine can process an array of elements for each integration point, it is denoted as a vectorized material subroutine. This corresponds to IVECT = 1 on the

  \*MAT\_USER\_DEFINED\_MATERIAL\_MODELS keyword. The subroutines are named umatXXv, where 41

  ≤ XX ≤ 50, and can be found in the dyn21umatv.f file.

While the scalar approach may be easier to code – since only scala data needs to be handled – the vectorized version will probably lead to faster solution timing; the reduced number of subroutine calls will lead to less overhead processing. It is in all cases to be expected that a user-defined version of even a simple material model will not be as fast as a built-in equivalent material model. If a user defined material model is to be implemented, it should be motivated by other benefits, for example new material behavior or enhanced post-processing, than improved solution speed.

The main subroutine for interfacing between LS-DYNA and the user defined material models is usrmat and can be found in the dyn21umat.f- file. It should normally not require editing, even though it is possible in order to pass additional information to the user defined subroutines [7].

As input to the material subroutine, LS-DYNA provides the

- Incremental strain,
- Current stress,
- History variables,
- Material parameters,
- Element type (solid, shell, beam, discrete beam ...) and
- Temperature.

The subroutine should, based on this and according to the material law, provide

• The stress in the next time step,

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- strain corrections for structural elements (beams / shells), and if required also
- updated history variables.

<sup>&</sup>lt;sup>3</sup> In addition, there is also the possibility to define cohesive user material models, by the subroutines umatXXc of the dyn21umatc.f – file. This will not be discussed further in the present revision of this Guideline.



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To be more specific, the parameter list for a scalar implementation of a user defined material model (in this case umat41) is:

```
subroutine umat41(cm,eps,sig,epsp,hsv,dt1,capa,etype,tt,
1 temper,failel,crv,nnpcrv,cma,qmat,elsiz,idele,reject)
```

An overview and brief description of the parameters to the subroutine is shown in Table 2.

Table 2. The arguments to the subroutine umat XX

Argument	Description	Input / Output
cm	Material constants	Input
eps	Local strain increment in Voigt <sup>(1)</sup> notation	Input /Output <sup>(2)</sup>
sig	Local stress in Voigt notation	Input / Output
epsp	Accumulated effective plastic strain	Input / Output
hsv	History variables	Input / Output
dt1	Current time step size	Input
capa	Reduction factor for transverse shear of shells and beams <sup>(3)</sup>	Input
etype	String describing element type	Input
tt	Current problem time	Input
temper	Current temperature	Input / Output
failel	Failure flag, set to .true. to indicate failure of an integration point	Input / Output
	(Requires $IFAL = 1$ , see Section 4.1)	
crv	Array representation of curves in the keyword deck	Input
nnpcrv	Number of discretization points <sup>(4)</sup> per curve	Input
qmat	Transformation matrix in case of IHYPER =3	Input
cma	Additional memory for material data	Input
elsiz	Characteristic element size	Input
idele	Element id	Input
reject	For implicit analysis: set to .true. if the current implicit iterate should	Output
	be rejected for some reason	

Notes: (1) This means that the symmetric stress and strain  $3 \times 3$  tensors are represented as  $6 \times 1$  vectors. (2) For shell elements, the through-thickness strain eps(3) shall be updated corresponding to the plane stress assumption. (3) This corresponds to SHRF on the \*SECTION\_SHELL or \*SECTION\_BEAM card. (4) The curves are internally re-discretized (to the number of points specified by LCINT on the \*CONTROL\_SOLUTION - keyword) in order to speed up the curve/table look up.

In LS-DYNA, the symmetric  $3 \times 3$  stress tensor ( $\sigma$ ) and the symmetric  $3 \times 3$  strain rate tensor  $\dot{\epsilon}$  are represented as  $6 \times 1$  vectors, stored in arrays  $\sin \alpha$  and  $\cos \beta$  using Voigt notation:

$$sig(1:6) = (\sigma_{11}, \sigma_{22}, \sigma_{33}, \sigma_{12}, \sigma_{23}, \sigma_{13})$$

and

eps(1:6) = 
$$(\dot{\varepsilon}_{11}, \dot{\varepsilon}_{22}, \dot{\varepsilon}_{33}, 2\dot{\varepsilon}_{12}, 2\dot{\varepsilon}_{23}, \dot{2}\dot{\varepsilon}_{13})$$
.

Note that this representation differs from for example Ref. [11].

The accumulated effective plastic strain (clearly a history variable) has its own parameter epsp outside the history variables array hsv. This means that for example basic  $J_2$  – plasticity can be defined (at least for explicit analysis) without any additional history variables (leaving NHV = 0 on

\*MAT USER DEFINED MATERIAL MODELS). In cases where both built-in materials (for example MAT\_24) and user-defined materials are used in the same FE-model, post-processing (for example fringe plotting the accumulated effective plastic strain) is easier if the user-defined materials also use the parameter epsp to store the accumulated effective plastic strain (if applicable).

The etype parameter is a string, describing the element type, see Table 3. If a material model should be generally valid also for structural elements (shells, beams) this will require special care with respect to different types of stress / strain updates in order to fulfill the assumptions (for example updating the through-thickness strain eps (3) for shells in order to fulfill sig(3) = 0.). This is illustrated further in the examples of Section 4.5. If is in any case good practice to add element type checks in the subroutine and stop LS-DYNA with a reasonable error description (see Section 4.4) in case there is an attempt to apply a user-defined material model to element types that are not properly supported.

Table 3. The element types and their string values

etype.eq.	Description	Coment
'solid'	3D solid elements	
'sph'	SPH, smoothed particle hydrodynamics	Not covered in this Guideline
'sldax'	2d solids, axisymmetric	Shells elform 14 and 15
'shl_t'	Shells with thickness stretch	Shells elform 25, 26, 27
'shell'	Shells without thickness stretch	All other shell elforms, and thick shell
		elform 1, 2
'tshel'	For thick shells	Thick shell elforms 3, 5. May use same
		material formulation as 3D solid
		elements
'hbeam'	For beam elements	Beam elform 1, 11
'tbeam'	For trusses	Beam elform 3
'dbeam'	Discrete element beams (springs, dashpots	Beam elform 6.
	etc.) see also Section 4.3.1	Not supported in implicit
'beam'	For beams	All other beam elforms

If the variable ITHERMAL = 1 on the \*MAT USER DEFINED MATERIAL MODELS - keyword, the material temperature at the current integration point is available in the temper parameter. By this, temperature dependent material behavior can be implemented. The conversion of deformation to heat is taken care of by LS-DYNA outside the user-defined material models, so there is no need for the user subroutine to update the temperature.

The user-defined material model can also involve a damage/failure model, and if the variable IFAIL = 1 on the corresponding \*MAT USERDEFINED MATERIAL MODELS -keyword, the material model can also indicate failure of an integration point, by setting the Boolean parameter failel = .true. The parameter is also input to the subroutine, which means that the logic of the subroutine must handle what to do with output of for example of stress and plastic strain after an integration point is failed: should the data be set to zero, or kept constant?

The capa parameter is the transverse shear reduction factor, corresponding to the value of SHRF on \*SECTION SHELL. How to use it will be demonstrated in the examples of Section 4.5.

For implicit analysis, the material routine can set the parameter reject = .true. to indicate that the current iterate should be rejected, for example if the increment of plastic strain (depsp in the example below) is above some tolerance threshold (epsinctol):

```
if(depsp.gt.epsinctol)then
  reject = .true.
  return
endif
```

This will trigger a RETRY in the non-linear implicit solver, meaning that the time step will be tried again with a smaller time increment.

For a vectorized implementation, which is invoked by setting IVECT = 1 on the

\*MAT\_USER\_DEFINED\_MATERIAL\_MODELS -keyword, data for a vector block of elements is passed to the user-defined material subroutine. It is required to include the file 'nlqparm' in the subroutine. The length of the vector block of integration point data is given by the parameter nlq, and the objective of the subroutine is to update the element data in the range from lft to llt.

To be more specific, the parameter list for a vectorized implementation of a user defined material model (in this case umat 41v) is:

```
subroutine umat41v(cm,d1,d2,d3,d4,d5,d6,sig1,sig2,
. sig3,sig4,sig5,sig6,epsps,hsvs,lft,llt,dt1siz,capa,
. etype,tt,temps,failels,nlqa,crv,nnpcrv,cma,qmat,elsizv,idelev,
. reject)
```

A brief overview of the parameters is presented in Table 4. The main body of a vectorized implementation is outlined in pseudo – code below:

```
subroutine umat41v (...,lft,llt,...)
include 'nlqparm'
. . . declare varibles and parametes ...

do k=lft,llt
. . .
    process element point k, update sig1(k), sig2(k)... sig6(k) etc.
    enddo
return
end
```

The implementation of an orthotropic material model can be simplified by setting <code>IORTHO = 1</code> on the material card (\*MAT\_USER\_DEFINED\_MATERIAL\_MODELS). By this, the local (material) coordinate system is defined by two additional cards, specifying how the coordinate system is formed and updated (this is described in some detail under \*MAT\_ORTHOTROPIC\_ELASTIC of Ref. [2]). With <code>IORTHO = 1</code>, all data passed to the constitutive routine <code>umatXX</code> (<code>umatXXv</code>) is in the local system and the transformation back to the global system is done outside this user defined routine.

Table 4. The arguments to the subroutine umat XXv

Argument	Description	Input / Output
CM	Material constants	Input
dX <sup>(1)</sup> (nlq)	The strain vector in element $k$ is $(d1(k), d2(k), d3(k),$	Input /Output
	d4(k), d5(k), d6(k))	
sigX(nlq)	Local stress in Voigt notation, components of the stress vector is	Input / Output
	$(sig1(k), sig2(k) \dots sig6(k))$	
epsp(nlq)	Accumulated effective plastic strains	Input / Output
hsvs(nlq, *)	History variables	Input / Output
lft ,llt	Loop over vectors, from lft to llt	
dt1siz(nlq)	Current time step sizes	Input
capa	Reduction factor for transverse shear of shells and beams	Input
etype	String describing element type	Input
tt	Current problem time	Input
temps(nlq)	Current temperature in element point $k$ is temps (k)	Input / Output
failels(nlq)	Failure flag, set to .true. to indicate failure of an integration point	Input / Output
crv	Array representation of curves in the keyword deck	Input
nnpcrv	Number of discretization points per curve	Input
cma	Additional memory for material data	Input
qmat	Transformation matrices in case of IHYPER =3	
elsizv(nlq)	Characteristic element sizes	Input
idelev(nlq)	Element ids	Input
reject	For implicit analysis: set to .true. if the current implicit iterate	Output
	should be rejected for some reason	

Notes: (1)  $\times$  goes from 1 to 6.

#### 4.3.1 Interface for discrete beam elements

Since discrete beam elements (ELFORM = 6 on \*SECTION BEAM), like springs, dashpots etc., normally work based on changes in element length rather than strains, this data is passed in the eps(1:6) array, for each degree of freedom, instead of strains. The objective of the user material subroutine for discrete elements is then to update the (generalized) forces, which are stored in the sig(1:6) array, for each degree of freedom. Thus, for discrete beam elements,

$$sig(1:6) = (F_1, F_2, F_3, M_1, M_2, M_3)$$

and

$$eps(1:6) = (du_1, du_2, du_3, dr_1, dr_2, dr_3),$$

both given in the local element coordinate system, see Figure 4. It is recommended to set scoor = ± 12 on \*SECTION BEAM, for correct update of the beam orientation.

User defined material models for discrete beams are currently not supported in implicit.

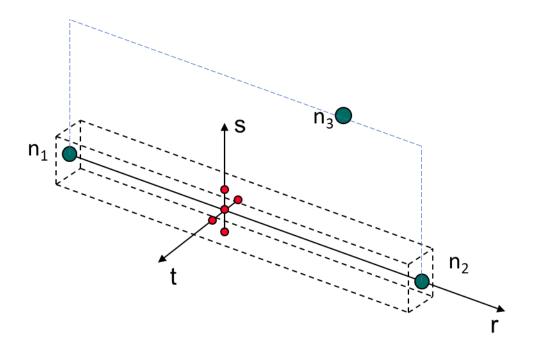


Figure 4. Orientation of discrete beams by a third node ( $n_3$ ). The ( $r_s$ ) – plane is defined by the three points ( $n_1$ ,  $n_2$ ,  $n_3$ ). See Ref [1] under \*SECTION BEAM for other options of orientation.

#### 4.3.2 Material tangent modulus subroutine utan for implicit analysis

For implicit analysis, also the material tangent modulus tensor for a user defined material must be provided. An algorithmically consistent implementation is required in order to achieve quadratic convergence rate in the global equilibrium iterations.

For the user material type XX, the material tangent modulus should be provided by the subroutine  $\mathtt{utan}XX$  in the case of a scalar material subroutine and by  $\mathtt{utan}XXv$  if a vectorized implementation of the material routine is given. These subroutines are found in the file  $\mathtt{dyn}21\mathtt{utan}.f$ . The input to the material tangent stiffness subroutine is similar the input to the material routine itself, but LS-DYNA also provides a flag  $\mathtt{unsym}$  in case the unsymmetrical linear equation solver is active (by  $\mathtt{LCPACK} = 3$  on \*CONTROL IMPLICIT SOLVER).

The subroutine  $\mathtt{utan}XX(v)$  should, based on the input and according to the material law, provide the consistent<sup>4</sup> material tangent modulus. This is to be stored in the 6 × 6 matrix es (or matrices  $\mathtt{dsave}(nlq, 6, 6)$  for a vectorized implementation). If the local coordinate system option for orthotropic materials ( $\mathtt{IORTHO} = 1$ ) is invoked for solid elements, then it should be expressed in this local system. For shell elements, it should be expressed in the co-rotational system defined for the current shell element. All transformations back to the global system are made by LS-DYNA after exiting the user-defined routine.

<sup>&</sup>lt;sup>4</sup> Or "best possible"



The parameter list for a scalar implementation of a user defined material tangent routine (in this case utan41) is:

```
subroutine utan41(cm,eps,sig,epsp,hsv,dt1,unsym,capa,etype,tt,
1 temper,es,crv,nnpcrv,failel,cma,qmat)
```

An overview of the parameters can be found in Table 2, with the main difference that for utan, almost all parameters except for es are purely input.

Since the subroutine umatxx has no information if an implicit or explicit analysis is taking place from the default parameters passed to it, one way of communicating this could be to let the utanxx subroutine set a material history variable (one entry in the hsv – array) as a flag. Then, in case some non-linear equation is solved (for example iterative radial return) different tolerances and iteration limits may be set to account for the higher accuracy requirements of an implicit analysis.

To be more formal, LS-DYNA computes the material stiffness matrix based on the constitutive modulus  $C_{ikjl}^{\sigma T}$ , relating the rate-of-deformation tensor to the Truesdell rate of Cauchy stress. The material stiffness matrix  $K^{\text{mat}}$  is expressed as

$$K_{iIjJ}^{\text{mat}} = \int_{\Omega} \frac{\partial N_I}{\partial x_k} C_{ikjI}^{\sigma T} \frac{\partial N_J}{\partial x_l} d\Omega$$

where N denotes the Finite Element basis functions. For more details, see Ref. [3].

## 4.4 Useful predefined subroutines

In this Section, some of the pre-defined subroutines available within the LS-DYNA usermat package that may come in handy for different common tasks in user defined material subroutines are described. See also Appendix A of Ref. [1] for detailed descriptions of some of the subroutines. In addition, some generally useful subroutines for other user-defined interfaces are presented.

Often, related to the material models, evaluation of curves input in the keyword deck via the \*DEFINE\_CURVE or \*DEFINE\_TABLE keywords, for example hardening curves, will be required. For this, the subroutines crvval and tabval may be called, in order to conveniently and consistently retrieve the values from the curve (or table) data to the subroutine. For example, in order to evaluate a curve to retrieve the yield stress corresponding to the current accumulated effective strain, the following could be used:

which will look up the curve given by ID lcid and return the yield stress at plastic strain epsp in the variable sigy, and in addition the slope of the curve (hardening modulus) in the variable h. Note that the curve ID should be passed as a float (lcid should be declared as real) which is convenient if for example element 5 of the material constants array cm should be the ID of a curve which specifies the yield stress as a function of plastic strain, then cm(5) can be passed directly to the subroutine crvval. The crv is the curve array and nnpcrv is the number of discretization points per curve; these data are

passed as parameters to the user material subroutine for the only purpose of evaluating curves, and should just be passed on to the look-up subroutines. The vectorized version of the call would be

where the plastic strain <code>epsp\_v</code>, the yield stress <code>sigy\_v</code>, and slopes <code>h\_v</code> will be arrays of dimension <code>nlq</code>.

The curvval and tabval subroutines will extrapolate y – values for x-values beyond the last curve entry based on the slope of the last segment on the curve. See also Table 5 for an overview of the parameters to the subroutine curvval.

Table 5. The arguments to the subroutine crvval

Argument	Description	Input / Output
crv	Array representation of curves in the keyword deck	Input
nnpcrv	Number of discretization points per curve	Input
lcid	Curve id (from *DEFINE_CURVE) as float	Input
epsp	x – value	Input
sigy	y – value	Output
h	Slope of the curve in point epsp	Output

The parameter list for a scalar implementation of the table-lookup subroutine tabval is:

subroutine tabval(crv,nnpcrv,lcid,dxval,yval,dslope,xval,slope) see Table 6 for an overview.

The curves are internally re-discretized (to the number of points specified by LCINT on the \*CONTROL\_SOLUTION - keyword, default is 100 points) in order to speed up the curve/table look up. If an evaluation based on the user-input curve data for curve (or table) ID lcid is desired, this can be requested by passing -1.\*lcid to the subroutine.

Table 6. The arguments to the subroutine tabval

Argument	Description	Input / Output
crv	Array representation of curves in the keyword deck	Input
nnpcrv	Number of discretization points per curve	Input
lcid	Table id (from *DEFINE_TABLE) as float	Input
dxval	x <sub>2</sub> – value	Input
yval	y – value	Output
dslope	Slope $\frac{\partial y}{\partial x_2}$ of the curve in point (x1, x2)	Output
xval	$x_1$ – value	Input
slope	Slope $\frac{\partial y}{\partial x_1}$ of the curve in point (x1, x2)	Output

When working with hyperelastic<sup>5</sup> materials, there are some useful pre-defined subroutines for common operations (which will also be discussed in more detail related to the example of Section 4.5.1). By setting <code>IHYPER = 1</code> on <code>\*MAT\_USER\_DEFINED\_MATERIAL\_MODELS</code>, the deformation gradient F will be passed to the user subroutine in the history variables array, as 9 components right after the requested number of history variables (<code>NHV</code>).

The transformation of a tensor from the reference configuration to the current configuration by use of the deformation gradient F is commonly denoted as a push-forward operation (see for example Section 3 of Ref. [13]). The push-forward of a  $2^{nd}$  order tensor is performed by the subroutine  $push_forward_2s$  (or the vectorized version  $subroutine_push_forward_2$ ) which is useful when going from a material model formulated with respect to the  $2^{nd}$  Piola – Kirchhoff stress tensor S to Cauchy stress  $\sigma$ , which is the expected output from the user defined material subroutine. In that case, it is important to remember that

$$\sigma = \frac{1}{\det \mathbf{F}} \mathbf{F} \mathbf{S} \mathbf{F}^T$$

which means that after the subroutine <code>push\_forward\_2s</code> is called, also a division by det F should be performed. The push-forward of a 4<sup>th</sup> order tensor (for example the material stiffness tensor) is performed by the subroutine <code>push\_forward\_4s</code> (or the vectorized version <code>subroutine</code> <code>push\_forward\_4</code>). Also, for working with shell elements and hyper-elastic materials, the subroutine <code>compute\_f3s</code> updates the third row of the deformation gradient considering the through-thickness stretch (eps (3) or \$\paralle{s}\_{33}\$ in the local coordinate system).

For solids elements, the LS-DYNA code will make the stress transformations required to obtain the objective Jaumann stress rate outside the user subroutine, but in cases where history variables also are stresses, for example the back stress tensor in a kinematic hardening model, the user must take care to apply the transformation to the history variables inside the user subroutine.

It is good practice to add some checking to the user subroutine, for example if the user tries to apply the material model to an unsupported element type, or if some of the input parameters are invalid, or if the material model is intended for explicit only. One option is to simply write a text message in the mes0\* - files to inform the user, using

```
write(iomsg,*) 'Warning message ...'
```

as outlined in Section 3.6.5. Another option is to issue an error message and stop the analysis. This can be done using the subroutine lsmsg. Calling this subroutine also requires that the file iouinits.inc is included. An example follows:

<sup>&</sup>lt;sup>5</sup> Or a material using a deformation-gradient based formulation



```
return
endif
if(cm(3).gt.0.5)then
   cerdat(1)='Illegal Poissons ratio'
   call lsmsg(3,MSG_SOL+1447,ioall,ierdat,rerdat,cerdat,0)
   return
endif
```

#### which will output

```
*** Error 41151 (SOL+1151)

element type beam can not be
run with the current material model.
```

in case an attempt is made to apply the user defined material model to a beam element, and

```
*** Error 41447 (SOL+1447)
Illegal Poissons ratio.
```

in case an invalid value is passed by the user as parameter 3 from the keyword file.

The relevant parameters to modify in the subroutine call are:

- MSG SOL + XXX, for changing the message ID / type
- cerdat (1) to specify what error message to print.

The error message ID MSG\_SOL + 1447 will display the text passed to cerdat (1) as in the second part of the above example.

Entities of the analysis model (nodes, parts, elements, etc.) will be stored using internal ID:s within the LS-DYNA code. These internal ID:s may differ from those specified by the user in the keywordfile. For example, if the user defines a node with ID 103, it may be assigned an internal ID of 1. In many cases, when handling user input to the user-defined interfaces, it will be required to convert between internal/external entity ID:s. Some useful functions for this purpose are listed in Table 7. In some cases, for example beam orientation nodes<sup>6</sup>, an offset may be applied to the internal

node numbers. An extra check is then required before converting internal to external node numbers:

if (i3.gt.100000000) i3=i3-1000000000

<sup>&</sup>lt;sup>6</sup> To obtain useful internal numbering of beam orientation nodes, it is required to set NREFUP = 1 on \*CONTROL OUTPUT.



Table 7. List of some functions for converting entity ID:s between internal and external (user defined) numbering

	Internal → External ID	External → Internal ID	
Entity		трр	smp
Node	lqfinv(ix,1) (1)	lqfe(ex,1) <sup>(2)</sup>	lqf8(ex,1)
Element solid	lqfinv(ix,2)	lqfe(ex,2)	lqf8(ex,2)
Element beam	lqfinv(ix,3)	lqfe(ex,3)	lqf8(ex,3)
Element shell	lqfinv(ix,3)	lqfe(ex,4)	lqf8(ex,4)
Part	lqfmiv(ipid)	lqfm(epid)	
Curve	ilcid(iid)	lcids(eid)	

Notes: (1) There is also a function lqfinv8 returing INTERGER\*8. (2) For mpp/LS-DYNA, lqfe will return -1 in case an entity is not found in the current mpi thread.

## 4.5 Subroutine examples

In this Section, two examples of user defined material models are presented, with quite extensive descriptions. Both Fortran code and keyword input are presented, while the complete examples, see also Section 4.6, can be found as attachments to this document. It shall be stressed that these examples are not intended for use in any kind of production analysis, and they may very well contain errors or flaws.

The Fortran files (dyn21umats.f, dyn21umatv.f and dyn21utan.f) of the usermat package already come with some examples of subroutines for user defined materials (which may vary slightly depending on version) and some general routines that can be used as a starting point for user defined subroutines, for example metalshl, metalshl (in dyn21umats.f) and metaltan (in dyn21utan.f) for  $J_2$ plasticity. Also, Appendix A of Ref. [1] has some examples, including descriptions.

#### 4.5.1 The Saint-Venant Kirchhoff model for solids and shells

This section describes the implementation of the Saint-Venant Kirchhoff model, a simple compressive isotropic hyperelastic material, for solids and shells. Using the Green-Lagrange strain tensor,

$$\mathbf{E} = \frac{1}{2} (\mathbf{F}^{\mathrm{T}} \mathbf{F} - \mathbf{I})$$

this model gives the 2<sup>nd</sup> Piola-Kirchhoff stress tensor S via a linear relation,

$$S = \mathbb{C}$$
: E

where C is a fourth order stiffness tensor (this is basically what the subroutine utanXX should return). Using the Lamé constants  $\mu$  and  $\lambda$ , the strain-energy function for the Saint-Venant Kirchhoff model is (see for example Section 6.5 of Ref. [13])

$$\Psi(\mathbf{E}) = \frac{\lambda}{2} (\text{tr}\mathbf{E})^2 + \mu tr \mathbf{E}^2$$

which, by

$$\mathbf{S} = \frac{\partial \Psi}{\partial \mathbf{E}}$$

$$S_{ij} = 2\mu E_{ij} + \lambda E_{kk} \delta_{ij}$$

The first step in implementing a material model, once the theory is known, is to consider the parameters to be input by the user on the \*MAT USER DEFINED MATERIAL\_MODELS - card. In the present implementation, it was decided to let the user input Youngs modulus E and Poisson's ratio v, since these elasticity parameters are often used in engineering applications, rather than the Lamé constants. By setting IHYPER = 1 on the \*MAT USER DEFINED MATERIAL MODELS - card, the deformation gradient (in Voight - notation, with components F11, F21, F31, F12, F22, F32, F13, F23 and F33) will be passed to the user subroutine in the history variables array hsv, on positions hsv (NHV+1:NHV+9). In this case, for a basic implementation of an elastic material, no history variables are required, so F can be retrieved from hsv(1:9). There is no indication by the interface parameters to the user subroutine (see Table 2) if the IHYPER variable actually is set to one, so the user subroutine will depend on correct user input.

In this case, the keyword interface to the Sain-Venant Kirchhoff model will be

```
*MAT USER DEFINED MATERIAL MODELS TITLE
Simple hyperelastic material UMAT43
$#1
     MID
               RO
                                LMC
                                         NHV
                                               IORTHO
                                                         IBULK
                                                                    ΙG
                                 8
  mat id
                        43
                                                                     4
              \rho
$#2 IVECT IFAIL ITHERM
                             IHYPER
                                                 LMCA
                                                        UNUSED UNUSED
                                        IEOS
                                 1
      P1
                                                           P7
               P2
                        P3
                                 Ρ4
                                          Р5
                                                   P6
                                                                    Р8
  Young's Poisson's
                         K
                                       el.ID
```

where blue text indicates that the user should input sensical data, and red text indicates values that should not be changed (since these fixed values also will be assumed by the Fortran implementation). By specifying an element ID for P5, some debug output (written to the mes0\* - files) will be activated.

The user subroutine umat 43 will be used to implement the material model in Fortran code. The first part of the subroutine follows:

```
subroutine umat43 (cm, eps, sig, epsp, hsv, dt1, capa, etype, tt,
     1 temper, failel, crv, nnpcrv, cma, qmat, elsiz, idele, reject)
      include 'nlgparm'
      include 'bk06.inc'
      include 'iounits.inc'
      dimension cm(*),eps(*),sig(*),hsv(*),crv(lq1,2,*),cma(*)
      character*5 etype
      logical failel
C --
      real S(6), defgrad(3,3), green(3,3), detF, g, g2
      real p, davg, lam, sigold, epsold, tol, deps
      integer iter, limiter
      tol=1.E-7
      limiter=10
C - initial output
      if (ncycle.le.1) then
        if (cm(2).ge.5.00000E-01) then
           write(iomsg,*) 'mat43 --- illegal possions number,',cm(2)
           cerdat(1)='Illegal Poissons number'
           call lsmsg(3,MSG SOL+1447,ioall,ierdat,rerdat,cerdat,0)
```

```
endif
        if (idele.eq.int (cm(5))) then
          write(iomsq,*) 'mat43 --- my hyperelastic code. E=',cm(1),
     1
            'nu=',cm(2),'capa=',capa
        endif
      endif
C -- Material parameters
      g2 = .5*abs(cm(1))/(1.+cm(2))
      lam = g2*cm(2)/(1.-2.*cm(2))
```

This part starts with the declaration of the subroutine, as described in Section 4.3, then some variable declarations follow. The local variables of the subroutine are also declared, for example S (6) is an array for storing the 2<sup>nd</sup> Piola-Kirchhoff stress tensor, defgrad (3, 3) is an array for storing the deformation gradient F, and detF for its determinant. The variable g2 is the shear modulus G or  $\mu$ , and lam is  $\lambda/2$ . Some initial checking of input parameters is done, and if cm(5) gives an element ID, also a message will be printed in the mes0\* - files, to confirm that the umat43 is active.

The next part of the subroutine performs the stress update for solid elements. It starts with storing the deformation gradient in the matrix defgrad from the hsv array and computing its determinant.

```
C -- for solids
      if (etype.eq.'solid') then
C -- extract deformation gradient
        defgrad(1,1) = hsv(1)
        defgrad(2,1) = hsv(2)
        defgrad(3,1) = hsv(3)
        defgrad(1,2) = hsv(4)
        defgrad(2,2) = hsv(5)
        defgrad(3,2) = hsv(6)
        defgrad(1,3) = hsv(7)
        defgrad(2,3) = hsv(8)
        defgrad(3,3) = hsv(9)
C ---
        detF = defgrad(1,1)*defgrad(2,2)*defgrad(3,3)+
     defgrad (1,2) *defgrad (2,3) *defgrad (3,1) +
     2
       defgrad(1,3)*defgrad(2,1)*defgrad(3,2)-
     3 \quad defgrad(1,3) * defgrad(2,2) * defgrad(3,1) -
     4 defgrad(1,2)*defgrad(2,1)*defgrad(3,2)-
         defgrad(1,1)*defgrad(2,3)*defgrad(3,2)
```

In the final part of the stress update for solid elements, the Green-Lagrange (or to be exact 2E) strain is computed, and based on this the 2<sup>nd</sup> Piola-Kirchhoff stress, which is finally transformed to Cauchy stress via a push-forward operation followed by division by det F.

```
C -- compute 2*Green strain
        do j=1,3
          do i=1,j
            green(i,j) = sum(defgrad(:,i)*defgrad(:,j))
            green(j,i) = green(i,j)
          enddo
        enddo
        green(1,1) = green(1,1) - 1
        green(2,2) = green(2,2) - 1
```



```
green(3,3) = green(3,3) - 1
         davg=lam*(green(1,1)+green(2,2)+green(3,3))
C -- Piola-Kirchhoff
         S(1) = g2*green(1,1) + davg
         S(2) = q2 * qreen(2, 2) + davq
         S(3) = g2*green(3,3) + davg
         S(4) = g2*green(1,2)
         S(5) = g2*green(2,3)
         S(6) = g2 * green(1,3)
C --- push forward for Cauchy stress
        call push forward 2s(S(1),S(2),S(3),S(4),S(5),S(6),hsv(1),
        hsv(2), hsv(3), hsv(4), hsv(5), hsv(6), hsv(7), hsv(8), hsv(9))
         sig(1:6) = S/detF
```

where the symmetry of E has been utilized in the nested do - loops. This concludes the stress update for solid elements. The strain and stress computations for shell elements are much more involved, since the condition  $\sigma_{33} = 0$  (sig (3) = 0) needs to be fulfilled. An iterative procedure, following the "Sample user subroutine 45" of Ref. [1] Appendix A, is applied. The secant method is used, which requires two starting guesses. The first one is given by plane stress elasticity, with

$$\epsilon_{33} = -\frac{v}{1-v}(\epsilon_{11} + \epsilon_{22})$$

and the second staring guess is simply  $\epsilon_{33} = 0$ . The Fortran code follows:

```
else if (etype.eq.'shell') then
C --- for shells
C --- secant iterations for zero z-stress, find eps(3)
      deps = 0.
      do iter=1,limiter
C first thickness strain increment initial guess
c assuming Poisson's ratio different from zero
С
        if (iter.eq.1) then
          eps(3) = -cm(2) * (eps(1) + eps(2)) / (1.-cm(2))
c second thickness strain increment initial guess
        else if (iter.eq.2) then
          sigold=sig(3)
          epsold=eps(3)
          eps(3) = 0.
c --- secant update of thickness strain increment
С
        else if (abs(sig(3)-sigold).gt.0.0) then
          deps=-(eps(3)-epsold)/(sig(3)-sigold)*sig(3)
          sigold=sig(3)
          epsold=eps(3)
          eps(3) = eps(3) + deps
        endif
c --- update last row of F
        call compute f3s(hsv(3), hsv(6), hsv(9), eps(3))
```

After that follows computation of the Green-Lagrange strain and 2<sup>nd</sup> Piola-Kirchhoff stress, just as for solids:

```
C --- compute strain and stress
        defgrad(1,1) = hsv(1)
        defgrad(2,1) = hsv(2)
        defgrad(3,1) = hsv(3)
        defgrad(1,2) = hsv(4)
        defgrad(2,2) = hsv(5)
        defgrad(3,2) = hsv(6)
        defgrad(1,3) = hsv(7)
        defgrad(2,3) = hsv(8)
        defgrad(3,3) = hsv(9)
        detF = defgrad(1,1)*defgrad(2,2)*defgrad(3,3)+
       defgrad(1,2)*defgrad(2,3)*defgrad(3,1)+
     1
        defgrad(1,3)*defgrad(2,1)*defgrad(3,2)-
     3 \quad defgrad(1,3) * defgrad(2,2) * defgrad(3,1) -
     4 defgrad(1,2)*defgrad(2,1)*defgrad(3,2)-
         defgrad(1,1)*defgrad(2,3)*defgrad(3,2)
C -- compute Green strain
        do j=1,3
          do i=1,j
            green(i,j) = sum(defgrad(:,i)*defgrad(:,j))
            green(j,i) = green(i,j)
          enddo
        green(1,1) = green(1,1) - 1
        green(2,2) = green(2,2) - 1
        green(3,3) = green(3,3) - 1
        davg=lam*(green(1,1)+green(2,2)+green(3,3))
C -- Piola-Kirchhoff stress
        S(1) = g2 * green(1, 1) + davg
        S(2) = g2*green(2,2) + davg
        S(3) = q2 * qreen(3, 3) + davq
        S(4) = g2*green(1,2)
        S(5) = q2 * green(2,3)
        S(6) = g2 * green(1,3)
```

The final transformation to Cauchy stress differs slightly, since the shear stress components  $\sigma_{23}$ ,  $\sigma_{13}$ (sig(5:6)) are multiplied by capa, the reduction factor for transverse shear in shells (corresponding to SHRF of the \*SECTION SHELL - keyword). Also, in order to save some floating-point operations, the update of these stress components is moved outside the secant iterations do – loop. The final part of the code for shell elements follows:

```
C --- push forward to get Cauchy stress
        call push forward 2s(S(1),S(2),S(3),S(4),S(5),S(6),hsv(1),
     1 hsv(2), hsv(3), hsv(4), hsv(5), hsv(6), hsv(7), hsv(8), hsv(9))
        sig(1:4) = S(1:4)/detF
C --- termination criteria
        if(abs(sig(3)).lt.tol*(abs(sig(1))+abs(sig(2))+abs(sig(4))))
     1
          exit
      enddo
      sig(5:6) = capa*S(5:6)/detF
```

Finally, an error message will be issued in case attempts are made to apply this user defined material model to other element types than solids or shells:

```
else
 cerdat(1)=etype
  call lsmsg(3,MSG SOL+1151,ioall,ierdat,rerdat,cerdat,0)
endif
return
end
```

which concludes the user defined material subroutine. For implicit analysis, also the tangent modulus must be computed. In this case, a push-forward of  $\mathbb C$  is (more or less) what is required. The user subroutine utan43 is used to implement the tangent modulus in Fortran code. The first part of the subroutine, which starts with the subroutine declaration according to Section 4.3.1, and some variable declarations follows:

```
subroutine utan43(cm,eps,sig,epsp,hsv,dt1,unsym,capa,etype,tt,
     1 temper, es, crv, nnpcrv, failel, cma, qmat)
С
      include 'nlqparm'
      dimension cm(*),eps(*),sig(*),hsv(*),crv(lq1,2,*),cma(*)
      integer nnpcrv(*)
      dimension es(6,*), qmat(3,3)
      logical failel, unsym
      character*5 etype
      real*8 f1, f2, f3, defgrad(3,3), detF, detFinv
      real*8 dmx(6,6)
```

Just as for umat 43, the deformation gradient F is stored in the defgrad matrix, and the determinant is computed:

```
С
     defgrad(1,1) = hsv(1)
     defgrad(2,1) = hsv(2)
     defgrad(3,1) = hsv(3)
     defgrad(1,2) = hsv(4)
     defgrad(2,2) = hsv(5)
     defgrad(3,2) = hsv(6)
     defgrad(1,3) = hsv(7)
     defgrad(2,3) = hsv(8)
     defgrad(3,3) = hsv(9)
     detF = defgrad(1,1)*defgrad(2,2)*defgrad(3,3)+
    1
        defgrad(1,2)*defgrad(2,3)*defgrad(3,1)+
    2
       defgrad(1,3)*defgrad(2,1)*defgrad(3,2)-
    3 defgrad(1,3)*defgrad(2,2)*defgrad(3,1)-
    4 defgrad(1,2)*defgrad(2,1)*defgrad(3,2)-
    5 defgrad(1,1)*defgrad(2,3)*defgrad(3,2)
     detFinv = 1.0 / max(detF, 1.E-12)
```

In the last line, some extra caution is taken when computing 1/det F, in order to avoid division by zero. The next step is to populate the  $\mathtt{dmx}$  – matrix, which holds the stiffness tensor  $\mathbb C$  (in the reference configuration). This is achieved by the following lines of code:

```
f1 = cm(1)/(1.0+cm(2))/(1.0-2.0*cm(2))
```



```
f2 = 1.0 - cm(2)
      f3 = 0.5 - cm(2)
      dmx = 0
      dmx(1,1) = f1*f2
      dmx(2,2) = dmx(1,1)
      dmx(3,3) = dmx(1,1)
      dmx(1,2) = f1*cm(2)
      dmx(1,3) = dmx(1,2)
      dmx(2,3) = dmx(1,2)
      dmx(2,1) = dmx(1,2)
      dmx(3,1) = dmx(1,3)
      dmx(3,2) = dmx(2,3)
C ---
      dmx(4,4) = f1*f3
      dmx(5,5) = dmx(4,4)
      dmx(6,6) = dmx(4,4)
```

The transformation to the current configuration is done by

```
C --- push forward
      call push forward 4s(dmx, hsv(1),
     1 hsv(2), hsv(3), hsv(4), hsv(5), hsv(6), hsv(7), hsv(8), hsv(9))
      es(1:6, 1:6) = dmx * detFinv
```

In addition, in order to account for the transverse shear reduction capa for shells the modification

```
c --- special for shell
      if (etype.eq.'shell') then
        es(5,5) = capa * dmx(5,5) * detFinv
        es(6,6) = capa * dmx(6,6) * detFinv
      endif
      return
      end
```

is made, which also concludes the tangent modulus subroutine.

#### 4.5.2 J<sub>2</sub>-plasticity for solids and shells

In this Section, the implementation of a hypoelastic-plastic material model with isotropic hardening for solids and shells is described. For a detailed theoretical background, see for example Ref. [11] and especially Section 17.4.1 for details on the derivation. The yield condition is

$$f(\mathbf{\sigma}, \overline{\epsilon}_p) = \sigma_{vM} - \sigma_y(\overline{\epsilon}_p) = 0$$

where  $\overline{\epsilon}_p$  is the accumulated effective plastic strain, and  $\sigma_{vM}$  is the von Mises effective stress, which is directly proportional to the norm of the stress deviator s given by

$$s_{ij} = \sigma_{ij} - \delta_{ij} \frac{\sigma_{kk}}{3}$$

The von Mises effective stress is then given by

$$\sigma_{vM} = \sqrt{\frac{3}{2} s_{ij} s_{ij}}$$



The hardening function  $\sigma_v(\overline{\epsilon}_v)$  is often given as a piecewise linear curve, determined from material testing.

An efficient implementation of  $J_2$  – plasticity is obtained by the radial return algorithm [4] [7] [11][21]. The starting point is the stress and strain at state 1, and the objective is to compute stress and increment of effective plastic strain at stat 2, given the strain increment  $\Delta \epsilon_{ij}$ . First, an elastic trial stress is computed according to

$$\mathbf{\sigma}^t = \mathbf{\sigma}^{(1)} + \mathbb{C}: \Delta \mathbf{\epsilon}$$

where  $\mathbb C$  is the isotropic elastic stiffness tensor. From the elastic trial stress, the von Mises effective stress is calculated as

$$\sigma_{vM}^t = \sqrt{\frac{3}{2} s_{ij}^t s_{ij}^t}$$

If the effective trial stress is below the yield limit,  $f(\sigma^t, \overline{\epsilon}_p) \leq 0$ , the elastic trial stress is accepted, and no further action is needed. In case yielding is indicated, the increment in effective plastic strain  $\Delta \bar{\epsilon}_v$  to satisfy the yield criterion must be determined. For solid elements, this can be done, following Box 17.5 of Ref. [11], by solving

$$f_p = \sigma_{vM}^t - 3G\Delta\overline{\epsilon}_p - \sigma_y \left(\overline{\epsilon}_p^{(1)} + \Delta\overline{\epsilon}_p\right) = 0$$

using Newton's method, that is

- 1. Set  $\Delta \overline{\epsilon}_n^1 = 0$
- 2. Compute  $\Delta \overline{\epsilon}_p^{k+1} = \Delta \overline{\epsilon}_p^k \frac{f_p^k}{df_p^k} = \Delta \overline{\epsilon}_p^k + \frac{f_p^k}{3G+H}$  where  $H = \frac{d\sigma_y(\overline{\epsilon}_p^k)}{d\overline{\epsilon}_p^k}$
- 3. Compute  $f_n^{k+1}$
- 4. If  $|f_{v}^{k+1}| > tol$  then let k = k + 1 and go to 2.
- 5. Update stress and accumulated effective plastic strain:

$$\overline{\epsilon}_{p}^{(2)} = \overline{\epsilon}_{p}^{(1)} + \Delta \overline{\epsilon}_{p} 
s_{ij}^{(2)} = \frac{\sigma_{y}^{(2)}}{\sigma_{vM}^{t}} s_{ij}^{t} 
\sigma_{ij}^{(2)} = s_{ij}^{(2)} + \frac{1}{3} \sigma_{kk}^{t} \delta_{ij}$$

In the case of linear hardening, where H = constant, this procedure will converge exactly in one iteration. From step 2 of the Newton scheme, it can also be noted that the slope of the hardening curve in practice will be limited by

$$3G + H > 0 \Leftrightarrow H > -3G$$

which means that from a mathematical viewpoint, a negative slope of the hardening curve can be tolerated, as long as it fulfills this condition.

For shell elements, some additional modifications are needed to fulfill  $\sigma_{33}=0$  and account for the reduction factor for transverse shear. This will be discussed in more detail in the context of the Fortran coding for shell elements below.

The keyword interface for this material model should allow for input of elastic constants (Young's modulus E and Poisson's ration  $\nu$ ) and a hardening curve. In addition, the user is given some control over the Newton iterations outlined above, with respect to the required tolerance and the maximum allowed number of iterations. Since epsp already is a separate parameter in the interface to the user subroutine, see Table 2, no history variables are, strictly speaking, required in this case. Still, one history variable will be used to indicate if yielding takes place or not. This will be used by the tangent stiffness routine utan later on. In all, this means that the keyword interface for the  $J_2$  - plasticity model will be

```
*MAT USER DEFINED MATERIAL MODELS TITLE
J2 plasticity by UMAT41
     MID
                                                 IORTHO
$#1
                RO
                         MT
                                 LMC
                                          NHV
                                                           IBULK
                                                                       ΙG
  mat id
               \rho
                         41
                                  8
                                                             3
                                                                       4
$#2 IVECT
            IFAIL
                     ITHERM
                              IHYPER
                                         IEOS
                                                  LMCA
                                                          UNUSED
                                                                   UNUSED
               P2
                         Р3
                                  Ρ4
                                           P5
                                                    Р6
                                                             P7
                                                                       Р8
     P1
  Young's Poisson's
                                  G
                                         LCID
                                                    tol
                                                                    el.ID
                         K
                                                         limiter
```

where blue text indicates that the user should input sensical data, and red text indicates values that should not be changed (since these fixed values also will be assumed by the Fortran implementation). The curve ID of the hardening curve (\*DEFINE CURVE) is input as P5. The tolerance for the Newton iterations is optionally input as P6 and the maximum number of allowed iterations is optionally input as P7. For additional debug output, an element ID may be specified as P8.

It shall be mentioned that the LS-DYNA usermat package already contains the subroutines metalshl, metalsld and metaltan (in dyn21umats.f and dyn21utan.f) which are ready-to-use subroutines for J2 - plasticity (but have no Newton iterations for the radial return).

The user subroutine umat41 will be used to implement the material model in Fortran code. The first part of the subroutine follows:

```
subroutine umat41 (cm, eps, sig, epsp, hsv, dt1, capa, etype, tt,
    1 temper, failel, crv, nnpcrv, cma, qmat, elsiz, idele, reject)
С
Livermore Software Technology Corporation (LSTC)
Сl
Сl
   Copyright 1987-2008 Livermore Software Tech. Corp
   All rights reserved
Сl
С
    isotropic elastic-plastic material
С
С
    Variables
С
С
    cm(1)=first material constant, here young's modulus
С
    cm(2)=second material constant, here poisson's ratio
С
С
С
С
С
    cm(n)=nth material constant
```

Here, many lines of detailed comments of the original Fortran file are omitted. The comments of the files of the usermat package in general provide valuable information in connection with the respective subroutine.

After this follows some variable declarations and initializations. The array sshl(6) will hold temporary values of the stress tensor. The arrays s (6) and s2 (6) will be used for stress deviator values. The von Mises effective stress will be stored in the variable vonMises. The history variable hsv (1) will indicate if yielding takes place or not, as a way of communicating this to the utan41 subroutine.

```
include 'nlqparm'
      include 'bk06.inc'
      include 'iounits.inc'
      dimension cm(*), eps(*), sig(*), hsv(*), crv(lq1, 2, *), cma(*), qmat(3, 3)
      integer nnpcrv(*)
      logical failel, reject
      character*5 etype
      integer idele
С
      real vonMises, h, sigy, s(6), depsp, gc, tol, ep tr
      real sig3e, sig3p, eps3e, eps3p, sshl(6), f1, f2
      real s2(6),p2,f3, strainlim
      integer iter, limiter
```

Then tolerances (tol) and iteration limit (limiter) are initialized, with their default values, but if the user provides reasonable input in cm(6) and cm(7), it replaces the default values. The strainlim is a (presently hard-coded) limit on the increment of accumulated effective plastic strain ( $\Delta \bar{\epsilon}_n$  above); for implicit, a reject will be issued in case this limit is exceeded. After that, an initial message is written to the mes 0 \* - files, to confirm that the umat 41 is active. Finally, some material constants are computed, and the yield limit for the current value of effective plastic strains is evaluated by crvval.

```
tol=1.E-4
      limiter=10
     if(cm(6).gt.0.)
     1 tol=cm(6)
     if(cm(7).gt.0.)
     1 limiter = int(cm(7))
      strainlim = 5.E-2
С
      if (ncycle.le.1) then
        if(idele.eq.int(cm(8)))then
          write(iomsg,*) 'mat41:iterative elastoplastic code. E=',cm(1),
            'nu=',cm(2),'lcid=',cm(5)
          write(iomsg,*) 'mat41:limiter =',limiter,'tol=',tol
        endif
      endif
C
      compute shear modulus, g
С
c E is cm(1), pr is cm(2)
      q2 = abs(cm(1))/(1.+cm(2))
      g = .5*g2
      gc =capa*g
```

```
hsv(1) = 0.
C -- Yield curve val and tangent h : starting vals
      call crvval(crv,nnpcrv,cm(5),epsp,sigy,h)
Next follows the implementation for solid elements, starting with computation of the
elastic trial stress:
      if (etype.eq.'solid') then
C
С
      --- For solids (iterative) ---
      Compute elastic trial stress
С
С
        davg=-sum(eps(1:3))/3.
        p=-davg*abs(cm(1))/(1.-2.*cm(2))
        sig(1:3) = sig(1:3) + p + g2*(eps(1:3) + davg)
        sig(4:6) = sig(4:6) + g*eps(4:6)
C --- Effective stress, first the deviatoric
        s(1:3) = sig(1:3) - sum(sig(1:3))/3.
        s(4:6) = sig(4:6)
      compute the von Mises stress
С
        vonMises = s(1)**2+s(2)**2+s(3)**2+2.*(s(4)**2+s(5)**2+s(6)**2)
        vonMises = sqrt(1.5*vonMises)
        if (vonMises.le.sigy) then
          return
        else
          hsv(1) = 1.
        endif
```

At this point, the subroutine will return in case of an elastic response. In the following, the iterative radial return scheme outlined above is implemented. In the convergence check, the tolerance is multiplied by the current yield stress to obtain a relative measure. The iterations will be aborted if the convergence criterion is met, otherwise limiter iterations will be performed, and a message is printed. Also, an extra message is written for user feedback.

```
C - radial return, iterative
      ep tr=epsp
      depsp = 0.
      f1 = vonMises - sigy
      do iter=1, limiter
        depsp = depsp + f1/(h+3.*g)
      re-eval hardening curve
C
        ep tr = epsp + depsp
        call crvval(crv,nnpcrv,cm(5),ep tr,sigy,h)
        f1 = vonMises - 3.*q*depsp - sigy
        if(abs(f1).lt.tol*sigy)
         exit
     1
      enddo
C --- debug
      if (iter.ge.limiter) then
        write(iomsg,*) 'mat41:iter=',iter,'idele=',idele,
           'stressdiff=',abs(sigy-vonMises)
      endif
      if (idele.eq.int(cm(8))) then
        write(iomsg,*) 'mat41:iter=',iter,'idele=',idele,
           'stressdiff=',abs(sigy-vonMises),'f1*sigy=',f1*sigy,
     1
           'depsp=',depsp
```

Finally, the stress is updated, and this concludes the part of the subroutine for solid elements. The check for the increment in effective plastic strain is performed after the stress update, since the reject option is only active for implicit analysis.

```
epsp = ep tr
s2 = sigy*s/vonMises
p2 = sum(sig(1:3))/3.
sig(1:3) = s2(1:3) + p2
sig(4:6) = s2(4:6)
if (depsp.gt.strainlim) then
  reject = .true.
endif
```

Then comes the part of the implementation for shell elements. Due to the requirement of zero normal stress, this becomes more involved. In addition, considering the reduction factor for transverse shear capa also complicates the calculations. The present approach uses a two-step scheme, where first the through-thickness strain  $\varepsilon_{33}$  is estimated by linear interpolation between two extrema. Then, the stress state is determined using a slightly different approach for the radial return algorithm based on the estimate of £33. This solution approach for shells is adopted from the subroutine metalshl already present in the usermat package.

First, the elastic trial stress is computed, and yielding is checked.

```
elseif(etype.eq.'shell')then
С
С
      --- For shells (iterative) ---
С
      Compute elastic trial stress and eps3
С
        eps(3) = -cm(2) * (eps(1) + eps(2)) / (1.-cm(2))
        davg=-sum(eps(1:3))/3.
        p=-davg*abs(cm(1))/(1.-2.*cm(2))
        sshl(1:2) = sig(1:2) + p + g2*(eps(1:2) + davg)
        sshl(3) = 0.
        sshl(4) = sig(4) + g*eps(4)
        sshl(5:6) = sig(5:6) + gc*eps(5:6)
C --- Effective stress, first the deviatoric
        s(1:3) = sshl(1:3) - sum(sshl(1:2))/3.
        s(4:6) = sshl(4:6)
      compute the von Mises stress
        vonMises = s(1)**2+s(2)**2+s(3)**2+2.*(s(4)**2+s(5)**2+s(6)**2)
        vonMises = sqrt(1.5*vonMises)
C -- check yield
        if (vonMises.le.sigy) then
          sig(1:6) = sshl(1:6)
          return
        endif
```

Again, if the response is elastic the subroutine returns. In the following, the required updates of stresses, effective plastic strain and through-thickness strain due to the plastic deformation are performed. The first step is to estimate the through-thickness strain. This is done from linear interpolation between the first elastic estimate, and a fully plastic estimate.

```
C --- secant iterations for eps(3) estimate
      radial return from elastic
        f1=4.5*q*(capa-1.)*(sshl(5)**2+sshl(6)**2)/(vonMises**2)
        depsp = (vonMises - sigy) / (h+3.*g+f1)
        sig3e =
                        - 3.*q*depsp*s(3)/vonMises
        eps3e = eps(3)
C --- first point is (eps3e, sig3e)
      second point, radial return from plastic
        eps3p = -eps(1) - eps(2)
        sshl(1:2) = sig(1:2) + g2*eps(1:2)
        sshl(3) = q2*eps3p
C --- Effective stress, first the deviatoric
        s(1:3) = sshl(1:3) - sum(sshl(1:3))/3.
      compute the von Mises stress
        vonMises = s(1)**2+s(2)**2+s(3)**2+2.*(s(4)**2+s(5)**2+s(6)**2)
        vonMises = sqrt(1.5*vonMises)
C -- check yield, radial return if required
        if (vonMises.ge.sigy) then
          f1=4.5*q*(capa-1.)*(sshl(5)**2+sshl(6)**2)/(vonMises**2)
          depsp = (vonMises - sigy)/(h+3.*g+f1)
          sshl(3) = sshl(3) - 3.*g*depsp*s(3)/vonMises
        endif
C --- second point is (eps3p, sshl(3))
      linear interpolation between p.1 and p.2
       if (abs(sig3e-sshl(3)).gt.tol*
            \max(abs(sig3e),abs(sig3p))) then
         eps(3) = eps3e - sig3e * (eps3e - eps3p) / (sig3e - sshl(3))
       else
         eps(3) = eps3p
       endif
```

In this case, the variable £1 holds extra terms of  $\frac{d\sigma_{\nu M}}{d\Delta \overline{\epsilon}_p}$  due to the transverse shear reduction factor. It vanishes in the case capa = 1. After this, it assumed that  $\epsilon_{33}$  is known, and it remains to determine the corresponding stress state. This starts with again computing an elastic trial stress:

```
С
С
      now we have estimate for eps(3): update stresses
С
       davg=-sum(eps(1:3))/3.
       p=-davg*abs(cm(1))/(1.-2.*cm(2))
       sig(1:2) = sig(1:2) + p + g2*(eps(1:2) + davg)
       sig(3)=0
       sig(4) = sig(4) + g*eps(4)
       sig(5:6) = sig(5:6) + gc*eps(5:6)
C --- Effective stress, first the deviatoric
       s(1:3) = sig(1:3) - sum(sig(1:3))/3.
       s(4:6) = sig(4:6)
      compute the von Mises stress
С
       vonMises = s(1)**2+s(2)**2+s(3)**2+2.*(s(4)**2+s(5)**2+s(6)**2)
       vonMises = sqrt(1.5*vonMises)
       if (vonMises.le.sigy) then
         return
       else
          hsv(1) = 1.
```

```
endif
ep_tr=epsp
depsp = 0.
f2 = 3.*g/vonMises
f3 = f2*capa
```

Then, as the final step for shell elements, iterative radial return is performed. For shells, a different approach than for solids is used, where the stresses are updated in each iteration. This makes it easier to account also for the transverse shear reduction factor. If the limiter iterations are not enough to reach the specified tolerance, a message is printed.

```
do iter=1,limiter
C --- radial return
        f1=4.5*g*(capa-1.)*(sig(5)**2+sig(6)**2)/(vonMises**2)
        depsp = depsp + (vonMises - sigy)/(h+3.*g+f1)
        sshl(1:2) = sig(1:2) - f2*depsp*s(1:2)
        sshl(3) = 0
        sshl(4) = sig(4) - f2*depsp*s(4)
        sshl(5:6) = sig(5:6) - f3*depsp*s(5:6)
c -- compute the von Mises stress
        s2(1:3) = sshl(1:3) - sum(sshl(1:3))/3.
        vonMises = sum(s2(1:3)**2)+2.*sum(sshl(4:6)**2)
        vonMises = sqrt(1.5*vonMises)
С
      re-eval hardening curve
        ep tr = epsp + depsp
        call crvval(crv,nnpcrv,cm(5),ep tr,sigy,h)
C --- check convergence
        if(abs(sigy-vonMises).lt.tol*abs(sigy))
     1
      enddo
C --- debug
      if (iter.ge.limiter) then
        write(iomsg,*) 'mat41:iter=',iter,'idele=',idele,
           'stressdiff=',abs(sigy-vonMises)
      endif
      sig(1:6) = sshl(1:6)
      epsp = ep_tr
      if (depsp.gt.strainlim) then
        reject = .true.
      endif
```

Most likely, the above implementation for shell element can be improved, both for increased efficiency and accuracy. The subroutine ends with printing an error message, in case attempts are made to apply the material model to other element types than solids or shells.

For implicit analysis, also the tangent modulus is required. The user subroutine utan41 is used to implement the tangent modulus in Fortran code. The implementation for solids is based on the results presented in Section 12.2 of Ref. [11], that is

$$C_{ijkl}^{ep} = C_{ijkl} - \frac{9G^2}{H + 3G} \frac{s_{ij}s_{kl}}{\sigma_{vM}^2}$$

For shells, an approximate tangent modulus is applied: basically, the same as for solid elements is used, only some minor corrections for the transverse shear reduction are made. The first part of the subroutine, which starts with the subroutine declaration according to Section 4.3.1, and some variable declarations follows:

```
subroutine utan41 (cm, eps, sig, epsp, hsv, dt1, unsym, capa, etype, tt,
    1 temper, es, crv, nnpcrv, failel, cma, qmat)
С
Livermore Software Technology Corporation (LSTC)
C \mid
   Copyright 1987-2008 Livermore Software Tech. Corp
c| All rights reserved
                    *************
С
     include 'nlqparm'
     include 'bk06.inc'
     include 'iounits.inc'
     dimension cm(*),eps(*),sig(*),hsv(*),crv(lq1,2,*),cma(*)
     integer nnpcrv(*)
     dimension es(6,*), qmat(3,3)
     logical failel, unsym
     character*5 etype
С
     real*8 factor, g,b, bg23,bg43
     real*8 vonMises, h, sigy, s(6), sf(6), depsp, gc, tol, ep tr
     real*8 sig3e, sig3p, eps3e, eps3p, sshl(6), f1
     real*8 dpfac, A, shrf
     integer k, l
С
     factor=1.
     if (failel) factor=1.e-8
```

This section ends with a stiffness reduction for elements that are indicated as failed. Then, some elastic constants are computed, element type is checked, and an attempt to make a small modification for shells is made:

```
g = factor*.5*abs(cm(1))/(1.+cm(2))
b=factor*abs(cm(1))/3./(1.-2.*cm(2))
bg23=b-2.*g/3.
bq43=b+4.*q/3.
if (etype.eq.'solid') then
  shrf=1.
elseif (etype.eq.'shell') then
  shrf=capa
else
  cerdat(1) = etype
  call lsmsq(3,MSG SOL+1151,ioall,ierdat,rerdat,cerdat,0)
  return
endif
```

This means that if the routine is applied to other element types than solids or shells, LS-DYNA will stop with an error message. The next step is the elastic part of the tangent stiffness modulus:

```
es(1,1) = bq43
es(2,2) = bq43
es(3,3) = bg43
es(2,1) = bg23
es(3,1) = bq23
es(3,2) = bg23
es(1,2) = es(2,1)
es(1,3) = es(3,1)
es(2,3) = es(3,2)
es(4,4)=q
es(5,5)=g*shrf
es(6,6)=q*shrf
```

and finally, the plastic part, which only is required in case the material routine umat 41 indicated that yielding takes place, by setting hsv(1) = 1.

```
if (hsv(1).gt.0.) then
  call crvval(crv,nnpcrv,cm(5),epsp,siqy,h)
  s(1:3) = sig(1:3) - sum(sig(1:3))/3.
  s(4:6) = sig(4:6)
  vonMises =s(1)**2+s(2)**2+s(3)**2+2.*(s(4)**2+s(5)**2+s(6)**2)
  vonMises = sqrt(1.5*vonMises)
  A = h + 3.*q
  dpfac = (9.*g**2)/A/(vonMises**2)
  sf = dpfac*s
  do k=1,6
    do l=k, 6
      es(k, 1) = es(k, 1) - sf(k) *s(1)
      es(l,k) = es(k,l)
    enddo
  enddo
endif
```

In this implementation, the symmetry of the tangent stiffness matrix has been utilized in the reduced inner DO - loop. This concludes the subroutine utan41 for the tangent modulus. Clearly, since the tangent is based on the implementation for solid elements, convergence properties for shell elements are not as good.

#### 4.5.3 Non-linear spring

This section describes the implementation of a simple material for discrete beam elements (ELFORM = 6 on \*SECTION BEAM) representing a (optionally non-linear) spring. The material can either represent a linear relationship between force and beam elongation,

$$F_r = k\delta$$

or a non-linear relationship given by a curve,

$$F_r = f(\delta)$$
.



In case the curve is only defined for positive elongation, the material model will use

```
F_r = f(|\delta|) sign(\delta).
```

This material model is similar to a mix of the built-in materials \*MAT\_LINEAR\_ELASTIC\_DISCRETE\_BEAM (MAT\_66) and \*MAT\_NONLINEAR\_ELASTIC\_DISCRETE\_BEAM (MAT\_67).

The keyword interface for this material model should allow for input of either a constant stiffness k or a curve ID. This is done by the parameter P1: if P1 > 0, it is assumed to be a constant stiffness value, and if P1 < 0, it is assumed that the curve ID is |P1|. The user should also provide reasonable values for K and G, which are important for calculating the explicit time step.

where blue text indicates that the user should input sensical data, and red text indicates values that should not be changed (since these fixed values also will be assumed by the Fortran implementation).

The user subroutine umat44 will be used to implement the material model in Fortran code. The first part of the subroutine follows:

```
subroutine umat44 (cm,eps,sig,epsp,hsv,dt1,capa,etype,tt,
    1 temper, failel, crv, nnpcrv, cma, qmat, elsiz, idele, reject)
C
Livermore Software Technology Corporation (LSTC)
Сl
   Copyright 1987-2008 Livermore Software Tech. Corp
   All rights reserved
C.
     include 'nlqparm'
     include 'bk06.inc'
     include 'iounits.inc'
     dimension cm(*), eps(*), sig(*), hsv(*), crv(lq1, 2, *), cma(*), qmat(3, 3)
     integer nnpcrv(*)
     character*5 etype
     logical failel, reject
    integer*8 idele
C
     real*8 stiff, lcid, yfval
     integer*8 iid
```

The first rows are the standard declarations for a user-defined material subroutine. After that the declaration of the model-specific local variables follow. The variable stiff holds the constant stiffness value, the variable lcid holds the curve ID and yfval is the current force at the current elongation, evaluated from the curve. The integer variable iid is the interval ID of the curve.

```
if (ncycle.eq.1) then
   write(iomsg, *) 'User defined mat44 for discrete beams'
   if(cm(1)>0.d0)then
     write(iomsg, *) '---discrete beam found.Constant stiffness'
     write(iomsg, *) ' useing k factor:',cm(1)
   else
    write(iomsg, *) '---discrete beam found.Curve'
    lcid=abs(cm(1))
    iid=lcids(nint(lcid))
    hsv(4) = crv(1, 1, iid)
    write(iomsq, *) ' using Curve ID:', nint(abs(cm(1)))
     if(hsv(4).lt.0.d0)then
      write(iomsg, *) ' curve exists also for negative x vals'
    else
      write(iomsg, *) ' reflected curve will be used for ',
1
             'negative x vals'
    endif
   endif
 endif
```

Then follows some initial checks and output of messages to the user, performed at cycle 1. The main check is if a constant stiffness or a curve ID is to be used, and if the curve also is defined for negative (compressive) elongation. This latter check is done by directly inspecting the first ordinate value of the curve, crv(1,1,iid), and storing it to history variable #4. The internal ID of the curve is found by lcids(nint(lcid)).

Then follows the force calculations, for discrete beams only:

```
if (etype.eq.'dbeam') then
```

In case a constant stiffness is used, the force update is uncomplicated.

```
if(cm(1).ge.0.d0)then
  stiff=cm(1)
  sig(1)=sig(1)+eps(1)*stiff
else
```

If a curve is used, different actions must be taken in case the elongation is negative or compressive. In case of a positive elongation, the curve can be evaluated directly:

```
lcid=abs(cm(1))
hsv(1)=hsv(3)+eps(1)
if(hsv(1).gt.0.d0)then
  call crvval(crv,nnpcrv,lcid,hsv(1),yfval,stiff)
else
```

The current elongation of the beam is stored in history variable #1, based on the previous elongation which is stored in history variable #3. Using history variables for storage of the elongation is also useful if a continuation of analysis using a dynain – file is to be performed.

If the elongation is compressive, we must check if the curve also exists for compression. If so, the curve can be evaluated directly:

```
if(hsv(4).lt.0.d0)then
```



```
call crvval(crv,nnpcrv,lcid,hsv(1),yfval,stiff)
else
```

Otherwise, the absolute value of the elongation is passed to crvval, and the sign of the obtained force is reversed.

```
call crvval(crv,nnpcrv,lcid,abs(hsv(1)),yfval,stiff)
   yfval=-1.d0*yfval
  endif
  endif
  sig(1)=yfval
endif
```

The remaining force components are set to zero. The spring only gives a force in the r – direction, between  $n_1$  and  $n_2$ , see Figure 4.

```
sig(2) = 0.0

sig(3) = 0.0

sig(4) = 0.0

sig(5) = 0.0

sig(6) = 0.0
```

The accumulated elongation is stored in history variable #3, based on the current time tt and the time from the previous call to the routine stored in history variable #2.

```
if(tt.ne.hsv(2))then
   hsv(3)=hsv(1)
endif
hsv(2)=tt
```

In case the user tries to apply the material model to other elements than discrete beams, an error message is issued.

```
else
   cerdat(1) = etype
   call lsmsg(3,MSG_SOL+1150,ioall,ierdat,rerdat,cerdat,0)
endif
return
end
```

Since user-defined material models cannot be used for discrete beam elements in implicit, the computation of a corresponding tangent modulus is not required.

# 4.6 Ansys LS-DYNA simulation examples

In this Section, some simulation examples to demonstrate and verify the user defined material models of Sections 4.5.1 - 4.5.3 are presented. Comparisons to results obtained using the pre-defined material models of LS-DNYA are made. All LS-DYNA keyword files are supplied as attachments to this guide.

### 4.6.1 Examples of the Saint-Venant Kirchhoff material model

The first example applies the hyperelastic material model of umat 45 to an explicit analysis using solid elements. Material parameters typical for steel were used, that is a Young's modulus of 200 GPa and a



Poisson's ratio of 0.5. A ¼ - model of a bar (20x20x100 mm) is subjected to alternating prescribed displacement, see Figure 5. The stress vs. strain in response in one element is shown in Figure 6, compared to results using LS-DYNA's built-in hypoelastic material \*MAT\_ELASTIC and to the hyperelastic material \*MAT\_ORTHOTROPIC\_ELASTIC. For stresses above 1 %, the response from umat45 differ noticeable from \*MAT\_ELASTIC, which also is expected [2], while the agreement to the hyperelastic material \*MAT\_ORTHOTROPIC\_ELASTIC (red dots in Figure 6) is very good throughout the whole strain range.

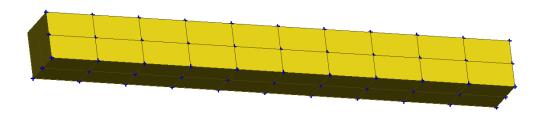


Figure 5. A ½ model of a solid bar. The model consists of 40 solid elements. The blue symbols indicate symmetry boundary conditions applied at the nodes. One short end is fixed, while the other short end is subjected to a pulsating prescribed displacement.

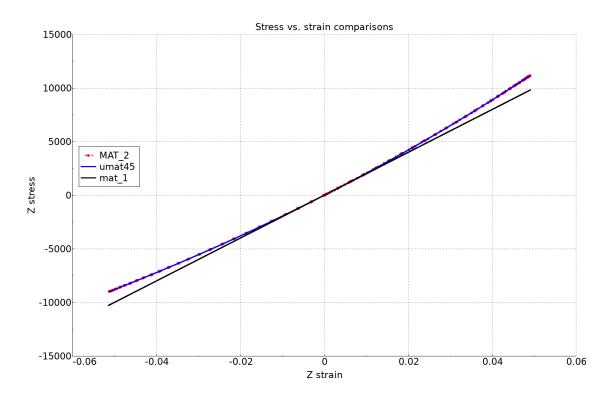


Figure 6. Stress vs. strain response for the hyperelastic Saint-Venant Kirchhoff model compared to the hypoelastic material \*MAT\_ELASTIC and the hyperelastic \*MAT\_ORTHOTROPIC\_ELASTIC, in an explicit analysis.

In the second example, shell elements are used to model a cantilever beam, see Figure 7. Elastic material properties typical for aluminum (E = 70 GPa, v = 0.31) are used. A prescribed displacement of 140 mm is applied at the bolt holes of the end bracket. Contact is considered between the square beam and the cylindrical rigid support, using \*CONTACT AUTOMATIC SINGLE SURFACE MORTAR. Results from implicit analyses using umat45 and \*MAT ELASTIC are compared in Figure 8 and Figure 9. Similar results are obtained. Differences are expected due to the differences in material model formulations. Also, similar performance with respect to iteration count and solution time is obtained for umat45 and \*MAT ELASTIC for the implicit case.

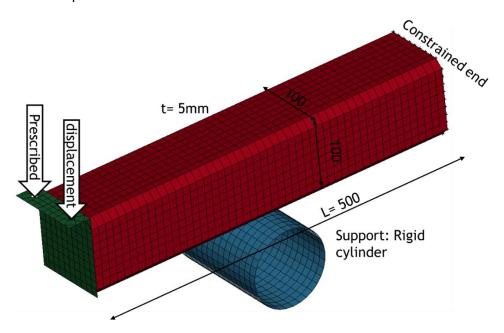


Figure 7. A square (100 × 100 mm, t = 5 mm) cantilever beam is subjected to prescribed displacement at the end bracket (green in the image) and contact with a rigid cylindrical support.

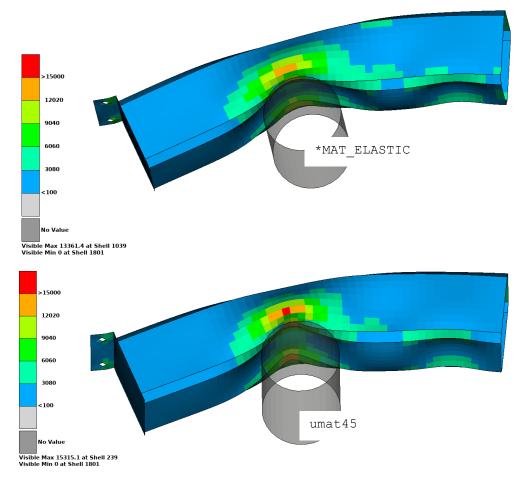


Figure 8. Fringe plot of von Mises effective at 140 mm displacement. The top image shows results using \*MAT ELASTIC, and the bottom image shows results using umat45.

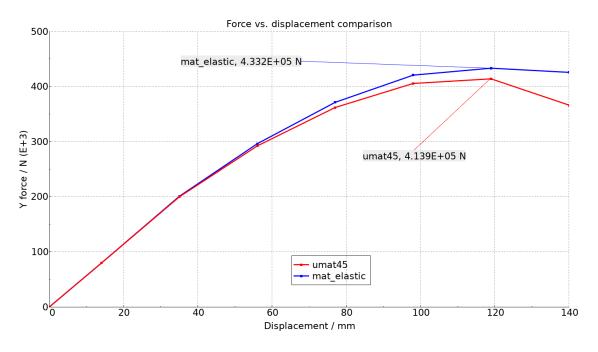


Figure 9. Comparison of global force vs. displacement response using \*MAT ELASTIC (blue curve) and umat 45 (red curve).

### 4.6.2 Examples of the J<sub>2</sub> – plasticity model

The first example is analysis of a tensile specimen, using shell elements, see Figure 10. Analyses were performed both using the explicit and implicit solver of Ansys LS-DYNA. Force vs. displacement results are compared in Figure 11. The peak forces for MAT\_24 and umat 41 are very similar, both for implicit and explicit, while after necking some differences are found. Since necking is a local instability, small differences will be magnified after this point. The results for the explicit analyses are quite close up to about 20 mm of displacement, while the implicit umat 41 results differ more. This may be due to the approximate nature of the implemented tangential stiffness matrix for shells, or insufficient accuracy in the through-thickness strain calculation. These analyses were performed with the update of shell thickness active (by ISTUPD = 4 on \*CONTROL SHELL).

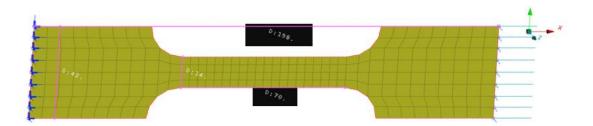


Figure 10. A tensile test specimen with gauge length ~ 70 mm, and width 14 mm. Thickness is 1.5 mm.

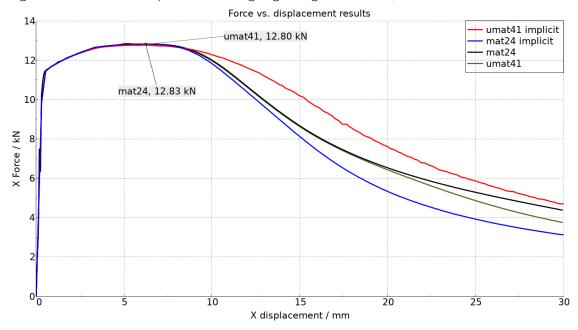


Figure 11. Force vs. displacement results for the tensile test specimen using different material models and the implicit (blue and red curves) or explicit (green and black curves) solver of LS-DYNA.

The second example is axial crushing of a crash box, see Figure 12. The thickness of the profiles is 1.5 mm, and a hardening curve corresponding to steel HX420LAD is applied. For umat 41, it was necessary to disable the shell thickness update (by ISTUPD = 0 on \*CONTROL SHELL) in order to obtain useful results. Instabilities caused premature termination when the shell thickness update flag was active, most likely due to too low accuracy when solving the through-thickness strain. The crushing force

using MAT\_24 and umat41 are compared in Figure 13. Due to the indeterminate nature of this load case, it is far from ideal for a benchmark, and exact agreement is hardly to be expected. However, the large deformations and high plastic strain values pose a great challenge to the material model, and with some modification of the control card settings, also the umat41 manages to handle this load case. With respect to the crushing force, reasonable agreement is obtained, while the final deformed configuration differs quite substantially, see Figure 14. The solution time using mpp/LS-DYNA with 8 processes<sup>7</sup> is 1 h 42 min using MAT\_24 and 2 h 25 min using the umat41, corresponding to an increase of about 42 % in this example. This should be seen as a rough indication of the effect of using a user defined material model on the solution time for an explicit analysis; probably the implementation of J2plasticity presented here as umat41 is not optimal, while MAT\_24 is one of the most efficient material models in LS-DYNA.

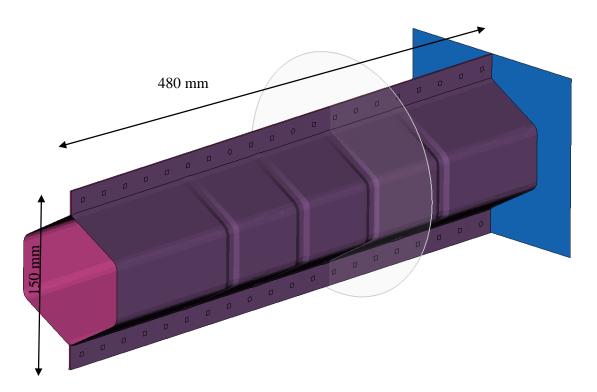


Figure 12. The crash box model. The profiles are made out of 1.5 mm thick HX420LAD steel, and the end plate (blue in the image) is made out of 2.7 mm HX340LAD. The open end of the profile is fully constrained.

<sup>&</sup>lt;sup>7</sup> Intel Xeon E5-2687W v4 CPU (from 2017).

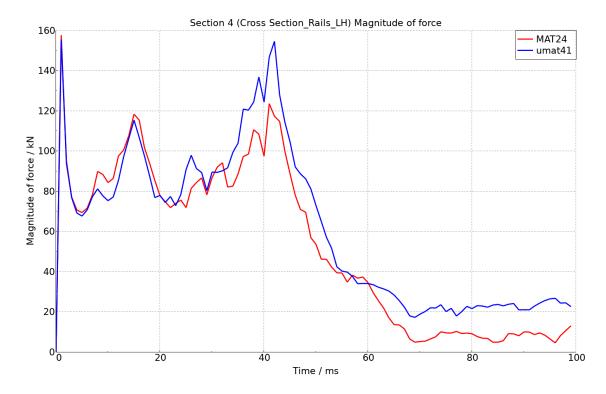


Figure 13. Comparison of the crushing force of the crash box using MAT\_24 and umat 41.

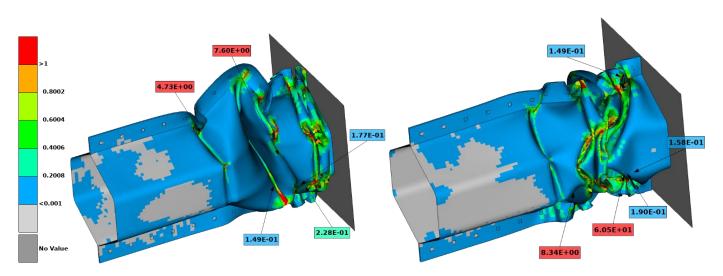


Figure 14. The final deformation of the crash box. The fringe colors show the accumulated effective plastic strain using umat 41 (left image) and MAT\_24 (right image).

The final example of  $J_2$  – plasticity consists of two pipes ( $\varnothing$  90 mm, t = 10 mm) connected by a flange joint with five bolts (M10, strength class10.9), see Figure 15. The geometry is meshed using solid elements, and the example involves bolt pre-tensioning and contacts. Pipe 2 is fully constrained at the free end (black dots to the left in Figure 15). After the bolt pre-tensioning is completed, a prescribed

displacement is applied to the CNRB (blue, to the right Figure 15) of Pipe 1. This load case is solved using the implicit solver in LS-DYNA.

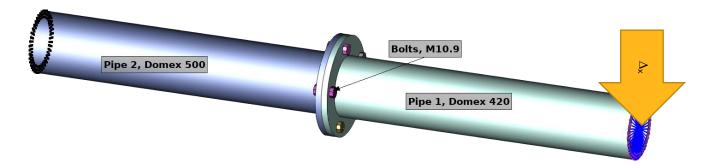
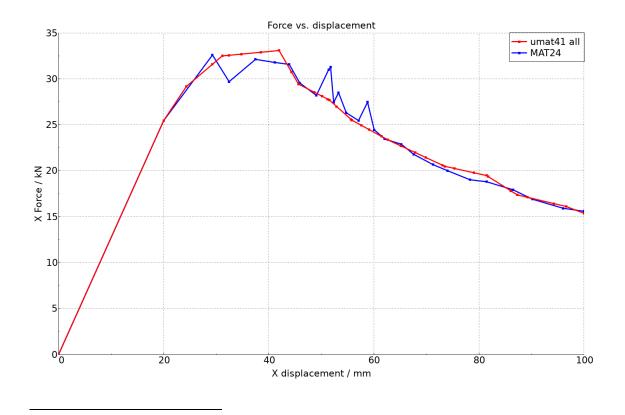


Figure 15. Two pipes are connected by five bolts in a flange joint.

Analyses were performed using umat 41 and MAT\_24 for reference. The global force vs. displacement results are compared in Figure 16. The results are in general agreement, with a slightly smoother response for the umat41. This is probably because more time steps are taken at critical stages (which is most likely triggered by the choice of strainlim = 5.E-2 in the user subroutine, see Section 4.5.2). The final configurations are compared in Figure 17. The peak accumulated effective plastic strain occurs in one of the bolts, and it is 1.73 using MAT\_24 and 1.71 using umat 41. The solution time using mpp/LS-DYNA with 4 processes<sup>8</sup> is approximately the same, 10 minutes, for both MAT\_24 and umat41.



<sup>8</sup> Intel Xeon SP 6148 CPU (from 2018).



Figure 16. Force vs. displacement for the pipe joint model.

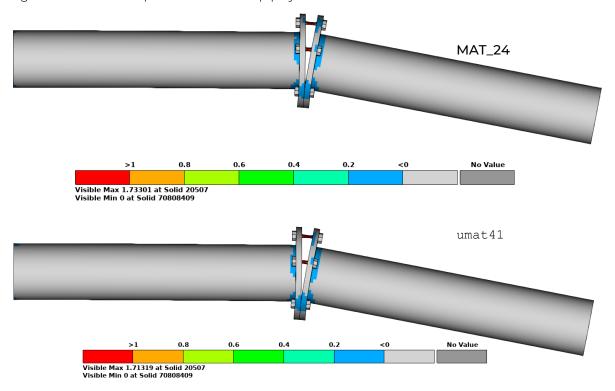


Figure 17. Final configuration of the pipe joint model. The fringe colors show accumulated effective plastic strain. The top image shows results using MAT\_24, and the bottom image shows results using umat 41.

### 4.6.3 Example of the non-linear spring material model

This basic example applies the non-linear spring material model of Section n4.5.3 to a seesaw-like model, see Figure 18. Five non-linear springs (discrete beams) are attached to one end of the solid beam (brown in Figure 18, 1350 hexas) via constrained nodal rigid bodies. The input force-displacement curve for the non-linear springs is shown in Figure 19. A linearly increasing loading is applied to a constrained nodal rigid body at the other end of the beam during 100 ms. The peak applied loading is 1 kN. The solid beam is constrained at 2/3 between the ends.

The force vs. displacement response in one of the springs is shown in Figure 20, compared to the response from LS-DYNA's built-in non-linear spring material model MAT\_67. The two models show good agreement.

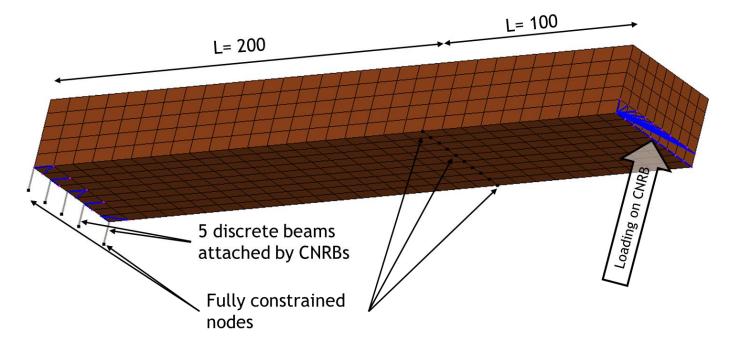


Figure 18. A simple FE-model for demonstrating the user-defined material model for discrete beams.

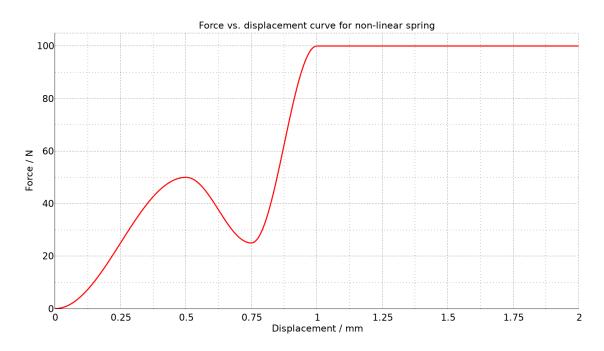


Figure 19. Non-linear force-displacement curve, input to the discrete beam material model.

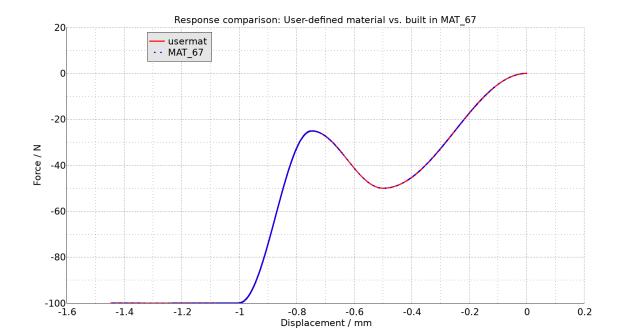


Figure 20. Comparison of the force-displacement response from the user-defined material model and the built-in material model MAT\_67.

## 5 Friction models interface

Ansys LS-DYNA offers a variety of different ways to handle contact between model entities, see for example Ref. [8] [9]. The related keywords start with \*CONTACT\_, see Ref. [1]. A crucial component in any sliding contact is the definition of friction. The standard friction models in LS-DYNA for 3D contacts include:

- Static and dynamic friction coefficients
- Viscous damping
- A cap shear stress, typically related to the yield stress of the materials involved, limiting the peak tangential force in a contact
- Friction coefficients depending on contact pressure and/or temperature
- Orthotropic friction (for some contacts only)

The offering for 2D contacts is limited to static and dynamic friction coefficients with a cap shear stress.

Friction coefficients can be defined per contact interface, per part using \*PART\_CONTACT, or via interaction tables (\*DEFINE FRICTION).

The user defined friction interface makes it possible to develop customized and general friction models for some of the different 3D contacts in LS-DYNA. Due to the internal architecture of the LS-DYNA code, different subroutines are required for the smp $^9$ , mpp and Mortar contact formulations. User defined friction does not apply to the non-Mortar segment-based contacts (SOFT = 2) and also not for the non-

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<sup>&</sup>lt;sup>9</sup> Since the subroutine for smp/LS-DYNA is more complex than the other options, it is left out of the current presentation. It will be included in coming revisions of this document.

Mortar single surface contacts (but \*CONTACT\_AUTOMATIC\_SINGLE\_SURFACE\_MORTAR is supported). Customized friction models are also provided by some third-party companies, for example Triboform [28].

It is currently not possible to define a user friction law for 2D contacts (\*CONTACT 2D).

Note that the user subroutines for user defined friction are called in each explicit time step or implicit iteration, but only for the segments in contact at that particular time.

See also Appendix G of the LS-DYNA Keyword manual [1].

# 5.1 Keyword interface to the user defined friction models

To activate a user defined friction model from a keyword file, three steps are required:

- 1. On Card 2 of \*CONTROL\_CONTACT, set *USRFRC* to the number of parameters passed to the user defined friction model, plus the number of history variables that are stored.
- 2. To make the user defined friction model active for a specific contact ID, the keyword \*USER INTERFACE FRICTION is used.
  - a. This also implies that the option \*CONTACT\_...\_ID must be used, in order to assign an ID to a specific contact definition.
  - b. Set the variable IFID (interface number) to the Contact ID of the \*CONTACT definition
  - c. The extent of material history variables passed to the user friction routines is determined by the NEHIS variable. By setting NEHIS = 0 (which is the default), the plastic strain, yield stress and material directions will be passed. By setting NEHIS > 0, the plastic strain and the element history variables up to NEHIS -1 (in original order) will be passed.
- 3. For the user defined friction to have effect on a specific contact interface, a non-zero static friction (FS) must be defined for that contact interface. Also, for the non-automatic contacts, the shell thickness offset must be activated. This is done by setting SHLTHK = 1 or 2 on \*CONTROL CONTACT, or on Optional Card B of the \*CONTACT ... keyword.

To specify different friction models for different contact interfaces in a subroutine (usrfrc or mortar\_usrfrc), it might be convenient to let the first user defined input parameter denote a reference number to a specific friction model.

#### A keyword example follows:

*USE	R_INTERF	ACE_FRICT	ION						
\$#	IFID	NOC	NOCI	NHSV	NEHIS	MHSV			
	35	2	2	1	0	0			
\$#	UC1	UC2	UC3	UC4	UC5	UC6	UC7	UC8	
	1.	3501.0							
*CONTACT AUTOMATIC SURFACE TO SURFACE MORTAR ID									
\$#	c <u>i</u> d	_		_	_			title	
	35Bl	ock to bas	se						
\$#1	SURFA	SURFB	SURFATYP	SURFBSTYP			SAPR	SBPR	
	2	5	3	3			1	1	
\$#	fs	fd	dc	VC	vdc	penchk	bt	dt	
	0.15								
\$#	sfsa	sfsb							

The variables of the \*USER INTERFACE FRICTION keyword are:

- IFID: The ID of the \*CONTACT to be affected. In this case, the user defined friction shall apply to Contact ID 35 (Block to base).
  - o If a Mortar contact is referred by the ID, the subroutine mortar\_usrfrc will be called, otherwise the subroutine usrfc will be called.
- NOC: The number of variables to be stored for the interface.
- NOCI: The number of variables to be initialized by the user (variables UC1, UC2 ... etc.). NOCI must be smaller or equal to NOC.
- NHSV: The number of history variables per interface node. For Mortar contact it is the number of history variables per tracked<sup>10</sup> segment.
- NEHIS: The number of material history variables to be passed to the subroutine usrfrc.
- MHSV: The number of history variables per reference<sup>11</sup> segment for Mortar contact (ignored by \*CONTACT AUTOMATIC SINGLE SURFACE MORTAR ID).
- UC1, UC2 etc.: Parameters to be passed to the user subroutine.

# 5.2 Post processing user defined friction models

The history variables of the user defined friction models can be post-processed from the intfor<sup>12</sup> – file using LS-PrePost 4.8 (or later). To be able to fringe plot the history variables, it is required to set SPR = 1 on the \*CONTACT\_ - card in question (setting also MPR = 1 is recommended). On the keyword \*DATABASE\_EXTENT\_INTFOR, specify the number of friction history variables to be written to the intfor file using the NHUF parameter. Finally, the keyword \*DATABASE\_BINARY\_INTFOR\_FILE is required to specify the filename (intfor is recommended) and output frequency of the contact data. See Figure 25 for an example of a fringe plot of a user defined friction history variable.

## 5.3 Interfaces to the user defined friction subroutines

Depending on the type of \*CONTACT\_ (non-Mortar or Mortar) that gets a user defined friction model associated with it, either the subroutine usrfrc or mortar\_usrfrc is called. They are both found in the Fortran file dyn21cnt.f. In both cases, curve data as defined by the keyword input is passed via the parameters crv and nnpcrv, in a similar way as for the user defined material routines. In order to evaluate curves, the subroutine crvval as described in Section 4.4 may be used. The smp and mpp versions of the subroutine usrfric differ quite substantially. Currently, only the mpp version will be discussed in this presentation.

The subroutine usrfrc is called for defining the friction coefficients in non-Mortar contacts (SOFT = 0 or 1). The subroutine definition was extended with additional arguments, adding input of more detailed temperature information, starting with R14 of LS-DYNA. For versions R11 to R13, the subroutine definition for implementing a user defined friction model for non-Mortar contacts in mpp/LS-DYNA is

```
subroutine usrfrc(fstt,fdyn,uc,nc,prs,temp,v,vx,vy,vz,uh,nh,
```

- . crv, nnpcrv, nosl,
- . ictype, side, time, ncycle, dt2, fric1, fric2, fric3, fric4, lsv, idele8,

<sup>&</sup>lt;sup>12</sup> This is a file for 3D visualization of contact results, for example contact pressure.



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<sup>&</sup>lt;sup>10</sup> Previously denoted as "slave".

<sup>11</sup> Previously denoted as "master".

- . sfac1, sfac2, insv, fni, areas, stfk, ix1, ix2, ix3, ix4, aream,
- . rn1, rn2, rn3, ue, ne, uhnew)

### For R14 and R15, the subroutine definition instead is

subroutine usrfrc(fstt,fdyn,uc,nc,prs,temp,v,vx,vy,vz,uh,nh,

- . crv, nnpcrv, nosl,
- . ictype, side, time, ncycle, dt2, fric1, fric2, fric3, fric4, lsv, idele8,
- . sfac1, sfac2, intr, fni, areat, stfk, ix1, ix2, ix3, ix4, arear,
- . rn1, rn2, rn3, ue, ne, uhnew, ttrs, trfs, flxrfs)

and an overview of the parameters to the subroutine is shown in Table 8.

The main objective of the subroutine is to compute user defined frictional coefficients, that shall be output in the variables fstt (static) and fdyn (dynamic).



Table 8. Overview of the arguments for the usrfrc subroutine

Argument	Description	Input / Output	
fstt	static frictional coefficient	Output	
fdyn	dynamic frictional coefficient	Output	
uc(nc)	user defined friction parameters	Input	
nc	number of user defined friction parameters	Input	
prs	interface pressure on reference side	Input	
temp	temperature <sup>(1)</sup>	Input	
V	magnitude of relative tangential velocity	Input	
VX,VY,VZ	components of relative tangential velocity	Input	
uh (nh)	user defined friction history variables	Input/Output	
uhnew(nh)	user defined friction history variables	Input/Output	
nh	number of user defined friction history variables	Input	
crv	curve array	Input	
nnpcrv	# of discretization points per crv	Input	
nosl	number of the sliding interface	Input	
ictype	contact type	Input	
side	Info on which side of the contact is being processed	Input	
time	current solution time	Input	
ncycle	number of current cycle	Input	
dt2	time step size at n+1/2	Input	
fric1	static friction coefficient FS from keyword	Input	
fric2	dynamic friction coefficient FD from keyword	Input	
fric3	decay constant DC from keyword	Input	
fric4	viscous friction coefficient VC from keyword	Input	
lsv	reference segment number	Input	
idele8	external user element number of reference segment	Input	
sfac1	Coulomb friction scale factor FSF from keyword	Input	
sfac2	Viscous friction scale factor VSF from keyword	Input	
insv	SURFA <sup>(5)</sup> node user id	Input	
fni	normal force	Input	
areas	SURFA node area <sup>(2)</sup>	Input	
stfk	penalty stiffness	Input	
ix1, ix2,	SURFB <sup>(6)</sup> segment nodes - internal node numbers <sup>(3)</sup>	Input	
ix3, ix4			
aream	SURFB segment area	Input	
rn1,rn2,rn3	737	Input	
ue	Element history data <sup>(4)</sup>	Input	
ne	number of element history variables	Input	
ttrs <sup>(7)</sup>	temperature of the tracked node	Input, from R14	
trfs <sup>(7)</sup>	averaged temperature on the reference segment	Input, from R14	
flxrfs <sup>(7)</sup>	averaged nodal flux vector on reference segment	Input, from R14	

Notes: (1) Available for coupled analysis, as the average value between SURFA and SURFB. Not yet supported for SOFT = 4 contact. (2) The SURFB node pressure is obtained from fni / areas. (3) To go from internal to external (user) node numbers, use ixlext=lqfinv8(ix1,1). (4) The extent of the element history data, ue, is determined by the parameter NEHIS on the keyword \*USER INTERFACE FRICTION, see Section 5.1. (5) Previously denoted "slave". (6) Previously denoted "master". (7) Additional thermal input, available from R14 of LS-DYNA.

From these arguments, it is possible to define a friction coefficient for non-Mortar contacts dependent on, for example,

- the effective plastic strain, or other history variables, of the involved materials (Note that this functionality is available from rev. 144575 of LS-DYNA),
- also, history variables associated with the contact segments,
- time, temperature, contact pressure and sliding velocity.

Note that in order to include the segment history variables in the infor file, the parameters uhnew should be used. The parameters uh will not be output in the intfor file.

Since only one subroutine usrfric is defined, accommodating for several friction models can be achieved by letting one of the user-defined parameters, for example uc(1), denote the ID of a friction model.

The subroutine mortar\_userfrc is called for defining the friction coefficient in Mortar contacts. The subroutine definition for implementing a user defined friction model for Mortar contacts, up until R13 of LS-DYNA is:

```
subroutine mortar_usrfrc(init,mfrc,nprm,cprm,shst,mhst,icnt,
1    selm,sprt,styp,stmp,seps,shis,
2    melm,mprt,mtyp,mtmp,meps,mhis,
3    cprs,vtan,crv,nnpcrv,dt)
```

For R14 and later versions of LS-DYNA, names of some of the parameters have changed:

```
subroutine mortar_usrfrc(init,mfrc,nprm,cprm,thst,rhst,icnt,

telm,tprt,ttyp,ttmp,teps,this,

relm,rprt,rtyp,rtmp,reps,rhis,

cprs,vtan,crv,nnpcrv,dt)
```

But the number of parameters is the same.

An overview of the parameters to the subroutine is shown in Table 9. The main objective of the subroutine is to compute a user defined frictional coefficient, that shall be output in the variables mfrc.

Table 9. The arguments to the subroutine mortar usrfric

Argument, R13	R14 →	Description	Input/Output
init		Initialization phase (.true. or .false.) (currently not active)	Input
mfrc		User defined friction coefficient	Output
nprm		number of user friction parameters	Input
cprm		list of user friction parameters	Input
shst	thst	SURFA friction history variables	input/output
mhst	rhst	rference friction history variables	input/output
icnt		contact interface id	Input
selm	telm	telm element id for SURFA segment	
sprt	tprt	tprt part id for SURFA segment	
styp	ttyp element type for SURFA segment ('beam ','solid','shell' or 'tshel')		Input
stmp	temperature of SURFA segment (n/a during initialization)		Input
seps teps		effective plastic strain on SURFA side	Input
shis this		material history variables for the SURFA segment	Input
melm relm		element id for SURFB segment (n/a during initialization)	Input
mprt	rprt	part id for SURFB segment (n/a during initialization)	Input
mtyp rtyp		element type for SURFB segment ('beam ','solid','shell' or 'tshel',	Input
		n/a during initialization)	
mtmp	rtmp	temperature of SURFB segment (n/a during initialization)	Input
meps	neps reps effective plastic strain on SURFB side		Input
mhis	rhis	material history variables for the SURFB segment	Input
cprs		contact interface pressure (n/a during initialization)	Input
vtan		tangential relative sliding velocity (n/a during initialization)	Input
crv		curve object (to be used in evaluating curve/table, n/a during	Input
		initialization)	
nnpcrv		curve parameters (to be used in evaluating curve/table, n/a during	Input
		initialization)	
dt		time step	Input

From these arguments, it is possible to define a friction coefficient for Mortar contacts dependent on, for example,

- the effective plastic strain, or other history variables, of the involved materials,
- history variables associated to the contact segments,
- temperature, contact pressure and sliding velocity.

The current solution time is not passed to the mortar usrfric subroutine. The time in contact can be obtained by using a history variable for summing up the time steps dt.

## 5.4 Subroutine examples

In this Section, examples of user defined friction models for mpp<sup>13</sup>/LS-DYNA are presented, for both Mortar and non-Mortar contacts. The related subroutines are found in the Fortran file dyn21cont.f. Both Fortran code and keyword input are presented, while the complete examples, see also Section 5.5, can be found as attachments to this document. It shall be stressed that these examples are not intended for use in any kind of production analysis, and there may very well contain errors or flaws.

### 5.4.1 Time dependent friction coefficient for Mortar contact

For Mortar contact, the implementation of a time dependent friction coefficient is described. Two methods for computing the time in contact will be implemented, either by

- friction model 1: dt is summed up and stored in a history variable for the tracked segment, or
- friction model 2: dt is summed up and stored in history variables for both tracked and reference segment, and the time is taken as the maximum value of these.

The user will have to select a friction model and give a curve ID specifying the coefficient of friction as a function of time. Also, a default friction coefficient can be given, to be use in case the curve evaluation should result in a negative value. The coefficient of friction will be stored as the second tracked side history variable. The keyword interface will be

```
*USER INTERFACE FRICTION
                                            MHSV
   IFID NOC
                    NOCI
                            NHSV
                                   NEHIS
contact ID
             3
                    3
                            2
                                             1
            UC2
                    UC3
                             UC4
                                     UC5
                                             UC6
                                                     UC7
                                                             UC8
    UC1
            LCID default µ
   1 or 2
```

where blue text indicates that the user should input sensical data, and red text indicates text that should not be changed.

The fist part of the subroutine mortar usrfrc, involving subroutine and variable declarations, follows:

```
subroutine mortar usrfrc(init, mfrc, nprm, cprm, shst, mhst, icnt,
     1
            selm, sprt, styp, stmp, seps, shis,
     2
           melm, mprt, mtyp, mtmp, meps, mhis,
           cprs, vtan, crv, nnpcrv, dt)
      implicit none
      include 'nlqparm'
      include 'iounits.inc'
      logical init
      real mfrc,cprm(*),shst(*),mhst(*),cprs,vtan,stmp,mtmp,crv(lq1,2,*)
      real shis(*), mhis(*), seps, meps, dt
      integer selm, sprt, melm, mprt, nprm, icnt, nnpcrv(*)
      character*5 styp, mtyp
C
      real dmdp, dmdv
```

<sup>&</sup>lt;sup>13</sup> An example for user defined friction in smp/LS-DYNA will be provided in coming versions of this Guideline.



Here, the implicit none statement will require all variables to be declared explicitly, reducing the risk for programming errors.

In the original dyn21cnt.f Fortran file of the usermat package, the comments after the subroutine declaration provide some documentation regarding the parameters of the user friction subroutine. These comments are omitted here.

The coding for the friction model 1 follows:

```
C --- Law 1: use curve to define friction coefficient, time is time in
С
             contact for the tracked segment
C ----
             cprm(1) = Law id, cprm(2)=lcid, cprm(3)=default friction
      if (cprm(1).eq.1.) then
         if (shst(1).eq.0) then
           write(iomsg,*) ' --- mortar usrfric law ',cprm(1),' using ',
     1
            'lcid',cprm(2),' default friction is ',cprm(3)
         endif
         shst(1) = shst(1) + dt
         call crvval(crv,nnpcrv,cprm(2),shst(1),mfrc,dmdp)
         if (mfrc.ge.0) then
            shst(2)=mfrc
         else
            mfrc=cprm(3)
            shst(2)=mfrc
         endif
      endif
```

It starts by writing a debug output message, in case the tracked side history variable is zero. Then the time in contact is updated and stored in the tracked side history variable shst(1). The subroutine crvval is then called to evaluate the curve and obtain the coefficient of friction in the mfrc variable. Finally, checking is done and in case a negative value was returned, it is replaced by the default coefficient of friction given as UC3 from \*USER INTERFACE FRICTION in the variable cprm(3). The applied coefficient of friction is stored in the 2<sup>nd</sup> history variable for visualization purposes.

The second friction model is similar:

```
C --- Law 2: use curve to define friction coefficient, time is max
             time in contact for the tracked or reference segment
С
c ----
             cprm(1) = Law id, cprm(2)=lcid, cprm(3)=default friction
      if (cprm(1).eq.2.) then
         if (shst(1).eq.0) then
           write(iomsg,*) ' --- mortar usrfric law ',cprm(1),' using ',
            'lcid',cprm(2),' default friction is ',cprm(3)
     1
         endif
         shst(1) = shst(1) + dt
         mhst(1) = mhst(1) + dt
         dmdv=max(shst(1), mhst(1))
         shst(1) = dmdv
         call crvval(crv,nnpcrv,cprm(2),dmdv,mfrc,dmdp)
         if (mfrc.ge.0) then
            shst(2) = mfrc
```

```
else
      mfrc=cprm(3)
      shst(2) = mfrc
   endif
endif
```

The difference to the friction model 1 is that also the reference side history variable mhst (1) is used for storing the time in contact. Then the maximum value of the time of the tracked and reference side is taken as the time in contact, and this value is then also stored in the 1st reference side history variable mhst (1). Again, in case a negative coefficient of friction should be obtained from the curve, it is replaced by the default value input as UC3 from \*USER INTERFACE FRICTION.

An example of a simulation using this mortar usrfrc subroutine is provided in Section 5.5.1.

### 5.4.2 Friction depending on contact pressure and plastic strain

To illustrate the implementation of a user defined friction model for non-Mortar contacts, a model using curves to scale the friction as a function of contact pressure and accumulated effective plastic strain is described in this section. The user interface will involve two curve ID:s, and a max and min value to limit the coefficient of friction. The coefficient of friction will be stored as the 1st history variable, the scale factor related to plastic strain as the 2<sup>nd</sup>, and the scale factor related to contact pressure as the 3<sup>rd</sup>. The keyword interface will be

```
*USER INTERFACE FRICTION
   IFID NOC
                   NOCI
                         NHSV
$#
                                 NEHIS
                                          MHSV
            11
contact ID
                   5
                           3
                                            3
  UC1
            UC2
                   UC3
                           UC4
                                  UC5
                                           UC6
                                                  UC7
                                                          UC8
           LCID1
                  LCID2 min.frict max.frict
```

where blue text indicates that the user should input sensical data, and red text indicates text that should not be changed. In this case, UC1 is reserved for (future) use as a friction model ID, but since only one model will be implemented, the only sensical input is 1. The LCID1 should correspond to a curve ID scaling the coefficient of friction by a factor depending on the accumulated effective plastic strain, and LCID2 should correspond to a curve ID scaling the coefficient of friction as a function of the contact pressure. The lower and upper bounds should be input as UC4 and UC5 respectively, in order to keep the applied coefficient within reasonable limits (making it possible to avoid unrealistic extrapolations in curves, which may be caused by for example very high values of plastic strain locally).

The fist part of the subroutine usrfrc, involving subroutine and variable declarations, follows:

```
subroutine usrfrc(fstt,fdyn,uc,nc,prs,temp,v,vx,vy,vz,uh,nh,
     . crv, nnpcrv, nosl,
     . ictype, side, time, ncycle, dt2, fric1, fric2, fric3, fric4, lsv, idele8,
     . sfac1, sfac2, insv, fni, areas, stfk, ix1, ix2, ix3, ix4, aream,
     . rn1,rn2,rn3,ue,ne,uhnew)
С
      implicit none
      include 'nlqparm'
      include 'iounits.inc'
      real fstt, fdyn
      integer nc,nh,nosl,ictype,ncycle,lsv,insv,ix1,ix2,ix3,ix4
      integer ne
```

Here, the implicit none statement will require all variables to be declared explicitly, reducing the risk for programming errors.

In the original dyn21cnt.f Fortran file of the usermat package, the comments after the subroutine declaration provide some documentation regarding the parameters of the user friction subroutine. These comments are omitted here.

In the next section of the subroutine, the output variables fstt and fdyn are initialized to the values input on the \*CONTACT ... keyword card, and a debug message is written in the mes0\* - files.

```
С
      set coefficients to keyword values
С
С
      fstt=fric1
      fdyn=fric2
C --- Law 1: use curves to scale fric1 as a function of plastic strain
С
    and contact pressure
C ----
             uc(1) = Law id, uc(2) = lcid for plascit strain, uc(3) = lcid
С
             for pressure, uc(4) = min friction, uc(5) = max friction
      if (uc(1).eq.1.) then
         if (time.le.1E-3) then
           write(iomsg,*) ' --- forming usrfrc law ',uc(1),' using ',
     1
            'lcid for plastic strain:',uc(2),
     2
            'lcid for contact pressure:',uc(3),
     3
            'min friction is ',uc(4),'max friction is ',uc(5)
         endif
```

In the final part of the subroutine, the curves are evaluated to extract the scale factors for plastic strain and contact pressure respectively, and a candidate friction coefficient crfc is computed by scaling the input static friction coefficient FS of the \*CONTACT\_ ... card, which is passed in the variable fric1. Finally, this candidate coefficient is checked against the min and max allowed values. The history variables uhnew are updated, for post-processing purposes, with the coefficient of friction as the lst history variable, the scale factor for plastic strain as the lst history variable and the scale factor for contact pressure as the lst history variable as the lst history variable and the scale factor for contact pressure as the lst history variable and the scale factor for contact pressure as the lst history variable and the scale factor for contact pressure as the lst history variable and the scale factor for contact pressure as the lst history variable and the scale factor for contact pressure as the lst history variable and lst history variable and the scale factor for contact pressure as the lst history variable and lst history variable as lst history variable and lst

```
call crvval(crv,nnpcrv,uc(2),ue(1),epsfac,tdum)
call crvval(crv,nnpcrv,uc(3),prs,prsfac,tdum)
cfrc = fric1*epsfac*prsfac
fstt = cfrc
if(cfrc.lt.uc(4))then
    fstt = uc(4)
else if(cfrc.gt.uc(5))then
```



```
fstt = uc(5)
endif
uhnew(1) = fstt
uhnew(2) = epsfac
uhnew(3) = prsfac
endif
```

An example of a simulation using this usrfrc subroutine is provided in Section 5.5.2. This friction model is also implemented for Mortar contact, as friction model 3 (UC1 = 3) in the  $mortar_usrfc$  subroutine provided as an attachment to this Guideline.

# 5.5 LS-DYNA simulation examples

In this section, two LS-DYNA simulation examples of user defined friction are presented.

## 5.5.1 Mortar contact: a cube on a tilting plane

This is an implicit simulation of a classical set-up for determining the coefficient of friction between two bodies, see Figure 21. A cube (50×50×50 mm) is placed on a plane, and during the first second, gravity is ramped up. Then, the plane is tilted to an angle of 8.3° from t = 2 to t = 7 seconds, see Figure 23. The coefficient of friction is ramped down according to Figure 22, causing the cube to start sliding. The final configurations, using friction model 1 and 2 respectively, are compared in Figure 24. Using friction model 1, the time in contact is defined only on the tracked segment side. This means that once the cube starts sliding as the coefficient of friction is reduced by the time in contact (following the curve of Figure 22), contact will be made with "new" segments with (initially) zero time in contact, causing a high coefficient of friction again, which in turn reduces the sliding velocity and, as the cube falls of the plane, causing it to rotate. Using the friction model 2, this phenomenon is reduced. This is also illustrated by the comparison of the final coefficient of friction (output as contact history variable #2) in Figure 25. Note that the contact history variables are found in the intfor file for fringe plotting.



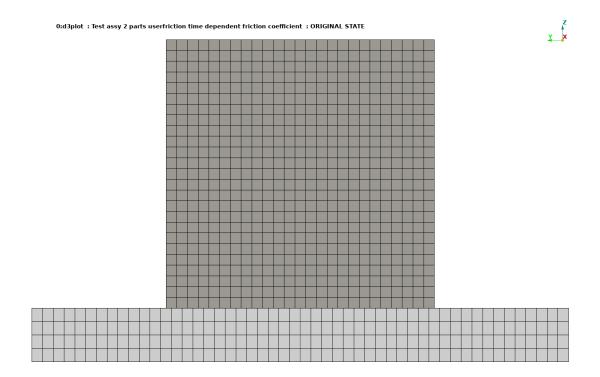


Figure 21. A cube (gray) is resting on a plane (light gray).

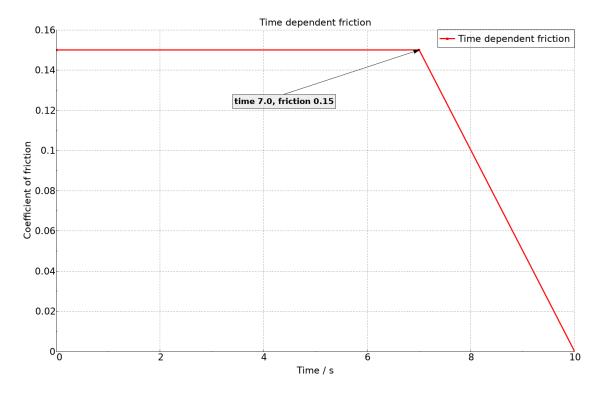


Figure 22. Time-dependent coefficient of friction used in the Mortar contact example.

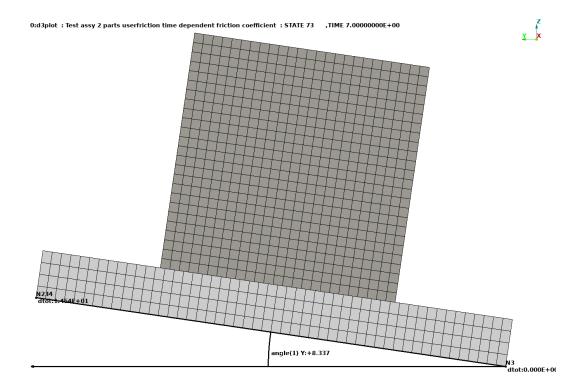


Figure 23. The plane is tilted to an angle of 8.3°, and then the coefficient of friction is decreased.

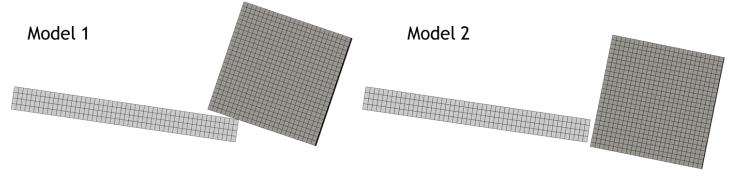


Figure 24. The left image shows the configuration at t = 10 s using friction model 1. The right image shows the configuration at t = 10 s using friction model 2.

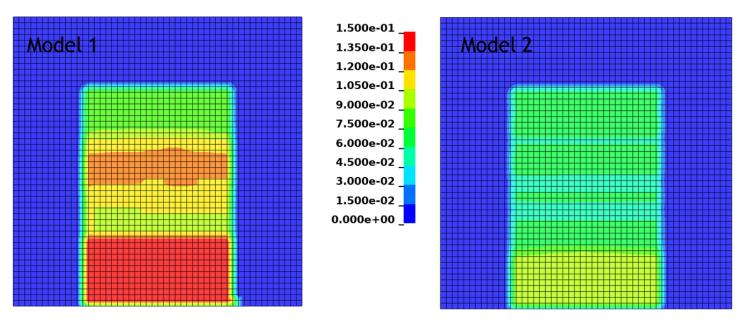


Figure 25. The fringe colors show the final coefficient of friction from 0 (blue) to 0.15 (red) using friction model 1 (left image) compared to the friction model 2 (right image).

## 5.5.2 Forming analysis using pressure and plastic strain dependent friction

In this example, the forming of a so-called S-rail [25] is studied, see

Figure 26. The S-rail forming simulation case.

. The analysis is performed using the explicit solver in LS-DYNA, with adaptive mesh refinement and three forming contacts (\*CONTACT FORMING ONE WAY SURFACE TO SURFACE ID). The assumed curves used for scaling the friction coefficient as a function of plastic strain and contact pressure are shown in Figure 27. In the reference simulation, a constant coefficient of friction ( $\mu$  = 0.125) was used. The shell thickness results are compared in Figure 28, where some minor differences in shell thickness due to the different friction models can be found. The distribution of the coefficient of friction at an intermediate state is shown in Figure 29.

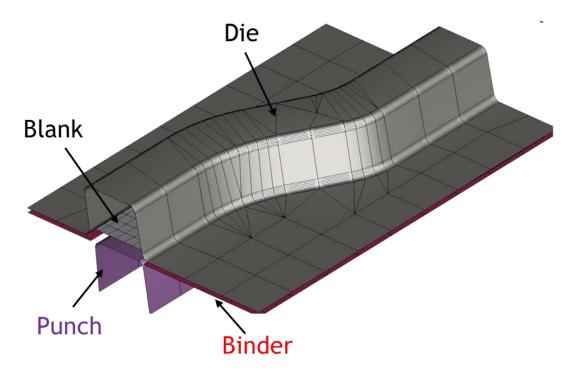


Figure 26. The S-rail forming simulation case.

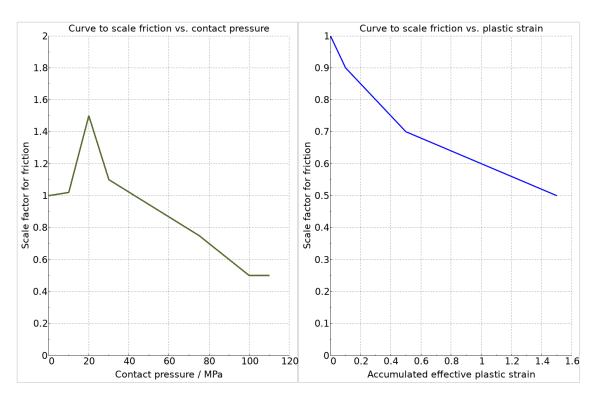


Figure 27. The left image shows the curve for scaling the friction coefficient as a function of contact pressure. The right image shows the curve for scaling the friction as a function of plastic strain.

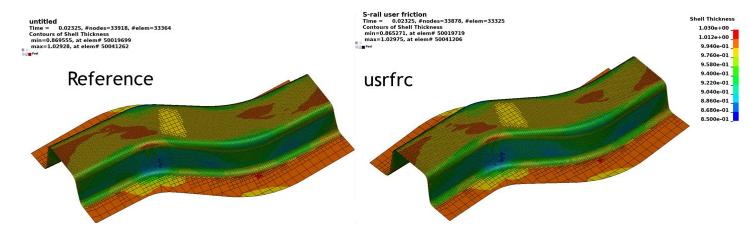


Figure 28. The resulting shell thickness distribution using constant friction (left image) and the user defined friction model.

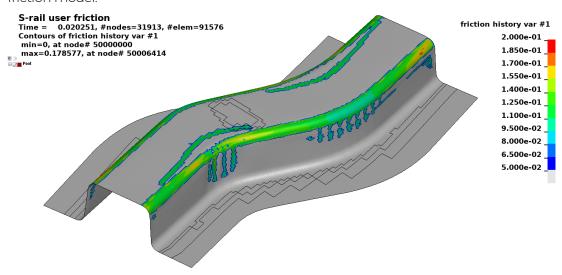


Figure 29. The friction history variable #1 for the contact between the bank and the punch, which is the current static friction coefficient, for the S-rail example.

## 6 Tied contact using Mortar weld tie

The original purpose of the tied weld options (\*CONTACT\_AUTOMATIC\_SURFACE\_TO\_SURFACE\_...
\_TIED\_WELD) is for simulating a welding process: a sliding contact is irreversibly transformed to a (penalty based) tied contact when the temperature exceeds a user-specified value. This behavior is useful for simulating for example two materials that are joined by partial melting of the surfaces by an external heat source (weld torch) and when the melted metal cools down, the parts remain joined.

The user defined tie condition is only available for the Mortar formulation of this contact. It also requires that a coupled thermomechanical analysis (SOLN = 2 on \*CONTROL\_SOLUTION) is performed. The user defined tied condition is available from revision 143414 of LS-DYNA.

Another possibility for tying surfaces which are originally separated together is to use the <code>OPTION = 1</code> of <code>\*CONTACT\_AUTOMATIC\_SURFACE\_TO\_SURFACE\_TIEBREAK</code>. By this, the surfaces will be permanently tied together as soon as they make contact, independent of the temperature.

#### 6.1 Keyword interface to the user defined weld tie condition

The keyword interface to the user defined tie weld condition is activated by specifying a negative value of TEMP on the contact defined via \*CONTACT\_AUTOMATIC\_SURFACE\_TO\_SURFACE\_MORTAR\_THERMAL\_TIED\_WELD\_ID. A keyword example follows:

*CO	*CONTACT_AUTOMATIC_SURFACE_TO_SURFACE_MORTAR_TIED_WELD_THERMAL_ID							
	32E	Base to fil	m					
\$#1	SURFA	SURFB	SURFATYP	SURFBSTYP			SAPR	SBPR
	3	2	3	3			1	1
\$#2	FS							
	0.15							
\$#3								
\$#4	TEMP	CLOSE	HCLOSE	NTPRM	NMHIS	NSTWH	NMTWH	
	-1.		1.E+3	3		2		
\$#	TPRM1	TPRM2	TPRM3	TPRM4	TPRM5	TPRM6	TPRM3	TPRM4
	0.5	102.	5.					
\$#	Thermal	properties						
	0.000	0.000	1.E+3	1.0000E-3	5.0000E-3	0.5	1	

The variables related to the user defined tie condition are:

- TEMP: Set -1000 < TEMP < 0 to activate a user defined tie condition. The absolute value of TEMP will be passed to the user subroutine mortar\_usrtie as the tied weld ID, which then can be used for specifying different user defined conditions.
- *CLOSE*: Segments within this distance are considered for tying. The default is 1 % of the characteristic mesh length scale.
- HCLOSE: The thermal contact conductivity when tied.
- NTPRM: Number of user defined weld tie parameters (TPRM1, TPRM2 etc.)
- NMHIS: Number of material history variables (in addition to plastic strain) to be accessible in the subroutine mortar usrtie
- NSTWH: Number of tracked side tied weld history variables.
- NMTWH: Number of reference side tied weld history variables.
- TPRM1, TPRM2, etc: User parameters.

Note that the accumulated effective plastic strain for reference and tracked side is accessible even if NMHTS = 0.

### 6.2 Post processing user weld tie condition

The history variables of the user defined weld tied condition subroutine can be post-processed from the intfor file using LS-PrePost 4.8 (or later). To be able to fringe plot the history variables, it is required to set SPR = 1 on the \*CONTACT\_...TIED\_WELD\_ID - card (setting also MPR = 1 is recommended). On the keyword \*DATABASE\_EXTENT\_INTFOR, specify the number of SURFA and SURFB weld tie history variables to be written to the intfor file using the 7<sup>th</sup> and 8<sup>th</sup> positions on Card 2, respectively. It is also recommended to activate the output of a tie indicator by setting NTIED = 1 on Card 2. The keyword \*DATABASE\_BINARY\_INTFOR\_FILE is required to specify the filename (intfor is recommended) and output frequency of the contact data. See Figure 32 for examples of fringe plots of user defined tied weld history variables.



## 6.3 Interface to the user defined tie condition in the subroutine mortar usrtie

The condition for switching from sliding to tied contact is defined in the subroutine mortar\_usrtie, which is found in the dyn21cnt.f Fortran file. The subroutine definition for implementing a user defined tie condition, up until R13 of LS-DYNA, is:

For R14 and later versions of LS-DYNA, names of some of the parameters have changed:

The subroutine is called for each pair of reference-tracked contact segments. Note that the user subroutine is only called for the segments in consideration for contact, which is also related to the <code>CLOSE</code> parameter value. For those segments further away than the <code>CLOSE</code> value, the contact is not considered as active, and the subroutine is not called. The objective of the subroutine is to indicate if the segments are to be tied together or not, by the logical parameter <code>tie</code>. Note that <code>tie</code> must be initialized to <code>.false</code>. in the subroutine in order to make sure that undesired tying is avoided. An overview of the parameters to the subroutine is shown in Table 10. Curve data as defined by the keyword input is passed via the parameters <code>crv</code> and <code>nnpcrv</code>, in a similar way as for the user defined material routines, and in order to evaluate curves, the subroutine <code>crvval</code> as described in Section 4.4 may be used.



Table 10. Overview of the arguments for the mortar usrtie subroutine

Argumen	nt From R14 →	Description	Input / Output
	tid	tie interface id	Input
tie		set to .true. if tie condition is met, otherwise .false.	Output
	init	initialization phase (.true. or .false.) (currently not active)	(Input)
nprm		number of user tie parameters	Input
	cprm	list of user tie parameters, use only nprm	Input
shst	Thst	tie history variables SURFA side	Input/output
mhst	Rhst	tie history variables SURFB side	Input/output
icnt		contact interface id	Input
selm	selm	element id for SURFA segment	Input
sprt	sprt	part id for SURFA segment	Input
styp	styp	element type for SURFA segment ('beam ','solid','shell' or	Input
		'tshel')	
stmp	stmp	temperature of SURFA segment (n/a during initialization)	Input
seps	seps	effective plastic strain on SURFA side	Input
shis	shis	material history variables for the SURFA segment	Input
melm	melm	element id for SURFB segment (n/a during initialization)	Input
mprt	mprt	part id for SURFB segment (n/a during initialization)	Input
mtyp	mtyp	element type for SURFB segment ('beam ','solid','shell' or	Input
		'tshel')	
mtmp	mtmp	temperature of SURFB segment (n/a during initialization)	Input
meps	meps	effective plastic strain on SURFB side	Input
mhis	mhis	material history variables for the SURFB segment	Input
	cprs	contact interface pressure (n/a during initialization)	Input
cshr		contact interface shear stress (n/a during initialization)	Input
crv		curve object (to be used in evaluating curve/table)	Input
	npcrv	curve parameters (to be used in evaluating curve/table)	Input
	time	simulation time	Input
	dt	time step size	Input

From these arguments, it is possible to define a tie condition depending on, for example,

- the effective plastic strain, or other history variables, of the involved materials,
- also history variables associated with the contact segments on the tracked (SURFA) and reference (SURFB) side,
- time, temperature, and contact pressure.

Once the tied condition is met, and tie is set to .true., the segment pair is removed from the checking loop, and will not be passed to the subroutine again. This means that the segment history variables cannot be updated once the tied condition is met.

### 6.4 Subroutine example

In this section, a basic tie condition depending on the time in contact, contact pressure and temperature will be implemented. It shall be stressed that this example is not intended for use in any kind of production analysis, and it may very well contain errors or flaws.

This may have some similarity with hot glue that cures (or solidifies) below a certain temperature, which in combination with a certain contact pressure being applied for a specified amount of time creates a glued bond. For this, three parameters would be required: a critical minimum contact

pressure  $p\theta$ , a critical transition temperature tthresh, and the time ctime required to create the bond. One history variable for storing the time in contact, during which these requirements are fulfilled, is a minimum, but in addition a second history variable for storing the first time of contact will be created for post-processing and visualization.

#### The keyword interface will be:

```
*CONTACT AUTOMATIC SURFACE TO SURFACE MORTAR TIED WELD THERMAL ID
CID
       Title
$#1 SURFA
          SURFB SURFATYP SURFBSTYP
                                                       SAPR
                                                              SBPR
Specify what should be in contact
     FS
Specify friction
$#3
$#4 TEMP CLOSE HCLOSE NTPRM NMHIS
                                             NSTWH
                                                      NMTWH
           therm.cond
    -1.
  TPRM1
          TPRM2 TPRM3
                                                      TPRM3
                                                              TPRM4
                            TPRM4
                                    TPRM5
                                             TPRM6
  tthresh p0
                   ctime
$# Thermal properties
Specify thermal properties
```

where blue text indicates that the user should input sensical data, and red text indicates values that should not be changed.

The first part of the subroutine mortar\_usrtie involving subroutine and variable declarations, follows:

```
subroutine mortar usrtie(tid,init,tie,nprm,cprm,
      shst, mhst, icnt,
1
      selm, sprt, styp, stmp, seps, shis,
2
      melm, mprt, mtyp, mtmp, meps, mhis,
      cprs,cshr,crv,nnpcrv,time,dt)
 implicit none
 include 'nlqparm'
 include 'iounits.inc'
 real dt
 logical init, tie
 real cprm(*), shst(*), mhst(*), cprs, cshr, stmp, mtmp, crv(lq1,2,*)
 real shis(*), mhis(*), seps, meps, time
 integer tid, selm, sprt, melm, mprt, nprm, icnt, nnpcrv(*)
 character*5 styp,mtyp
```

In the original dyn21cnt.f Fortran file of the usermat package, the comments after the subroutine declaration provide some documentation of the user tie subroutine, regarding for example parameters. These comments are omitted here.

The main part of the subroutine starts by initializing the tie indicator to .false. and writing a message to the mes0\* files, to confirm what subroutine is active.

```
cprm(2), " and time in contact is > ",
cprm(3)
endif
```

Then the history variables are updated. The first time in contact, with a contact pressure above the critical value  $p\theta$ , is stored in the first history variable shst(1). If the contact pressure is above  $p\theta$  and temperature is below tthresh, then the time in contact, stored in the  $2^{nd}$  history variable shst(2), is incremented by the current time step size dt.

```
if((stmp.le.cprm(1)).and.(cprs.ge.cprm(2))) then
    shst(2) = shst(2) + dt
endif
if((shst(1).lt.time).and.(cprs.lt.cprm(2)))then
    shst(1) = 0.
endif
if((shst(1).eq.0.).and.(cprs.ge.cprm(2)))then
    shst(1) = time
endif
```

Finally, the tie condition is checked: if the contact pressure is above the critical value  $p\theta$  and the temperature is below tthresh and the time in contact, stored in the  $2^{nd}$  history variable shst(2), exceeds the required ctime, the tie flag is set to true, and a message is printed to confirm which element that got tied.

```
tie when tracked temperature is below cprm(1) and contact
С
      pressure is above cprm(2) and time in contact is greater than
С
С
      cprm(3)
         if((stmp.le.cprm(1)).and.(cprs.ge.cprm(2))) then
           if(shst(2).gt.cprm(3)) then
             tie=.true.
             write (iomsg,*) '--- Test Mortar usertie 1:elem ',
             selm, 'now tied at temp ', stmp, 'cpress ',
     1
             cprs,' time in contact ',shst(2)
           endif
         endif
      endif
```

This is a rather basic example. Much more involved criteria are possible based on the accessible model quantities, and the history variables of the contact segments can of course have more intricate evolution laws. An example of a simulation using this mortar\_usrtie subroutine is provided in Section 6.5.

### 6.5 LS-DYNA simulation example

In this example, a rubber sheet is pressed against an aluminum plate by a steel cube, see Figure 30 for geometries and initial temperatures. The test case is analyzed using the implicit solver of LS-DYNA, as a coupled thermomechanical simulation.



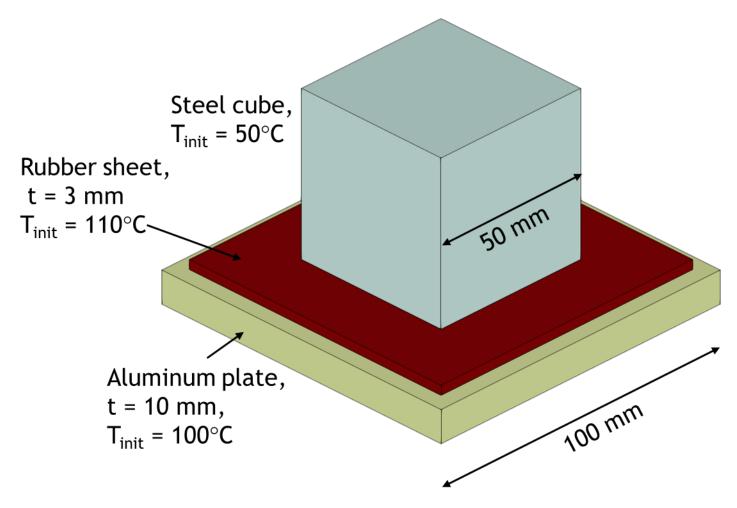


Figure 30. The geometry and initial temperatures for the test case.

A user defined weld tie condition is applied between the rubber sheet and the aluminum plate, using the subroutine as described in Section 6.4, with a critical pressure  $p\theta$  = 0.5 MPa, a critical transition temperature tthresh, = 102°C and a required time ctime = 5 seconds.

Mechanical loading is applied to the set-up in two steps:

- 1. A distributed loading is applied to the topside of the steel cube: ramped up for 1 second, kept constant for 9 seconds, and then ramped down.
- 2. A distributed loading is applied to the topside of the rubber sheet, in order to illustrate which segments that got bonded with the aluminum plate.

The loading history is also illustrated in Figure 31. The tie indicator and contact history variable #2 (time in contact) is shown in Figure 32. The final deformed configuration is shown in Figure 33, from which it is concluded that the tied condition between the rubber sheet and the aluminum plate is enforced correctly.

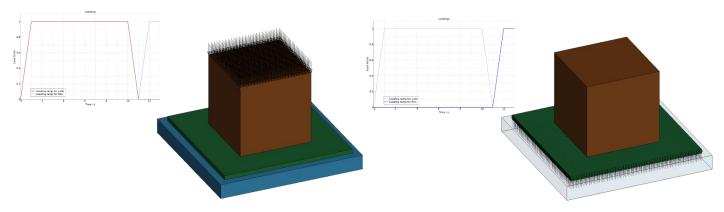


Figure 31. Load case description. First the cube is pushed down (left image) then the sheet is lifted (right image).

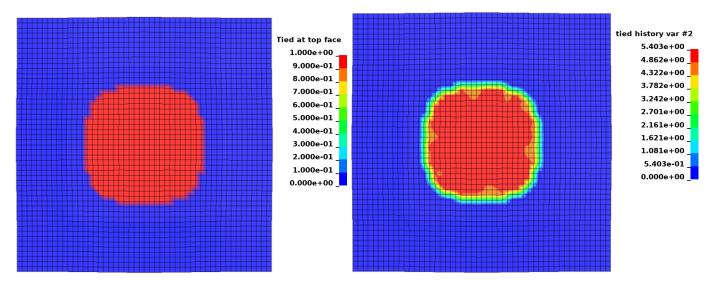


Figure 32. The left image shows a fringe plot of the tied indicator from the intfor file. The right image shows the 2<sup>nd</sup> history variable, time in contact.

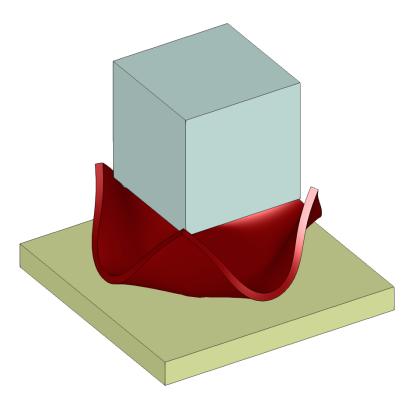


Figure 33. Deformed configuration during load step 2, application of the distributed loading to the rubber sheet.

#### Mortar tiebreak contact

The main purpose of a tiebreak contact is to model surfaces that are initially connected (bonded, glued, welded etc.) but due to high loading or other effects may separate during the analysis. An initially tied contact is irreversibly transformed into a sliding contact. Alternatively, cohesive material models, for example \*MAT COHESIVE MIXED MODE or \*MAT COHESIVE GENERAL may be applied in an interface layer with cohesive elements between parts that may separate.

In this section only the Mortar formulation of the tiebreak contacts is described. The relevant keyword for the tiebreak functionality is \*CONTACT AUTOMATIC SURFACE TO SURFACE TIEBREAK USER MORTAR. The tiebreak contacts have many predefined criteria for damage and failure of the tied interface, see Ref. [1], based on stress, energy release rate or cohesive models, see for example Figure 34.

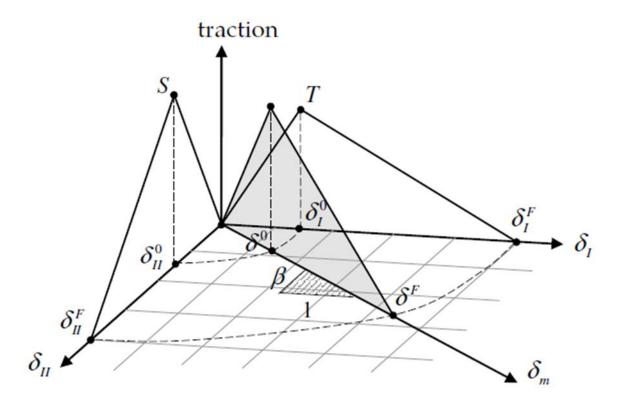


Figure 34. Cohesive mixed-mode law (according to \*MAT COHESIVE MIXED MODE) for traction-separation, invoked by option = 9 in the tiebreak contacts.

In a sense, the tiebreak contact is the inverse of the Mortar weld tie contact (see Section 6). It is possible to inherit history variables from a joining simulation where a \*CONTACT AUTOMATIC SURFACE TO SURFACE\_MORTAR\_THERMAL\_TIED\_WELD\_ID was used and formulate a separation criterion based on the adhesion process results.

The user defined mortar tiebreak condition is fully supported from revision R13-1674-g3b7bda8165 of LS-DYNA, but with exception of the access to material history variables and plastic strain is available already in R13.0.0.

For the contact segments where the tied contact is released, a "normal" Mortar sliding contact remains, keeping the segments from penetrating each other.

#### Keyword interface to the user defined tiebreak condition 7.1

The keyword interface to the user defined tiebreak condition is activated by the keyword \*CONTACT AUTOMATIC SURFACE TO SURFACE TIEBREAK USER MORTAR ID. A keyword example follows:

```
*CONTACT AUTOMATIC SURFACE TO SURFACE TIEBREAK USER MORTAR ID
       32Base to film
           SURFB SURFATYP SURFBSTYP
$#1 SURFA
                                                                SAPR
                                                                          SBPR
```

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	3	2	3	3			1	1
\$#2	FS							
	0.15							
\$#3								
Ċ II A	ODELON	NTTTT 7	QE O QN	CNI		N111N47 (T)	NIIITI D	
\$#4	OPTION	NHV	CT2CN	CN	OFFSET	NHMAT	NHWLD	
	103	3	1.			3	2	
\$#	UP1	UP2	UP3	UP4	UP5	UP6	UP7	UP8
	0.5	102.	5.					
\$#	UP9	UP10	UP11	UP12	UP13	UP14	UP15	UP16
Τ "	013	0110	0111	0112	0110	0111	0110	0110
\$#	SOFT	SOFSCL	LCIDAB	MAXPAR	SBOPT	DEPTH	BSORT	FRCFRQ
Υπ	SOFI	SOFSCI	LCIDAD	MAXIAN	55011	DELIII	DOORT	rncrng
\$#	PENMAX	THKOPT	SHLTHK	SNLOG	ISYM	I2D3D	SLDTHK	SLDSTF
ΥII	1.0	IIIIOII	ЭППТПК	DNIIOO	10111	12000	DEDIIII	DHDDII
	1.0							

The variables related to the user defined tiebreak condition are:

- OPTION: User tiebreak type. 101 ≤ OPTION ≤ 105.
- NHV: Number of history variables
- CT2CN: Ratio of tangential stiffness to normal stiffness.
- CN. Normal stiffness. If left blank, the penalty stiffness divided by the segment area is used (default).
- OFFSET: Not applicable to Mortar contact.
- NHMAT: Number of material history variables (in addition to plastic strain) to be accessible in the subroutine mortar usrtbrk
- NHWLD: Number of tied weld history variables to be read in the user tiebreak routine, assuming they have been carried over from a previous simulation.
- *UP1..UP16*: User parameters.

Note that the accumulated effective plastic strain for reference and tracked side is accessible even if NHMAT = 0.

Surfaces that are initially close enough will be tied. The tolerance distance for tying can be set by the PENMAX parameter (1.0 on the last row of the keyword example above).

## 7.2 Post processing user tiebreak condition

The tied status, as an indicator from 1 (meaning perfectly tied) to 0 (meaning completely released) can be output by setting NTIED =1 on Card 2 of \*DATABASE EXTENT INTFOR, and visualized in LS-PrePost, see Figure 38 for an example. The keyword \*DATABASE BINARY INTFOR FILE is required to specify the filename (intfor is recommended) and output frequency of the contact data.

# 7.3 Interface to the user defined tiebreak condition in the subroutine mortar\_usrtbrk

The condition for releasing the tied contact is defined in the subroutine mortar\_usrtbrk, which is found in the dyn21cnt.f Fortran file. The subroutine definition for implementing a user defined tiebreak condition is<sup>14</sup>:

```
subroutine mortar_usrtbrk(tid,dmg,pn,ps,iconv,dpndn,dpnds,dpsdn,dpsds,istif,prm,hst,whst,selm,melm,sprt,mprt,styp,mtyp,dn,ds,en,es,ts,tm,seps,meps,shis,mhis,time,dt)
```

For R14 and later versions of LS-DYNA, names of some of the parameters have changed:

The subroutine is called for each pair of reference-tracked contact segments. Note that the user subroutine is only called for the segments in consideration for tied contact, which is also related to the *PENMAX* parameter value. For those segments that are not tied, the subroutine is not called. The objective of the subroutine is to update the damage parameter dmg, where 0 indicates no damage (a complete tied contact) and 1 indicates a complete release of the tied contact. If required for implicit calculations (indicated by istif.ne.0), also the tangent stiffness matrix should be calculated.

An overview of the parameters to the subroutine is shown in Table 11.

<sup>&</sup>lt;sup>14</sup> Re-formatted from the original in dyn21cnt.f (where 12 continuation lines are used).



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Table 11. Overview of the arguments for the mortar usrtbrk subroutine

Argument From	m R14→	Description	Input / Output
tid		tiebreak interface id, integer between 101 and 105	Input
dmg		damage. should be increased between 0 (completely tied) and	Input/Output
		dmg = 1. indicates complete release	
		and 1 (completely released)	
pn		normal traction (unit: pressure)	Output
ps		tangential traction (unit: pressure)	Output
iconv		flag for converged step in implicit (check iconv.eq.0)	Input
dpndn	1	tangent of normal traction wrt normal separation	Output
		(unit: pressure / length) needed when istiff.ne.0	
dpnds	3	tangent of normal traction wrt to tangential separation	Output
		(unit: pressure / length) needed when istif.ne.0	
		NOTE: dpnds = dpsdn is required	
dpsdn	1	tangent of tangential traction wrt to normal separation	Output
		(unit: pressure / length) needed when istif.ne.0	
		NOTE: dpsdn = dpnds is required	
dpsds	3	tangent of tangential traction wrt to tangential separation	Output
		(unit: pressure / length) needed when istif.ne.0	
istif		flag if stiffness is needed for implicit, check istif.ne.0	Input
prm		input parameters	Input
hst		history variables	Input/Output
whst		Weld tie history variables from previous tied weld analysis	Input
	elm	element id for SURFA segment	Input
	elm	element id for SURFB segment	Input
	ort	part id for SURFA segment	Input
	ort	part id for SURFB segment	Input
styp tt	Ур		•
mtyp rt	Ур	Element type for SURFB segment ('beam ','solid','shell' or 'tshel')	Input
dn		Normal separation (unit: length, positive means tensile)	Input
ds		Tangential separation (unit: lengt, always positive)	Input
en		Normal stiffness (unit: pressure / length)	Input
es		Tangential stiffness (unit: pressure / length)	Input
ts tt		Temperature of SURFA segment	Input
tm tr		Temperature of SURFB segment	Input
	eps	Effective plastic strain on SURFA side	Input
	eps	Effective plastic strain on SURFB side	Input
shis this		material history variables for the SURFA segment	Input
	nis	material history variables for the SURFB segment	Input
time		simulation time	Input
dt		time step size	Input

From these arguments, it is possible to define a tiebreak condition depending on, for example,

- the effective plastic strain, or other history variables, of the involved materials,
- also, history variables associated with the contact segments on tracked and reference side, which can be inherited from a previous analysis using MORTAR\_TIED\_WELD.
- time, temperature, and contact pressure.

Note that it is required that the tangent stiffness matrix (stored in variables dpndn, dpnds, dpsdn, dpsds)

$$\mathbf{D} = \begin{pmatrix} \frac{\partial p_n}{\partial d_n} & \frac{\partial p_n}{\partial d_s} \\ \frac{\partial p_s}{\partial d_n} & \frac{\partial p_s}{\partial d_s} \end{pmatrix}$$

be symmetrical,  $\mathbf{D} = \mathbf{D}^{\mathrm{T}}$ , which in turn implies  $\frac{\partial p_n}{\partial d_s} = \frac{\partial p_s}{\partial d_n}$ . If a model with an unsymmetrical stiffness matrix is used, it must be symmetrized internally in the mortan userbork routine.

In the intfor file, 1 – dmg (see Table 11) is output as the "tied at top face" / "tied at bottom face" tie indicator, which can be visualized in LS-PrePost, see Figure 38.

## 7.4 Subroutine example

In this section, a basic tiebreak condition depending on a history variable of the materials in contacts will be implemented. This example requires revision R13-1674-g3b7bda8165 or later of LS-DYNA. It shall be stressed that this example is not intended for use in any kind of production analysis, and it may very well contain errors or flaws.

The damage will be mapped linearly from 0 to 1 when the material history goes from a lower threshold value thresh to an upper limit maxlim. For this, three parameters would be required: in addition to the limits also an identifier for the (tracked side) history variable to scale the damage.

The keyword interface will be:

```
*CONTACT_AUTOMATIC_SURFACE_TO_SURFACE_TIEBREAK_USER_MORTAR_ID
$#1 SURFA
           SURFB SURFATYP SURFBSTYP
                                                          SAPR
                                                                   SBPR
Specify what should be in contact
$#2
      FS
Specify friction
$#3
$# OPTION
                                      OFFSET NHMAT
              NHV
                                CN
                     CT2CN
                                                         NHWLD
                                           hisvarid
     103
     UP1
              UP2
$#
                               UP4
                                         UP5 UP6
                                                          UP7
                                                                   UP8
                       UP3
thresh maxlim hisvarid
     UP9 UP10
                               UP12
                                       UP13
                                                UP14
                                                          UP15
                                                                   UP16
$#
   SOFT
          SOFSCL
                    LCIDAB
                             MAXPAR
                                       SBOPT
                                                DEPTH
                                                         BSORT
                                                                 FRCFRO
$# PENMAX
           THKOPT
                     SHLTHK
                              SNLOG
                                        ISYM
                                                I2D3D
                                                        SLDTHK
                                                                 SLDSTF
Optional distance for tying
```

where blue text indicates that the user should input sensical data, and red text indicates values that should not be changed. By setting hisvarid = 0, the maximum plastic strain from either tracked or reference side material will scale the damage.

The first part of the subroutine mortar usrtbrk involving subroutine and variable declarations, follows:

subroutine mortar usrtbrk(tid,dmg,pn,ps,iconv,dpndn,dpnds,dpsdn,



```
dpsds, istif, prm, hst, whst, selm, melm, sprt, mprt, styp, mtyp,
1
2
      dn, ds, en, es, ts, tm, seps, meps, shis, mhis, time, dt)
 implicit none
 include 'nlqparm'
 include 'iounits.inc'
 integer tid
 real dmg
 real pn,ps
 integer iconv
 real dpndn, dpnds, dpsdn, dpsds
 integer istif
 real prm(*), hst(*), whst(*)
 integer selm, melm
 integer sprt, mprt
 character*5 styp, mtyp
 real dn, ds
 real en, es
 real ts, tm
 real seps, meps
 real shis(*), mhis(*)
 real time, dt
 real pd,prmdiff,seval
 integer ihvar
```

In the original dyn21cnt.f Fortran file of the usermat package, the comments after the subroutine declaration provide some documentation of the user tiebreak subroutine, regarding for example parameters. These comments are omitted here. The variable pd will be used to store the previous damage value (dmq) when the subroutine is called. The variable prmdiff will hold the difference between maxlim and thresh, and seval is the quantity to compare with (plastic strain or history variable value). The variable *ihvar* simply is the number of the history variable.

The main part of the subroutine starts by writing a message to the mes0\* files, to confirm what subroutine is active.

```
if (tid.eq.103) then
C --- add some initial diagnose print out
        if(time.le.1.E-3)then
         write(iomsg, *) ' --- mortar tiebreak 103'
          if (int(prm(3)).gt.0) then
           write(iomsq, *) ' using material history variable ',
     1
                                int(prm(3))
         else
           write(iomsg, *) ' using plastic strain'
         endif
         write(iomsq, *) ' damage scaled from ',prm(1),' to ',prm(2)
         write(iomsg, *) ' --- '
        endif
```

Then the linear, undamaged surface tractions are computed, and the reference quantity seval is evaluated:

```
pn=en*dn
ps=es*ds
```



C

```
pd = dmg
ihvar=int(prm(3))
if(ihvar.eq.0)then
   seval=max(seps, meps)
else
   seval=shis(ihvar)
endif
```

Then follows the damage calculation, where it is ensured that the damage cannot decrease, nor exceed unity.

```
prmdiff = max(1.E-5,abs(prm(2) - prm(1)))
if(seval.gt.prm(2)) then
   dmg = 1.0
elseif(seval.gt.prm(1)) then
   dmg = (seval-prm(1))/prmdiff
endif
dmg=max(pd, dmg)
dmg=min(dmg,1.)
```

The damage value is stored in the 1st history variable, and the damaged surface tractions are computed.

```
if(dmg.gt.0.)then
  hst(1)=dmg
  pn=(1.-dmg)*pn
  ps=(1.-dmg)*ps
endif
```

A message is output if the tied contact is fully released.

```
if (dmg.eq.1.and.pd.lt.1..and.iconv.eq.0) then
  write (iomsg,1) styp,selm,sprt,mtyp,melm,mprt,time
endif
```

Finally, the tangential stiffness is computed. Since there are no couplings between the normal and tangential components, we get  $\frac{\partial p_n}{\partial d_s} = \frac{\partial p_s}{\partial d_n} = 0$ .

which concludes this subroutine example. A simulation example using this subroutine is presented in Section 7.5.

### 7.5 LS-DYNA simulation example

In this example, a cantilever beam (similar to the example of Section 4.6.1, 4.6.2) with a weld-on reinforcement plate, purple in Figure 35, is subjected to severe overloading. In the first stage, a



prescribed vertical displacement of -140 mm and then +140 mm is applied at the bolt holes of the end bracket. In the second stage, longitudinal compression of 500 mm is applied to the deformed beam.

Elastic-plastic material properties typical for aluminum (E = 70 GPa, v = 0.31,  $\sigma_V$  = 140 MPa) are used. Contact is considered between the square beam and the cylindrical rigid support, using \*CONTACT AUTOMATIC SURFACE TO SURFACE MORTAR, as well as self-contact within the C-section beam (red in Figure 35). The side-plate reinforcement is attached to the C-section using a tiebreak contact (\*CONTACT\_AUTOMATIC\_SURFACE\_TO\_SURFACE TIEBREAK USER MORTAR) with user-defined release condition based on plastic strain in the materials involved, according to the subroutine of Section 7.4. Deformation results from implicit analysis are shown in Figure 36 - Figure 37. The tied indicator is shown in a fringe plot in Figure 38 and as a history plot in Figure 39. Some of the nodes release already during the initial vertical loading (t < 1) while most damage to the tied contact occur during the final axial compression (t > 4).

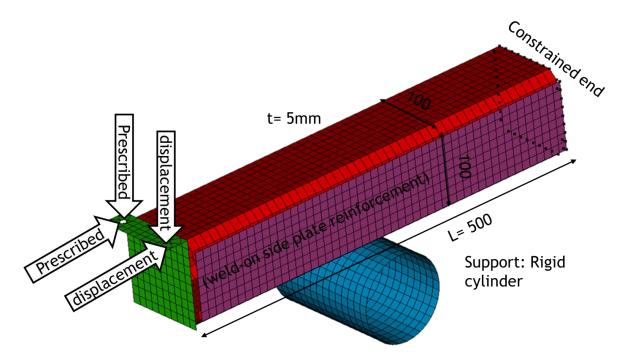


Figure 35. A C-profile (100 × 100 mm, t = 5 mm) cantilever beam with a weld-on reinforcement is subjected to prescribed displacement at the end bracket (green in the image) and contact with a rigid cylindrical support.

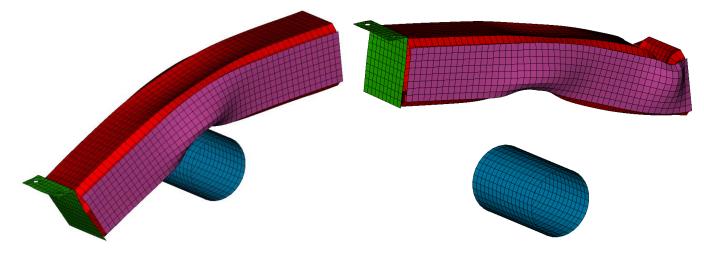


Figure 36. The left image shows the initial vertical displacement of -140 mm (t=1), and the right image shows the deformed configuration after the vertical displacement of +140 mm (t=3).

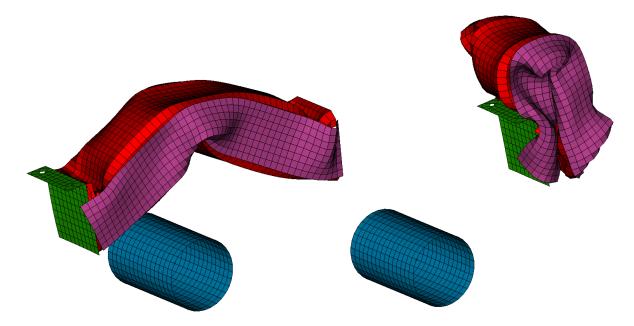


Figure 37. The left image shows an initial phase of the axial compression. The right image shows the final configuration.

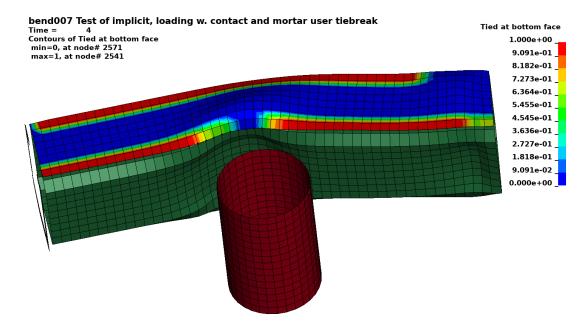


Figure 38. The fringe plot shows the tied indicator at the beginning of the axial compression stage from 0 (blue) to 1 (red).

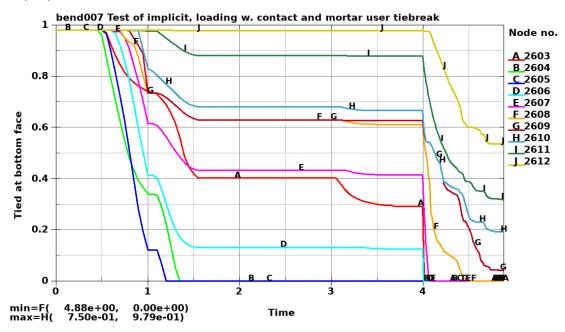


Figure 39. The evolution of the tied indicator in some of the nodes closes to the cylindrical support (marked by black dots in Figure 38).

## Loads interface

There are many ways of defining customized loading using the built-in LS-DYNA keywords. For example, the \*LOAD {OPTION} keywords for applying loads to nodes or segments also accept definition of the loading scale factor not only via curves (\*DEFINE CURVE) but also via functions of time, initial coordinate and current coordinate using \*DEFINE FUNCTION, via a C-like programming language (see Example 2 under \*DEFINE FUNCTION in Ref. [1], where the definition of a hydrostatic pressure is

demonstrated). By the \*LOAD\_SEGMENT\_{OPTION}\_NONUNIFORM keyword, pressure loading acting at a specified direction to the surface (not necessarily the normal direction) can be applied.

In addition to the built-in keywords for loading definitions, it is possible to provide user defined loadings by the keywords \*USER\_LOADING and \*USER\_LOADING\_SET and the corresponding Fortran subroutines loadud and loadsetud of the file dyn21.f in the usermat package. These subroutines also give access to nodal accelerations, velocities, and masses etc. These user defined loadings are supported by both the implicit and explicit mechanical solver of LS-DYNA. Under certain conditions, the user defined loading subroutines can also involve element deletion.

An overview (including both keyword and Fortran code examples) of the user defined loading options has previously been presented in Ref. [19]. A very brief description of the loadsetud subroutine can be found in Ref. [1] under the \*USER LOADING SET keyword.

## 8.1 Keyword interface to the user defined loadings

The keyword interface to the user defined loadings is given by

- \*USER LOADING, mainly for applying nodal loads and
- \*USER LOADING SET, for applying more general loadings, including temperatures.

The keyword input for the \*USER LOADING option is quite straight-forward:

*US	SER LOADING	j					
\$	PARM1	PARM2	PARM3	PARM4	PARM5	PARM6	PARM7
	1.	2041.	2.	100.	5.		

The variables of this keyword (PARM1, PARM2 etc.) are simply parameter values to be read by the user subroutine loadud. The documentation of the subroutine should preferably describe what each parameter corresponds to. This is discussed further in Sections 8.3 and 8.4.

An example of keyword input for the \*USER LOADING SET option follows:

*USE	R_LOADIN	G_SET						
\$	SID	LTYPE	LCID	CID	SF1	SF2	SF3	IDULS
	2	PRESSS	100	0	0.0	0 - 0	0.0	1

#### where the variables are:

- SID: The ID of the set that the loading should be applied to. The set type is determined by the loading type (LTYPE).
- LTYPE: Loading type, for example PRESSS for pressure on segments.
- LCID: Curve ID for scaling the loading.
- CID: Coordinate system ID. Default is the global coordinate system.
- SF1, SF2, SF3: Scale factors with different meanings depending on the loading type.
- IDULS: An ID number that is passed to the subroutine loadsetud.

These keywords can also be combined. By this, additional parameters for \*USER\_LOADING\_SET can be specified by the \*USER\_LOADING keyword input (see Section 8.5.3 for an example of this).

#### 8.2 Post processing user defined loadings

Currently, there are no options to output data or results from the user defined loading subroutines, other than writing text in the mes0\* and d3hsp-files. While for example forces applied using  $*LOAD_NODE_{OPTION}$  are output for post-processing in the bndout-file, the forces from the user defined loadings are not.

#### 8.3 Interfaces to the user-defined loading subroutines

There are two different subroutines for defining the loading:

- loadud, corresponding to the \*USER\_LOADING keyword, for applying nodal loadings by direct modification of the global load vector. This is a *scalar* subroutine.
- loadsetud, corresponding to the \*USER\_LOADING\_SET keyword, for providing a user defined load scale factor. This is a *vectorized* subroutine.

They are both found in the dyn21.f file. The parameter list for the user defined loading subroutine is:

```
subroutine loadud(fnod,dt1,time,ires,x,d,v,a,ixs,
  numels,ixb,numelb,idrflg,tfail,isf,p,npc,fval,iob,iadd64,numelh,
  ixh,nhex_del,nbeam_del,nshell_del,hexarray,hextim,bemarray,
  bemtim,shlarray,shltim,parm,numnp,fnodr,dr,vr,ndof,xmst,xmsr)
```

From R13 of LS-DYNA, some additional parameters related to thick shell elements were added:

```
subroutine loadud(fnod,dt1,time,ires,x,d,v,a,ixs,
. numels,ixb,numelb,idrflg,tfail,isf,p,npc,fval,iob,iadd64,numelh,
. ixh,nhex_del,nbeam_del,nshell_del,hexarray,hextim,bemarray,
. bemtim,shlarray,shltim,parm,numnp,fnodr,dr,vr,ndof,xmst,xmsr,
. numelt,ixt,ntsh del,tsharray,tshtim)
```

See also the attached Fortran file example for details. The objective of the loadud subroutine is to modify the force array fnod and/or the moment array fnodr, based on the input parameters of the array parm and other model data. An overview and brief description of the parameters to the subroutine is shown in

#### Table 12.

If the ires parameter has a negative value, it means that |ires| input parameters should be read in and stored in the parm array. NOTE! This must be done by explicit coding inside the loadud subroutine. A template for this is provided in the example code of the usermat package. See further Section 8.4.1 for an example.

In addition to allowing access to (almost) all nodal data of the current model (coordinates, velocities, accelerations, masses, etc.) the loadud subroutine may also trigger element deletion. This requires that the keyword \*DEFINE ELEMENT DEATH is present in the main keyword deck, for one or more elements



of the type to be deleted. The deletion time (variable <code>TIME</code> of the \*DEFINE\_ELEMENT\_DEATH keyword) can be set to a value that greatly exceeds the termination time for the run.

Table 12. The arguments to the subroutine loadud

time Current time step size linput  time Current problem time linput  ires Restart flag <sup>(1)</sup> linput / Output  A Coriginal nodal coordinates linput  d Nodal displacements linput  v Nodal velocities linput  a Nodal accelerations linput  ixs Shell element connectivities linput  numels Number of shell elements linput  numelb Number of beam elements linput  idrflg Nonzero if dynamic relaxation phase linput  tfail Shell element failure time linput / Output  isf Shell element failure flag (=1 → On) linput  p Load curve data pairs (abscissa, ordinate) linput  ppe Pointer into p  fval fval (1c) is the value of load curve ID 1c at the current time linput  ixh Solid element failure time linput  ixh Solid element connectivities linput  ixh Solid element connectivities linput  ixh Solid element connectivities linput  ixh Solid element one connectivities linput  ixh Solid element connectivities linput  nbeam_del if >0, element deletion option is active for solids linput  nshell_del if >0, element deletion option is active for shells linput  beamarray Time to delete solid elements, the value should be > time Output  beamarray Time to delete solid elements, the value should be > time Output  beamarray Time to delete shell elements, the value should be > time Output  shlarray Time to delete shell elements, the value should be > time Output  shlarray Time to delete shell elements, the value should be > time Output  shlarray Time to delete shell elements, the value should be > time Output  shlarray Time to delete shell elements, the value should be > time Output  shlarray Time to delete shell elements, the value should be > time Output  shlarray Time to delete shell elements, the value should be > time Output  shlarray Array for storing input parameters linput  nummp Number of nodal points linput  frod Nodal rotational displacements linput  nump Number of degrees of freedom per node in the solution phase (= 0) linput	Argument	Description	Input / Output
time Current problem time Input  ires Restart flag(¹) Input / Output  A Original nodal coordinates Input / Output  d Nodal displacements Input  v Nodal velocities Input  a Nodal accelerations Input  ixs Shell element connectivities Input  nume1s Number of shell elements Input  ixb Beam element connectivities(²) Input  ixb Beam element connectivities(²) Input  ixfail Shell element failure time Input / Output  ief Shell element failure flag (=1 → On) Input  p Load curve data pairs (abscissa, ordinate) Input  ipput fval (1c) is the value of load curve ID 1c at the current time Input  iob I/o buffer  iadd64  nume1h Number of solid elements Input  ixh Solid element connectivities Input  ixh Solid element deletion option is active for solids Input  nbex_del if >0, element deletion option is active for beam Input  shex_ray  if >0, element deletion is checked when time is ≥ hextim Input  beamarray Time to delete solid elements, the value should be > time Output  beamarray Time to delete shell elements, the value should be > time Output  shlarray Time to delete shell elements, the value should be > time Output  shlarray Time to delete shell elements, the value should be > time Output  shlarray Time to delete shell elements, the value should be > time Output  shlarray Time to delete shell elements, the value should be > time Output  shlarray Time to delete shell elements, the value should be > time Output  shlarray Time to delete shell elements, the value should be > time Output  shlarray Time to delete shell elements, the value should be > time Output  shlarray Time to delete shell elements, the value should be > time Output  shlarray Time to delete shell elements, the value should be > time Output  shlarray Time to delete on option is checked when time is ≥ shltim Input  year Array for storing input parameters  numnp Number of nodal points  input / Output  numnp Number of nodal points  input Input / Output  numnp Number of degrees of freedom per node in the solution phase (= 0)  Input	fnod	Global nodal forces	Input / Output
ires Restart flag <sup>(1)</sup> X Original nodal coordinates  Nodal displacements  Input  Nodal volocities  Input  Nodal volocities  Nodal accelerations  Input  Number of shell elements  Input  Number of shell elements  Input  Number of beam elements  Input  Number of beam elements  Input  Number of beam elements  Input  Nonzero if dynamic relaxation phase  Input  Isia Shell element failure time  Input  Isia Shell element failure flag (=1 → On)  Pointer into p  Input  Input	dt1	Current time step size	Input
X	time	Current problem time	Input
Nodal displacements   Input	ires	Restart flag <sup>(1)</sup>	Input / Output
Nodal velocities  Nodal accelerations  Nodal accelerations  Nodal accelerations  Number Shell element connectivities  Number of shell elements  Input  Number of shell elements  Number of shell elements  Number of beam elements  Input  Shell element failure time  Input / Output  Shell element failure flag (=1 → On)  Pounter into p  Load curve data pairs (abscissa, ordinate)  Input  Number of solid elements  Input  Input  Input  Input  Input  Input  Input  Solid element connectivities  Input  Input  Input  Insem_del if >O, element deletion option is active for solids  Input  Inshell_del if >O, element deletion option is active for shells  Input  Inshell_del if >O, element deletion option is active for shells  Input  Inshell_del if >O, element deletion option is active for shells  Input  Inshexarray  Time to delete solid elements, the value should be > time  Output  Deamarray  Time to delete beam elements, the value should be > time  Output  Deamarray  Time to delete beam elements, the value should be > time  Output  Deamarray  Time to delete shell elements, the value should be > time  Output  Shell element deletion is checked when time is ≥ bentim  Input  Number of solid points  Input  Number of solid points  Input  Number of solid points  Input  Number of nodal points  Input  Nodal rotational displacements  Input  Number of degrees of freedom per node in the solution phase (= 0)  Input  Number of degrees of freedom per node in the solution phase (= 0)	X	Original nodal coordinates	Input / Output
a Nodal accelerations  ixs Shell element connectivities  numels Number of shell elements  ixb Beam element connectivities <sup>(2)</sup> Number of beam elements  ixb Beam element connectivities <sup>(2)</sup> Number of beam elements  idrflg Nonzero if dynamic relaxation phase  input  idrflg Nonzero if dynamic relaxation phase  Input  input  Shell element failure flag (=1 → On)  p Load curve data pairs (abscissa, ordinate)  input  pput  pput  input  input  input  fval (1c) is the value of load curve ID 1c at the current time  iob i/o buffer  iadd64  numelh Number of solid elements  ixh Solid element connectivities  input  nhex_del if >0, element deletion option is active for solids  input  nshell_del if >0, element deletion option is active for beam  input  nshell_del if >0, element deletion option is active for shells  hexarray  Time to delete solid elements, the value should be > time  Output  beamarray  Time to delete beam elements, the value should be > time  Output  beamarray  Time to delete beam elements, the value should be > time  Output  beamarray  Time to delete beam elements, the value should be > time  Output  Shalarray  Time to delete shell elements, the value should be > time  Output  Shell element deletion is checked when time is ≥ bentim  Input  Shell element deletion is checked when time is ≥ shltim  Input  Number of nodal points  Input  Number of nodal points  Input  Nodal rotational displacements  Input  Nodal rotational velocities  Input  Number of degrees of freedom per node in the solution phase (= 0)  Input	d	Nodal displacements	Input
ixs Shell element connectivities Input  numels Number of shell elements Input  ixb Beam element connectivities <sup>(2)</sup> Input  numelb Number of beam elements Input  idrflg Nonzero if dynamic relaxation phase Input  tfail Shell element failure time Input / Output  isf Shell element failure flag (=1 → On) Input  p Load curve data pairs (abscissa, ordinate) Input  ppc Pointer into p Pointer into p Pointer into p Input  fval (1c) is the value of load curve ID 1c at the current time Input  iob i/o buffer  iadd64  numelh Number of solid elements Input  ixh Solid element connectivities Input  ixh Solid element deletion option is active for solids Input  nbeam_del if >0, element deletion option is active for shells Input  nshell_del if >0, element deletion option is active for shells Input  hexarray Time to delete solid elements, the value should be > time Output  beamarray Time to delete beam elements, the value should be > time Output  beamarray Time to delete beam elements, the value should be > time Output  beamarray Time to delete beam elements, the value should be > time Output  beamarray Time to delete beam elements, the value should be > time Output  beamarray Time to delete beam elements, the value should be > time Output  shlarray Time to delete shell elements, the value should be > time Output  shlarray Time to delete shell elements, the value should be > time Output  shlarray Time to delete shell elements, the value should be > time Output  shlarray Shell element deletion is checked when time is ≥ bemtim Input  yearm Array for storing input parameters Input/Output  nump Number of nodal points Input  fnodr Global nodal moments Input  Nodal rotational displacements Input  vr Nodal rotational velocities Input  ndof Number of degrees of freedom per node in the solution phase (= 0) Input	V	Nodal velocities	Input
numels         Number of shell elements         Input           ixb         Beam element connectivities <sup>(2)</sup> Input           numelb         Number of beam elements         Input           idrflg         Nonzero if dynamic relaxation phase         Input           tfail         Shell element failure time         Input / Output           isf         Shell element failure flag (=1 → On)         Input           p         Load curve data pairs (abscissa, ordinate)         Input           npc         Pointer into p         Input           fval (1c) is the value of load curve ID 1c at the current time         Input           iob         i/o buffer         Input           iadd64         Input         Input           numelh         Number of solid elements         Input           ixh         Solid element connectivities         Input           inhex_del         if >0, element deletion option is active for solids         Input           nbeam_del         if >0, element deletion option is active for shells         Input           hexarray         Time to delete solid elements, the value should be > time         Output           hextim         Solid element deletion is checked when time is ≥ hextim         Input           beamarray         Time to delete shel	a	Nodal accelerations	Input
ixb   Beam element connectivities   Input     numelb   Number of beam elements   Input     idrflg   Nonzero if dynamic relaxation phase   Input     tfail   Shell element failure time   Input / Output     isf   Shell element failure flag ( =1 → On)   Input     p	ixs	Shell element connectivities	Input
numelb         Number of beam elements         Input           idrflg         Nonzero if dynamic relaxation phase         Input           tfail         Shell element failure time         Input / Output           isf         Shell element failure flag (=1 → On)         Input           p         Load curve data pairs (abscissa, ordinate)         Input           npc         Pointer into p         Input           fval         fval (1c) is the value of load curve ID 1c at the current time         Input           iob         i/o buffer         Input           iadd64         Input         Input           numelh         Number of solid elements         Input           ixh         Solid element connectivities         Input           nhex_del         if >0, element deletion option is active for solids         Input           nbeam_del         if >0, element deletion option is active for shells         Input           nshell_del         if >0, element deletion option is active for shells         Input           nexarray         Time to delete solid elements, the value should be > time         Output           hextim         Solid element deletion is checked when time is ≥ bemtim         Input           beamarray         Time to delete shell elements, the value should be > time         Outpu	numels	Number of shell elements	Input
idrf1g   Nonzero if dynamic relaxation phase   Input    tfail   Shell element failure time   Input / Output    isf   Shell element failure flag (=1 → On)   Input    p   Load curve data pairs (abscissa, ordinate)   Input    pointer into p   Input    fval fval (1c) is the value of load curve ID 1c at the current time   Input    iob   i/o buffer   Input    iadd64   Input   Number of solid elements   Input    ixh   Solid element connectivities   Input    nhex_del   if >0, element deletion option is active for solids   Input    nbeam_del   if >0, element deletion option is active for shells   Input    nshell_del   if >0, element deletion option is active for shells   Input    nshell_del   if >0, element deletion option is active for shells   Input    hexarray   Time to delete solid elements, the value should be > time   Output    hextim   Solid element deletion is checked when time is ≥ hextim   Input    beamarray   Time to delete beam elements, the value should be > time   Output    bemtim   Beam element deletion is checked when time is ≥ bemtim   Input    shlarray   Time to delete shell elements, the value should be > time   Output    shlarray   Time to delete shell elements, the value should be > time   Output    shlarray   Time to delete shell elements, the value should be > time   Output    shlarray   Time to delete shell elements, the value should be > time   Output    shlarray   Time to delete shell elements, the value should be > time   Output    shlarray   Time to delete shell elements, the value should be > time   Output    shlarray   Time to delete shell elements, the value should be > time   Output    shlarray   Time to delete shell elements, the value should be > time   Output    shlarray   Time to delete shell elements, the value should be > time   Output    shlarray   Time to delete shell elements, the value should be > time   Output    shlarray   Time to delete shell elements, the value should be > time   Output    shlarray   Time to delete shell elements, the value should be > time   Output    shla	ixb	Beam element connectivities <sup>(2)</sup>	Input
tfail Shell element failure time	numelb	Number of beam elements	Input
isf Shell element failure flag (=1 → On) Input  P Load curve data pairs (abscissa, ordinate) Input  p Pointer into p Input  fval fval (1c) is the value of load curve ID 1c at the current time Input  iob i/o buffer  iadd64  numelh Number of solid elements Input  ixh Solid element connectivities Input  if >0, element deletion option is active for solids Input  if >0, element deletion option is active for shells Input  if >0, element deletion option is active for shells Input  nshell_del if >0, element deletion option is active for shells Input  hexarray Time to delete solid elements, the value should be > time Output  beamarray Time to delete beam elements, the value should be > time Output  beamarray Time to delete beam elements, the value should be > time Output  bemtim Beam element deletion is checked when time is ≥ bemtim Input  shlarray Time to delete shell elements, the value should be > time Output  shlarray Time to delete shell elements, the value should be > time Output  Shlarray Time to delete shell elements, the value should be > time Input  shlarray Time to delete shell elements, the value should be > time Output  shlarray Time to delete shell elements, the value should be > time Input  Shell element deletion is checked when time is ≥ shltim Input  parm Array for storing input parameters Input/Output  numnp Number of nodal points Input  fnodr Global nodal moments Input / Output  Nodal rotational displacements Input  vr Nodal rotational velocities Input  ndof Number of degrees of freedom per node in the solution phase (= 0) Input	idrflg	Nonzero if dynamic relaxation phase	Input
p Load curve data pairs (abscissa, ordinate) Input  npc Pointer into p fval fval (lc) is the value of load curve ID lc at the current time Input  iob i/o buffer  iadd64  numelh Number of solid elements Input  ixh Solid element connectivities Input  nhex_del if >0, element deletion option is active for solids Input  if >0, element deletion option is active for shells Input  nshell_del if >0, element deletion option is active for shells Input  nshell_del if >0, element deletion option is active for shells Input  hexarray Time to delete solid elements, the value should be > time Output  hextim Solid element deletion is checked when time is ≥ hextim Input  beamarray Time to delete beam elements, the value should be > time Output  bemtim Beam element deletion is checked when time is ≥ bemtim Input  shlarray Time to delete shell elements, the value should be > time Output  shlarray Time to delete shell elements, the value should be > time Input  shlarray Time to delete shell elements, the value should be > time Input  Shell element deletion is checked when time is ≥ shltim Input  Array for storing input parameters Input/Output  numnp Number of nodal points Input  fnodr Global nodal moments Input  vr Nodal rotational displacements  Input  Number of degrees of freedom per node in the solution phase ( = 0 Input	tfail	Shell element failure time	Input / Output
Pointer into p  fval fval(lc) is the value of load curve ID lc at the current time Input  iob i/o buffer  iadd64  numelh Number of solid elements Input  ixh Solid element connectivities Input  if >0, element deletion option is active for solids Input  nbeam_del if >0, element deletion option is active for shells Input  nshell_del if >0, element deletion option is active for shells Input  hexarray Time to delete solid elements, the value should be > time Output  hextim Solid element deletion is checked when time is ≥ hextim Input  beamarray Time to delete beam elements, the value should be > time Output  beamarray Time to delete beam elements, the value should be > time Output  beamarray Time to delete shell elements, the value should be > time Output  shlarray Time to delete shell elements, the value should be > time Output  shlarray Time to delete shell elements, the value should be > time Input  shlarray Time to delete shell elements, the value should be > time Output  shlarray Time to delete shell elements, the value should be > time Input  Shell element deletion is checked when time is ≥ shltim Input  Shell element deletion is checked when time is ≥ shltim Input  Number of storing input parameters Input/Output  numnp Number of nodal points Input  Vr Nodal rotational displacements Input  Number of degrees of freedom per node in the solution phase ( = 0 Input	isf	Shell element failure flag ( =1 $\rightarrow$ On)	Input
fval (lc) is the value of load curve ID lc at the current time input i/o buffer  iadd64  numelh Number of solid elements Input ixh Solid element connectivities Input if >0, element deletion option is active for solids Input if >0, element deletion option is active for shells Input if >0, element deletion option is active for shells Input if >0, element deletion option is active for shells Input if >0, element deletion option is active for shells Input if >0, element deletion option is active for shells Input hexarray Time to delete solid elements, the value should be > time Output hextim Solid element deletion is checked when time is ≥ hextim Input Deamarray Time to delete beam elements, the value should be > time Output bemtim Beam element deletion is checked when time is ≥ bemtim Input shlarray Time to delete shell elements, the value should be > time Output shltim Shell element deletion is checked when time is ≥ shltim Input Input Shell element deletion is checked when time is ≥ shltim Input Input Output of Slobal nodal moments Input/Output Input Output Input Output Input Nodal rotational displacements Input Nodal rotational displacements Input Nodal rotational velocities Input Number of degrees of freedom per node in the solution phase (= 0 Input	р	Load curve data pairs (abscissa, ordinate)	Input
iob i/o buffer  iadd64  numelh Number of solid elements Input  ixh Solid element connectivities Input  nhex_del if >0, element deletion option is active for solids Input  nbeam_del if >0, element deletion option is active for beam Input  nshell_del if >0, element deletion option is active for shells Input  hexarray Time to delete solid elements, the value should be > time Output  hextim Solid element deletion is checked when time is ≥ hextim Input  beamarray Time to delete beam elements, the value should be > time Output  bemtim Beam element deletion is checked when time is ≥ bemtim Input  shlarray Time to delete shell elements, the value should be > time Output  shlarray Time to delete shell elements, the value should be > time Output  shltim Shell element deletion is checked when time is ≥ shltim Input  parm Array for storing input parameters Input/Output  numnp Number of nodal points Input  fnodr Global nodal moments Input  Vr Nodal rotational displacements Input  Number of degrees of freedom per node in the solution phase ( = 0 Input	npc	Pointer into p	Input
numelh Number of solid elements Input  ixh Solid element connectivities Input  nhex_del if >0, element deletion option is active for solids Input  nbeam_del if >0, element deletion option is active for beam Input  nshell_del if >0, element deletion option is active for shells Input  nshell_del if >0, element deletion option is active for shells Input  hexarray Time to delete solid elements, the value should be > time Output  hextim Solid element deletion is checked when time is ≥ hextim Input  beamarray Time to delete beam elements, the value should be > time Output  bemtim Beam element deletion is checked when time is ≥ bemtim Input  shlarray Time to delete shell elements, the value should be > time Output  shlarray Time to delete shell elements, the value should be > time Output  shltim Shell element deletion is checked when time is ≥ shltim Input  parm Array for storing input parameters Input/Output  numnp Number of nodal points Input  fnodr Global nodal moments Input  Vr Nodal rotational displacements Input  Number of degrees of freedom per node in the solution phase (= 0 Input	fval	fval(lc) is the value of load curve ID lc at the current time	Input
nume1h       Number of solid elements       Input         ixh       Solid element connectivities       Input         nhex_del       if >0, element deletion option is active for solids       Input         nbeam_del       if >0, element deletion option is active for beam       Input         nshell_del       if >0, element deletion option is active for shells       Input         hexarray       Time to delete solid elements, the value should be > time       Output         hextim       Solid element deletion is checked when time is ≥ hextim       Input         beamarray       Time to delete beam elements, the value should be > time       Output         bemtim       Beam element deletion is checked when time is ≥ bemtim       Input         shlarray       Time to delete shell elements, the value should be > time       Output         shltim       Shell element deletion is checked when time is ≥ shltim       Input         parm       Array for storing input parameters       Input/Output         numnp       Number of nodal points       Input         dr       Nodal rotational displacements       Input         vr       Nodal rotational velocities       Input         ndof       Number of degrees of freedom per node in the solution phase (= 0)       Input	iob	i/o buffer	
ixh Solid element connectivities Input  nhex_del if >0, element deletion option is active for solids Input  nbeam_del if >0, element deletion option is active for beam Input  nshell_del if >0, element deletion option is active for shells Input  hexarray Time to delete solid elements, the value should be > time Output  hextim Solid element deletion is checked when time is ≥ hextim Input  beamarray Time to delete beam elements, the value should be > time Output  bemtim Beam element deletion is checked when time is ≥ bemtim Input  shlarray Time to delete shell elements, the value should be > time Output  shlarray Time to delete shell elements, the value should be > time Output  shltim Shell element deletion is checked when time is ≥ shltim Input  parm Array for storing input parameters Input/Output  numnp Number of nodal points Input  fnodr Global nodal moments Input  vr Nodal rotational displacements Input  Number of degrees of freedom per node in the solution phase ( = 0 Input	iadd64		
nhex_del if >0, element deletion option is active for solids  Input  nbeam_del if >0, element deletion option is active for beam  Input  nshell_del if >0, element deletion option is active for shells  Input  nshell_del if >0, element deletion option is active for shells  Input  hexarray Time to delete solid elements, the value should be > time  Solid element deletion is checked when time is ≥ hextim  Input  beamarray Time to delete beam elements, the value should be > time  Output  bemtim Beam element deletion is checked when time is ≥ bemtim  Input  shlarray Time to delete shell elements, the value should be > time  Output  shlarray Shell element deletion is checked when time is ≥ shltim  Input  parm Array for storing input parameters  Input  numnp Number of nodal points  Input  fnodr Global nodal moments  Input  Vr Nodal rotational displacements  Input  Number of degrees of freedom per node in the solution phase (= 0)  Input	numelh	Number of solid elements	Input
nbeam_del if >0, element deletion option is active for beam Input  nshell_del if >0, element deletion option is active for shells Input  hexarray Time to delete solid elements, the value should be > time Output  hextim Solid element deletion is checked when time is ≥ hextim Input  beamarray Time to delete beam elements, the value should be > time Output  bemtim Beam element deletion is checked when time is ≥ bemtim Input  shlarray Time to delete shell elements, the value should be > time Output  shltim Shell element deletion is checked when time is ≥ shltim Input  parm Array for storing input parameters Input/Output  numnp Number of nodal points Input  fnodr Global nodal moments Input / Output  dr Nodal rotational displacements Input  vr Nodal rotational velocities Input  ndof Number of degrees of freedom per node in the solution phase ( = 0 Input	ixh	Solid element connectivities	Input
nshell_del if >0, element deletion option is active for shells Input  hexarray Time to delete solid elements, the value should be > time Output  hextim Solid element deletion is checked when time is ≥ hextim Input  beamarray Time to delete beam elements, the value should be > time Output  bemtim Beam element deletion is checked when time is ≥ bemtim Input  shlarray Time to delete shell elements, the value should be > time Output  shltim Shell element deletion is checked when time is ≥ shltim Input  parm Array for storing input parameters Input/Output  numnp Number of nodal points Input  fnodr Global nodal moments Input  vr Nodal rotational displacements Input  Number of degrees of freedom per node in the solution phase ( = 0 Input	nhex_del	if >0, element deletion option is active for solids	Input
hexarray Time to delete solid elements, the value should be > time Output  hextim Solid element deletion is checked when time is ≥ hextim Input  beamarray Time to delete beam elements, the value should be > time Output  bemtim Beam element deletion is checked when time is ≥ bemtim Input  shlarray Time to delete shell elements, the value should be > time Output  shltim Shell element deletion is checked when time is ≥ shltim Input  parm Array for storing input parameters Input/Output  numnp Number of nodal points Input  fnodr Global nodal moments Input  dr Nodal rotational displacements Input  vr Nodal rotational velocities Input  ndof Number of degrees of freedom per node in the solution phase ( = 0 Input	nbeam_del	•	Input
hextim       Solid element deletion is checked when time is ≥ hextim       Input         beamarray       Time to delete beam elements, the value should be > time       Output         bemtim       Beam element deletion is checked when time is ≥ bemtim       Input         shlarray       Time to delete shell elements, the value should be > time       Output         shltim       Shell element deletion is checked when time is ≥ shltim       Input         parm       Array for storing input parameters       Input/Output         numnp       Number of nodal points       Input         fnodr       Global nodal moments       Input         dr       Nodal rotational displacements       Input         vr       Nodal rotational velocities       Input         ndof       Number of degrees of freedom per node in the solution phase ( = 0       Input	nshell_del	if >0, element deletion option is active for shells	Input
beamarray Time to delete beam elements, the value should be > time Output  bemtim Beam element deletion is checked when time is ≥ bemtim Input  shlarray Time to delete shell elements, the value should be > time Output  shltim Shell element deletion is checked when time is ≥ shltim Input  parm Array for storing input parameters Input/Output  numnp Number of nodal points Input  fnodr Global nodal moments Input  dr Nodal rotational displacements Input  vr Nodal rotational velocities Input  ndof Number of degrees of freedom per node in the solution phase ( = 0 Input	hexarray	Time to delete solid elements, the value should be $> time$	Output
bemtim       Beam element deletion is checked when time is ≥ bemtim       Input         shlarray       Time to delete shell elements, the value should be > time       Output         shltim       Shell element deletion is checked when time is ≥ shltim       Input         parm       Array for storing input parameters       Input/Output         numnp       Number of nodal points       Input         fnodr       Global nodal moments       Input / Output         dr       Nodal rotational displacements       Input         vr       Nodal rotational velocities       Input         ndof       Number of degrees of freedom per node in the solution phase ( = 0       Input	hextim	Solid element deletion is checked when time is $\geq$ hextim	Input
shlarray       Time to delete shell elements, the value should be > time       Output         shltim       Shell element deletion is checked when time is ≥ shltim       Input         parm       Array for storing input parameters       Input/Output         numnp       Number of nodal points       Input         fnodr       Global nodal moments       Input / Output         dr       Nodal rotational displacements       Input         vr       Nodal rotational velocities       Input         ndof       Number of degrees of freedom per node in the solution phase ( = 0 Input	beamarray	Time to delete beam elements, the value should be > time	Output
shltim       Shell element deletion is checked when time is ≥ shltim       Input         parm       Array for storing input parameters       Input/Output         numnp       Number of nodal points       Input         fnodr       Global nodal moments       Input / Output         dr       Nodal rotational displacements       Input         vr       Nodal rotational velocities       Input         ndof       Number of degrees of freedom per node in the solution phase ( = 0       Input	bemtim	Beam element deletion is checked when time is ≥ bemtim	Input
parm       Array for storing input parameters       Input/Output         numnp       Number of nodal points       Input         fnodr       Global nodal moments       Input / Output         dr       Nodal rotational displacements       Input         vr       Nodal rotational velocities       Input         ndof       Number of degrees of freedom per node in the solution phase ( = 0 Input	shlarray	Time to delete shell elements, the value should be > time	Output
numnp     Number of nodal points     Input       fnodr     Global nodal moments     Input / Output       dr     Nodal rotational displacements     Input       vr     Nodal rotational velocities     Input       ndof     Number of degrees of freedom per node in the solution phase ( = 0 Input	shltim	Shell element deletion is checked when time is $\geq$ shltim	Input
fnodr Global nodal moments Input / Output  dr Nodal rotational displacements Input  vr Nodal rotational velocities Input  ndof Number of degrees of freedom per node in the solution phase ( = 0 Input	parm	Array for storing input parameters	Input/Output <sup>(3)</sup>
dr     Nodal rotational displacements     Input       vr     Nodal rotational velocities     Input       ndof     Number of degrees of freedom per node in the solution phase ( = 0 Input	numnp	Number of nodal points	Input
Nodal rotational velocities Input  Number of degrees of freedom per node in the solution phase ( = 0 Input	fnodr	Global nodal moments	Input / Output
ndof Number of degrees of freedom per node in the solution phase ( = 0 Input	dr	Nodal rotational displacements	Input
	vr	Nodal rotational velocities	Input
in the initialization phase)	ndof	Number of degrees of freedom per node in the solution phase ( = 0	Input
in the initialization phase)		in the initialization phase)	
Reciprocal of nodal translational masses in solution phase Input	xmst	Reciprocal of nodal translational masses in solution phase	Input

xmsr	Reciprocal of nodal rotational masses in solution phase	Input
numelt	Number of thick shell elements	Input, from R13
ixt	Thick shell element connectivities	Input, from R13
ntsh_del	if >0, element deletion option is active for thick shells	Input, from R13
tsharray	Time to delete thick shell elements, the value should be > time	Input, from R13
tshtim	Thick shell element deletion is checked when time is $\geq$ tshltim	Input, from R13

Notes: (1) The ires parameter has special meanings, for example ires < 0 means that |ires| input parameters should be read. (2) To get also the third node defining beam orientation, set NREFUP = 1 on \*CONTROL OUTPUT. (3) The subroutine should populate the array during the initialization phase, no data passed by LS-DYNA.

The parameter list for the user subroutine loadsetud is:

subroutine loadsetud(time,lft,llt,crv,iduls,parm,nod,nnm1)

The parameter list for this subroutine is quite short, see Table 13 for an overview, since an approach involving extensive use of common declarations is used. The objective of the loadsetud subroutine is to provide a scale factor for the loading, to be stored in the udl array, based on the accessible data and the parameters of the array parm (which is read by the loadud subroutine). Note that the loading type defined by the ltype variable of the keyword \*USER LOADING SET is not passed to the subroutine loadsetud, but the value of the variable iduls can be passed as a load model ID for providing different user-defined loading models for different sets.

Table 13. The arguments to the subroutine loadsetud

Argument	Description	Input / Output
time	Current problem time	Input
lft, llt	Start, stop indices of arrays for vectorized input/output	Input
crv	Value of LCURV $^{(1)}$ at the current problem time	Input
iduls	ID of user loading set <sup>(1)</sup>	Input
parm	Array for storing input parameters <sup>(2)</sup>	Input
nod	internal node numbers	Input
nnm1	offset for node block	Input

Notes: (1) From the \*USER LOADING SET - keyword. (2) from the \*USER LOADING keyword.

In the subroutine loadsetud access to nodal coordinates, displacements, temperatures etc. is provided via a common block, see Table 14 for a brief overview.

Table 14. Some of the arrays accessible via common block declarations in the subroutine loadsetud. Note that the dimension of all arrays is nlq

Argument	Description	Input / Output
x1,x2,x3	Current coordinate of node or element center	Input
d1,d2,d3	Displacement of node or element center	Input
v1, v2, v3	Velocity of node or element center	Input
temp	temperature of node or element center	Input
udl(nlq)	Array for storing user defined load scale factor	Output

#### 8.4 Subroutine examples

In this Section, some basic examples of user defined loadings via the subroutines loadud and loadsetud are given. It shall be stressed that these examples are not intended for use in any kind of production analysis, and there may very well be errors or flaws in them.

#### 8.4.1 Example of subroutine loadud

In order to illustrate the definition of a user loading subroutine loadud, three different cases are considered:

- 1. Application of a nodal force controlled by a curve as a function of time. This is similar to the built-in \*LOAD NODE keyword.
- 2. Application of a nodal force in a fixed direction in space, proportional to the magnitude of the nodal displacement and scaled by a load curve as a function of time.
- 3. Application of nodal force, counteracting the displacement and scaled by a load curve as a function of time. This is similar to the built-in functionality of an \*ELEMENT DISCRETE spring.
- 4. Read in parameters for \*USER LOADING SET.

The user will have to select which load model to use, to which node the loading should be applied, a scale factor and a curve ID for scaling the loading as a function of time. From the \*USER\_LOADING keyword, the following variables will be used for models 1 - 3:

- P1: load model
- P2: node ID
- P3: Translational degree of freedom (1 3)
- P4: curve ID
- P5: scale factor.

The first part of the (pre R13) subroutine loadud follows, with subroutine and variable declarations:

```
С
      parameter (NPARM=1000)
      common/usrldv/parm(NPARM)
С
С
      integer*8 iadd64
      real*8 x
      real*8 d, dr
      dimension a(3,*), v(3,*), d(3,*), fnod(3,*), ixs(5,*), ixb(4,*),
     x(3,*), tfail(*), p(*), npc(*), fval(*), iob(*), ixh(9,*),
     . hexarray(*), bemarray(*), shlarray(*), parm(*), fnodr(3, *),
     . vr(3,*), dr(3,*), xmst(*), xmsr(*)
С
      integer nid, idcrv, nidof, kk, nodext
      real sclfac, dist, f1, f2, f3, dfv(3)
```

In the original dyn21.f Fortran file of the usermat package, the comments after the subroutine declaration provide some documentation of the user loading subroutine, regarding for example parameters and some details on element deletion. These comments are omitted here.

The next part of the subroutine is active in the initialization phase. It reads in the parameter values (P1, P2, etc.) of the \*USER LOADING keyword and stores them in the array parm. Also, some messages are written in the mes 0 \* files to confirm what parameter values are read. This part was taken (more or less) from the original example subroutine provided in the usermat package.

```
if (ires.lt.0) then
     n=abs(ires)
     write (iomsg, 1030)
     call prludparm(0,parm,0,0)
     mssg='reading user loading subroutine'
     if (longs) then
       do 11 i=1, n, 8
       call gttxsg (txts,lcount)
       read (txts, '(8e20.0)', err=400) (parm(j), j=i, min(i+3, n))
       write(iomsg, 1040) (j, parm(j), j=i, min(i+3, n))
       call prludparm(1,parm,i,min(i+3,n))
11
       continue
     else
       do 10 i=1, n, 8
       call gttxsg (txts,lcount)
       read (txts, 1020, err=400) (parm(j), j=i, min(i+7, n))
       write (iomsg, 1040) (j, parm(j), j=i, min(i+7, n))
       call prludparm(1,parm,i,min(i+7,n))
10
       continue
     endif
     write (iohsp, 1050)
     call prludparm(2,parm,0,0)
     return
   endif
   if (ndof.eq.0) return
   if(parm(1).eq.4.) return
```

The last two rows of this part will return from the subroutine in case LS-DYNA is in the initialization phase (indicated by ndof = 0) or the load model 4 is chosen, in which case only the parameter reading

C

should be active. The final part of the subroutine applies the nodal forces, depending on what load model is selected. First, the internal<sup>15</sup> node ID to which the force should be applied, must be obtained.

```
nodext = int(parm(2))
nid =lqfe(nodext,1)
```

For going from the user-defined node ID given by the variable P2 of the \*USER\_LOADING keyword to the node ID used by LS-DYNA internally, the lqfe function must be used in an mpp implementation. In case the particular node ID is not accessible by the current mpi thread, lqfe will return a value  $\leq 0$  (while use of lqf or lqf8 will cause Error termination. However, lqf8 is required for use with smp/LS-DYNA). Then follows some conversion of the entries of the parm array to more useful variables.

```
if (nid.gt.0) then
  nidof = int(parm(3))
  idcrv = lcids(int(parm(4)))
  sclfac = parm(5)
  if(sclfac.eq.0.0) sclfac = 1.
```

By the last row, a similar behavior to the built-in  $*LOAD_{-}$  ... keywords is obtained, since the load scale factor is reset to a default value of 1 in case zero or no value is input by the user. Then follows the actual modifications of the force vector, corresponding to the different loading models:

```
if(parm(1).eq.1.)then
          fnod(nidof, nid) = fnod(nidof, nid) + fval(idcrv)*sclfac
        elseif(parm(1).eq.2.)then
          dist = sqrt(sum(d(1:3,nid)**2))
          fnod(nidof, nid) = fnod(nidof, nid) -
               fval(idcrv)*abs(sclfac)*dist
        elseif(parm(1).eq.3.)then
          dfv(1:3) = -d(1:3, nid) *abs(sclfac) *fval(idcrv)
          fnod(1:3, nid) = fnod(1:3, nid) + dfv(1:3)
        else
          cerdat(1) = 'Unsuported user loading model'
          call lsmsg(3,MSG SOL+1447,ioall,ierdat,rerdat,cerdat,0)
        endif
      endif
C
      return
```

In case another load model than 1, 2, 3 or 4 is requested, the subroutine will trigger an error termination. Some examples of simulations using this loadud subroutine are provided in Section 8.5.1 and 8.5.2.

<sup>&</sup>lt;sup>15</sup> In Table 7 some other useful functions for converting between internal and keyword-input numbering are listed.



#### 8.4.2 Example of subroutine loadsetud

As an example of the subroutine loadsetud, the application of a hydrostatic (buoyancy) loading was implemented. This will require the user to input, via the \*USER\_LOADING\_SET,

- SID: a segment set ID,
- LTYPE: PRESSS to specify pressure on a segment set,
- LCID: a curve ID for scaling the loading,

and via the \*USER LOADING keyword the parameters

- P1: 4.
- P2: the direction of the gravity, in the global coordinate system, (1 for X-direction, 2 for Y and 3 for Z which also is the default)
- P3: density of the fluid,
- P4: the gravitational acceleration,
- P5: the reference level for zero pressure.

The coding for the loadud subroutine, required for reading in the parameters P1 – P5 to the parm array, was already presented in Section 8.4.1. The first part of the subroutine loasetdud follows, with subroutine, common block and variable declarations:

```
subroutine loadsetud(time,lft,llt,crv,iduls,parm,nod,nnm1)
С
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Сl
  ______
c| Copyright 1987-2008 Livermore Software Tech. Corp
c| All rights reserved
C*********************
С
     Input (not modifiable)
C
C
     time : analysis time
     x : coordinate of node or element center
С
          : displacement of node or lement center
С
      v : velocity of node or lement center
C
С
     temp: temperature of node or element center
      crv : value of LCURV at time=time
С
     iduls: id of user loading set
С
     parm : parameters from user_loading
nod : internal node numbers
С
С
     nnm1 : offset for node block
С
С
    Output (defined by user)
     udl : user-defined load curve value
С
     include 'nlqparm'
     include 'iounits.inc'
C
    common/aux8loc/
    & x1(nlq), x2(nlq), x3(nlq), v1(nlq),
```

<sup>&</sup>lt;sup>16</sup> Hydrostatic loading in LS-DYNA in general does not require the coding of a user subroutine but can be achieved by the \*DEFINE FUNCTION keyword (see the example of Ref. [1]).



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```
& v2(nlq),v3(nlq),d1(nlq),d2(nlq),
& d3(nlq),temp(nlq),udl(nlq),
& xx11(nlq),xx21(nlq),xx31(nlq),
& xx12(nlq),xx22(nlq),xx32(nlq),
& xx13(nlq),xx23(nlq),xx33(nlq),
& xx14(nlq),xx24(nlq),xx34(nlq),
& xctr(nlq),yctr(nlq),zctr(nlq),
& f(nlq), tr1(nlq), tr2(nlq), tr3(nlq)
c
    dimension parm(*),nod(*)
C
    integer dof, kk
    real grav, rho, refl, fact, cord(nlq)
```

In this case, the original comments of the dyn21. f file are kept. Then follows a translation of the values of the parm array to explicit variables:

```
dof = int(parm(2))
rho = parm(3)
grav = parm(4)
refl = parm(5)
```

In order to simplify the final calculations, an array cord is assigned the current coordinate corresponding to the direction of the gravity:

```
C --- z is the default direction
    cord(1:nlq) = x3(1:nlq)
    if(dof.eq.1) then
        cord(1:nlq) = x1(1:nlq)
    elseif(dof.eq.2) then
        cord(1:nlq) = x2(1:nlq)
    endif
```

And then the final calculation of the hydrostatic force and the corresponding load factor udl follows

```
do kk=lft,llt
  fact = max(0., (refl-cord(kk))*grav*rho)
  udl(kk) = crv*fact
  enddo
```

which concludes the user subroutine.

## 8.5 LS-DYNA simulation examples

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In this Section, some LS-DYNA simulation examples of the user defined loading options are presented.

#### 8.5.1 Nodal force by load curve

An L-shaped beam, see Figure 40, is fully constrained at its base and subjected to transverse loading (5 kN). The tip deflection when the loading is applied by the built-in keyword \*LOAD\_NODE and the user defined loading is compared in Figure 41. For the explicit analyses, the Y-displacement of the tip of the beam is identical for the two different load application methods. The differences between the implicit static analysis and the explicit analyses are explained by the dynamic effects induced by the ramp-up



time of 20 ms. It is concluded that the coding of Section 8.4.1 and the related interfaces of the usermat package gives an equivalent result as the built-in load application functionality.

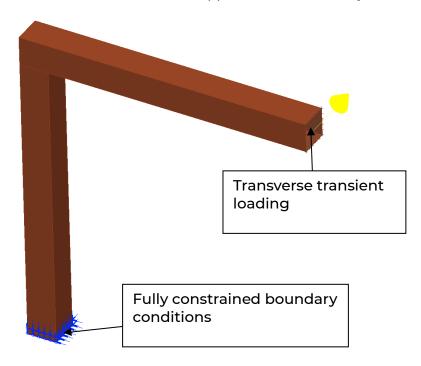


Figure 40. An L-shaped beam is fully constrained at its base and subjected to transverse loading at the center node of a \*CONSTRAINED NODAL RIGID BODY.

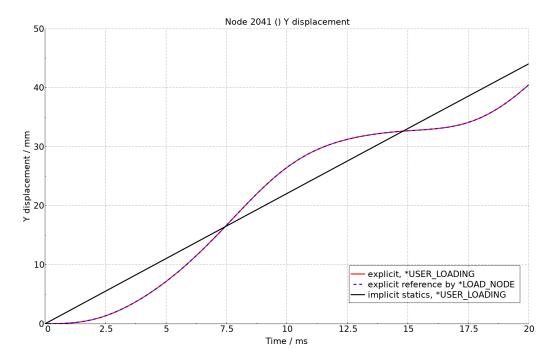


Figure 41. Results of LS-DYNA explicit and implicit simulations where the load is applied using \*LOAD\_NODE or \*USER LOADING.

#### 8.5.2 Nodal force proportional to nodal displacement

In the second set of example analyses, the load model 3 as described in Section 8.4.1 is compared to a model version where a discrete spring element is used to apply a force counteracting the nodal displacement. The geometry of Figure 40 is used also in this case, and the loading (spring element) is applied at the center node of the constrained nodal rigid body at the end of the horizontal beam part. A transverse displacement of 40 mm is ramped-up during using \*BOUNDARY\_PRESCRIBED\_MOTION with a death time of 20 ms, and the vibration motion of the L-beam is studied. If a constant curve is used, the user defined loading version gives the same tip deflection as the version using the built-in feature \*ELEMENT\_DISCRETE, see Figure 42. It is again concluded that the coding of Section 8.4.1 and the related interfaces of the usermat package gives an equivalent result as the built-in load functionality.

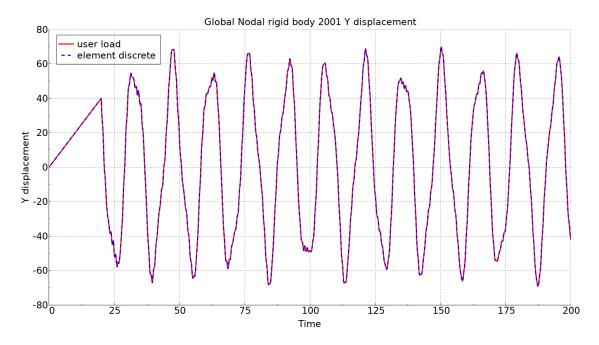


Figure 42. Comparison of tip displacement from the \*USER\_LOADING version (solid red curve) and the discrete element version (dashed blue curve).

By defining a non-constant curve for scaling the force, see for example the left image of Figure 43, corresponding to a discrete spring with time-varying stiffness, functionality that is not present by the built-in keywords is obtained. Clearly, the vibrational results are influenced by this time-varying stiffness, see the right image of Figure 43.

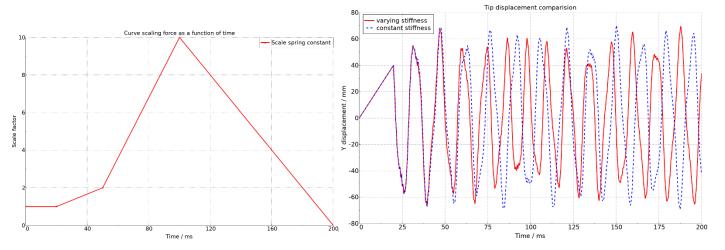


Figure 43. The left image: Curve for scaling the force as a function of time, corresponding to a varying spring stiffness. The right image: results comparison.

#### 8.5.3 Hydrostatic pressure loading

In this example, the user defined hydrostatic loading as described in Section 8.4.2 is compared to the built-in keywords \*LOAD\_SEGMENT\_SET and \*DEFINE\_FUNCION in an implicit, transient dynamic, analysis. The geometry of this example is shown in Figure 44, where the reference level is outlined as a semi-transparent plane. The Z-displacement of node ID 5375 is compared in Figure 45. It is concluded that the results of the different load application techniques are very close.

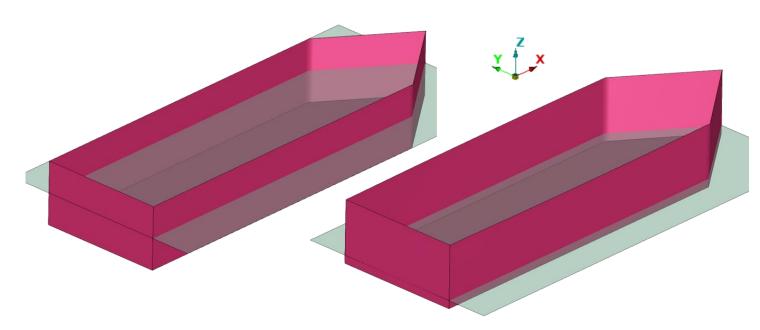


Figure 44. Geometry for the hydrostatic loading example. The left image shows the initial (guessed) configuration, and the right image shows the (equilibrium) position after 20 seconds. The reference level is indicated by a semi-transparent plane.

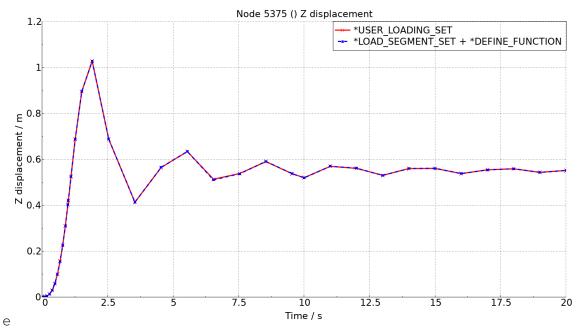


Figure 45. Comparison of the Z-displacement of the node ID 5375 from the simulation with user defined loading (solid red curve) and the simulation using built-in load application functionality (dashed blue curve).

#### 9 Other user interfaces

In this section some other possibilities for user defined interaction with LS-DYNA are listed. Some of them will be described in more detail in coming revisions of this document.

- For wear analysis (\*CONTACT\_ADD\_WEAR) a customized wear law can be defined by setting WTYPE < 0 and using the subroutine usrwear in dyn21cnt.f.
- For non-Mortar tiebreak contacts (\*CONTACT\_AUTOMATIC\_SURFACE\_TO\_SURFACE\_TIEBREAK\_USER), custom tie failure conditions can be defined using the subroutines utb101, ..., utb105 in dvn21cnt.f.
- User defined thermal conductance for the 3D thermal contacts can be defined by the subroutine usrhcon in dyn21.f. It is invoked by the keyword \*USER INTERFACE CONDUCTIVITY.
- Shell and solid elements can be defined by the user, either via the keywords

  \*DEFINE\_ELEMENT\_GENERALIZED\_{SHELL/SOLID} or via the user-defined interfaces of

  subroutines ushl\_byyy in dyn21ushl.f and usld\_bzzz in dyn21usld.f, see Appendix C of Ref.

  [1], and Ref. [18]
- User defined damage / failure criteria can be defined for some materials by the subroutine  $matusr_24$  in dyn21.f. The user define failure criteria is then invoked by setting FAIL < 0 for the material types: 24, 36, 114, 123, 124, 133, 155, 182, 225, 238, 243, 251, 255. Also for MATIO3, by the subroutine  $matusr_103$ .
- For interacting with other solvers in coupled analysis, the keyword \*COUPLE\_TO\_OTHER\_CODES can be used, which requires the family of user subroutines in the Fortran file couple2other user.f.
- For implicit analysis, it is possible (from R11) to provide a user-defined linear equation solver by setting LSOLVR = 90 on \*CONTROL IMPLICIT SOLVER.

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## 10 Executing user-compiled LS-DYNA binaries under Windows

In some cases, the execution of a user-compiled LS-DYNA executable may be prevented by the different security measures of Windows 10. The text of this Appendix is based on Ref. [27], and describes the background and possible resolution to such situation.

To defend Windows 10 computers against unsafe or malicious software being executed or installed typically several approaches are used, including:

- Microsoft Defender SmartScreen (MDS)
- Microsoft anti-virus software (AV)
- Third party anti-virus software (AV)

MDS and AV will typically prevent execution of any executable that is not white-listed and/or code signed traceable back to a trusted root authority.

An independent software vendor will typically white-list and code sign all their executables and software installers with Microsoft and major third-party AV-software products to avoid issues when the end user is to install or execute the software from the ISV.

However, developing user-defined features in LS-DYNA in a Windows environment will result in a new, customized LS-DYNA binary (.exe – file) and therefore AV and/or MDS may prevent execution. In turn, this makes it impossible to make use of the user-defined features with LS-DYNA. A solution to this needs to be sought by cooperation between the engineers developing user-defined features and the department responsible for Windows IT security and management at each specific company (or university). Some of the following approaches are typically applied:

- 1. Trusted users are allowed to locally white-list the custom LS-DYNA executables at the company or on specified computers.
  - This method works well, though it can have some IT-security implications.
- 2. The policy to require white-listed/signed code is disabled on selected computers and/or for trusted users.
  - This method works well, though it can have some IT-security implications that are larger than method 1.
- 3. Allow trusted users to override warnings or blockings from the AV software or MDS for each execution.
  - This is rarely used with LS-DYNA because it prevents the use of queuing systems with LS-DYNA and that reduces work productivity – sometimes considerably.
- 4. The custom LS-DYNA executable is white-listed at the relevant AV software companies and with MDS (code signing).
  - This is usually not doable as the during development of user-defined features as these
    are rebuilt many times per day and the white-listing process is lengthy.



Keywords: Ansys LS-DYNA; user-defined interfaces; Fortran

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