

LS-TaSC™ Product Status

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Overview

- Multipoint approach
 - Motivation
 - Global vs. local variables
 - Examples
- Other new features
- Current development

Multipoint Method

Motivation

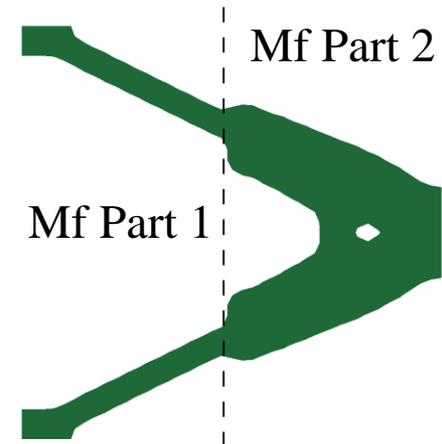
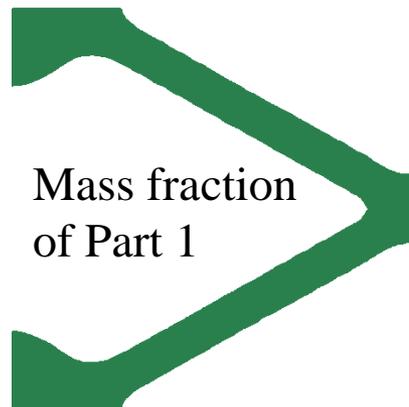
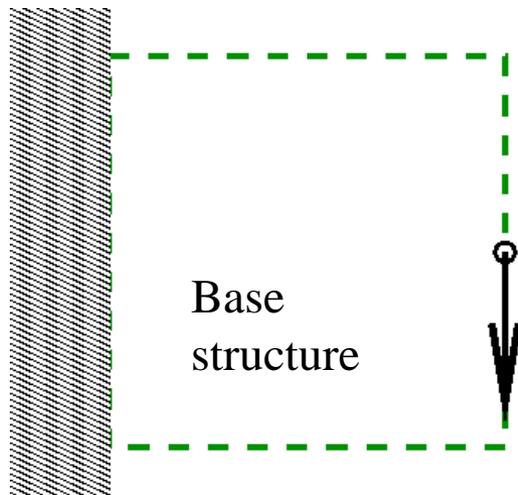
- Topology optimization for non-linear, dynamic load cases
 - Design Sensitivity Analysis too expensive
 - Heuristic methods (Hybrid Cellular Automata, Evolutionary Topology Optimization, Prescribed Plastic Strain/Stress, ...)
 - Global constraints possible (force, displacement, ...)
 - Constraints violated → increase or decrease mass to satisfy constraint
- Constrained topology optimization (multi-disciplinary problems including crash load cases)
 - Multipoint method
 - Allows more general constraints

Multipoint Method – Basic Idea

- Design Sensitivities
 - impracticable for standard topology variables,
 - but possible for a few global variables
- Gradient-based optimization methods, metamodel methods, ... with respect to global variables
 - Used in constrained optimization

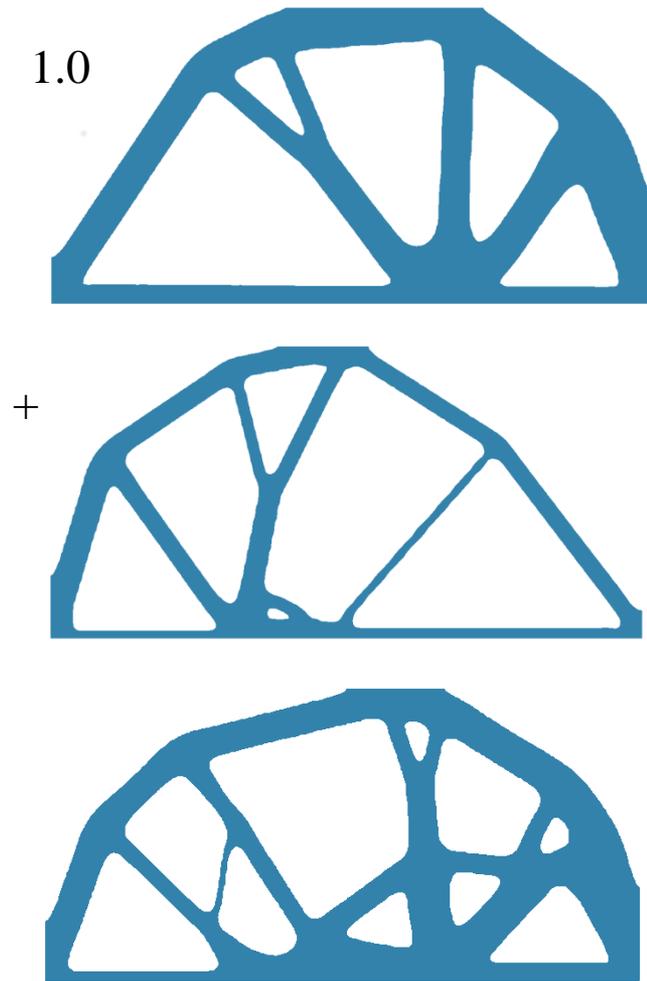
Multipoint Method - Variables

- Local variables
 - Amount of material in an element
 - Global variables
 - Part mass fraction
 - Load case weights
 - Geometry kernels
 - Reduced bases
- } Currently available in LS-TaSC



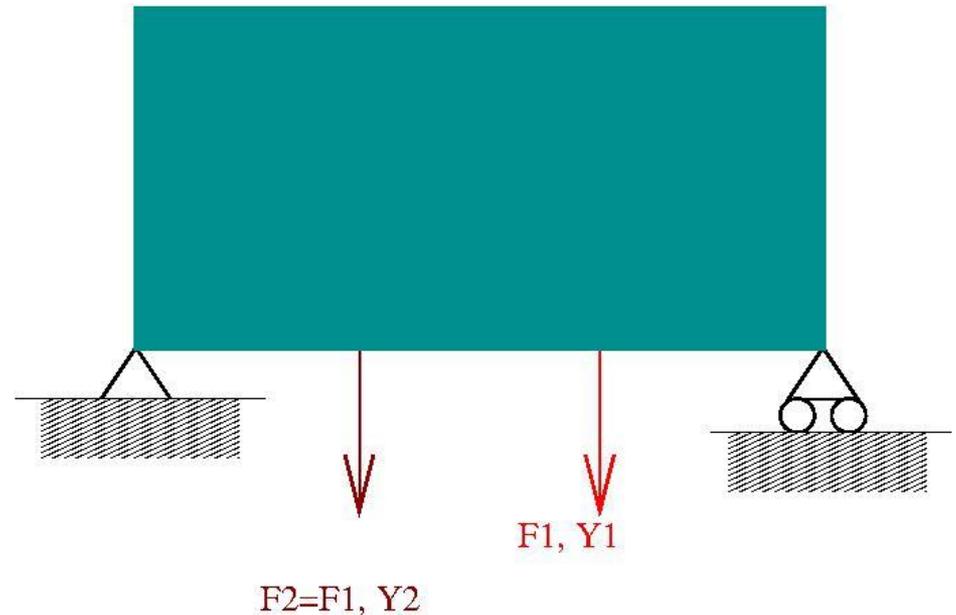
Multipoint Method - Variables

- Load case weights
 - Critical design variable for the analysis of multiple load cases
 - Controls relative values of constraints, as opposed to the absolute values
 - Variations weighing a specific time step are also possible



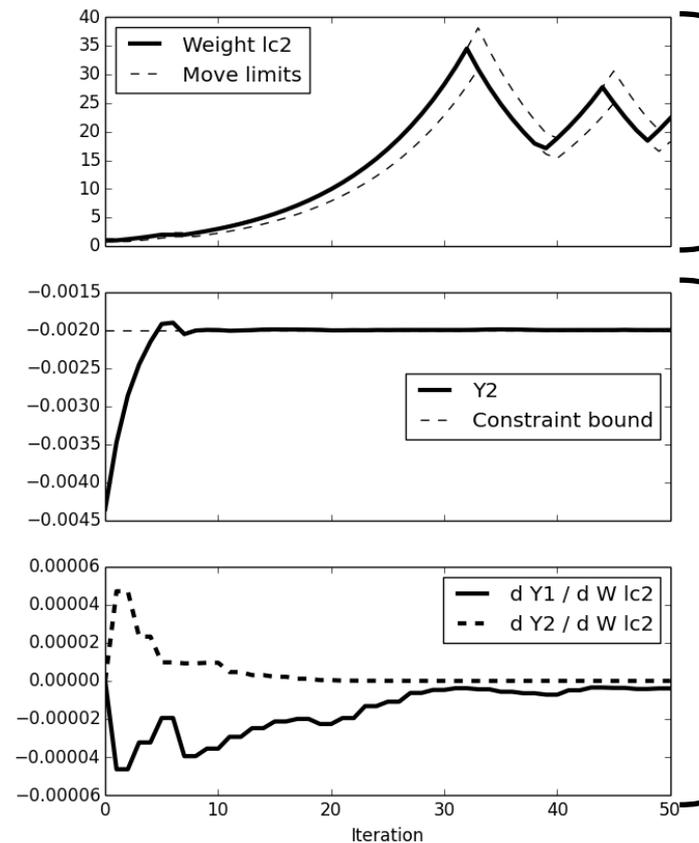
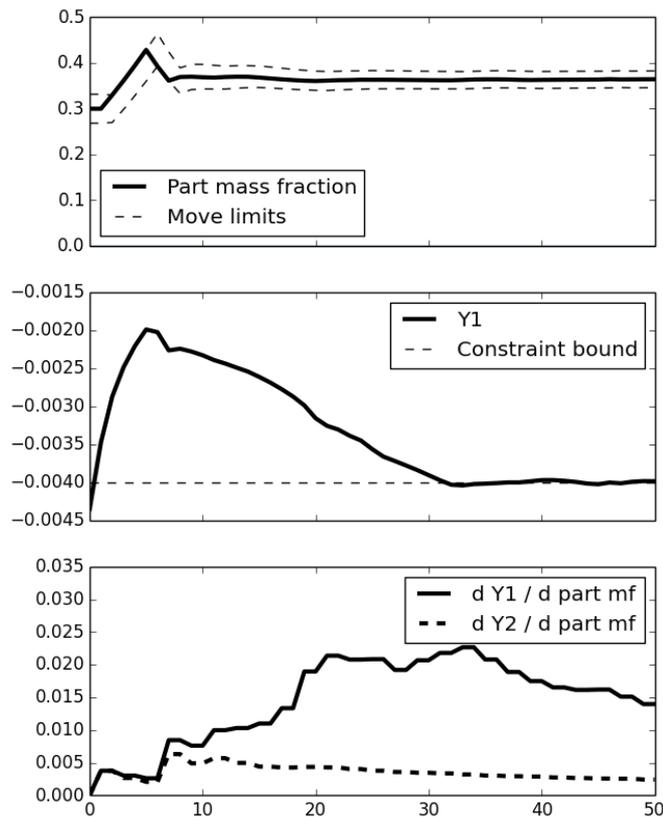
Example

- Two Load Cases
 - $F1 = F2$
- Objective
 - Minimize Mass
- Constraints
 - $Y1 > -0.002$
 - $Y2 > -0.004$
 - Asymmetric results from a symmetric setup are therefore required
- Global variables
 - Part mass
 - Load case weight ratio



Example - Results

■ Design Histories



Global variables

Constraints:
values and
derivatives

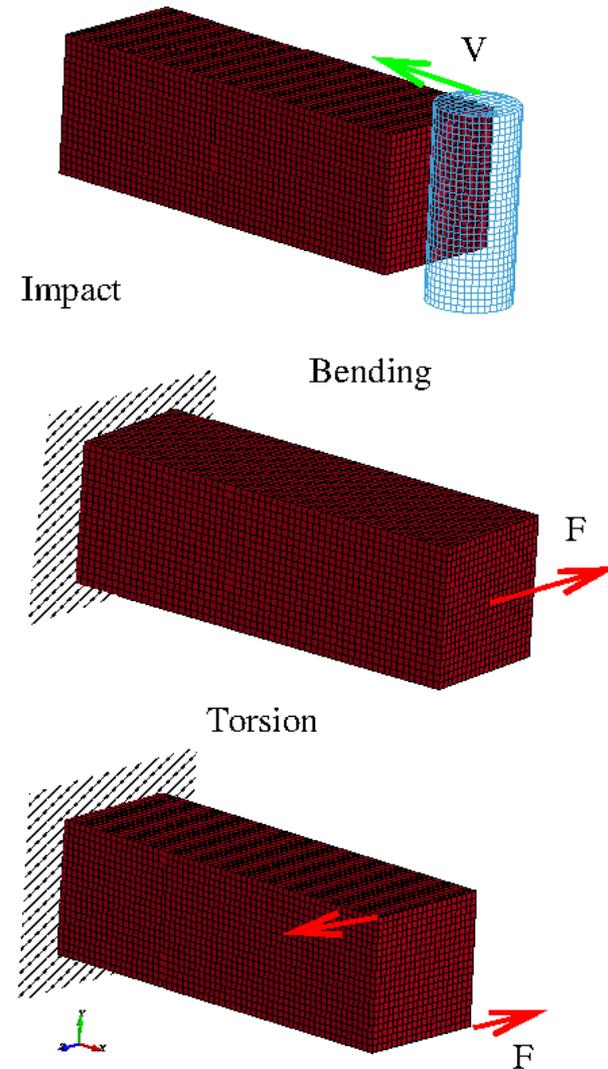
Example - Results

- Final topology
→ asymmetric due to the asymmetric constraints



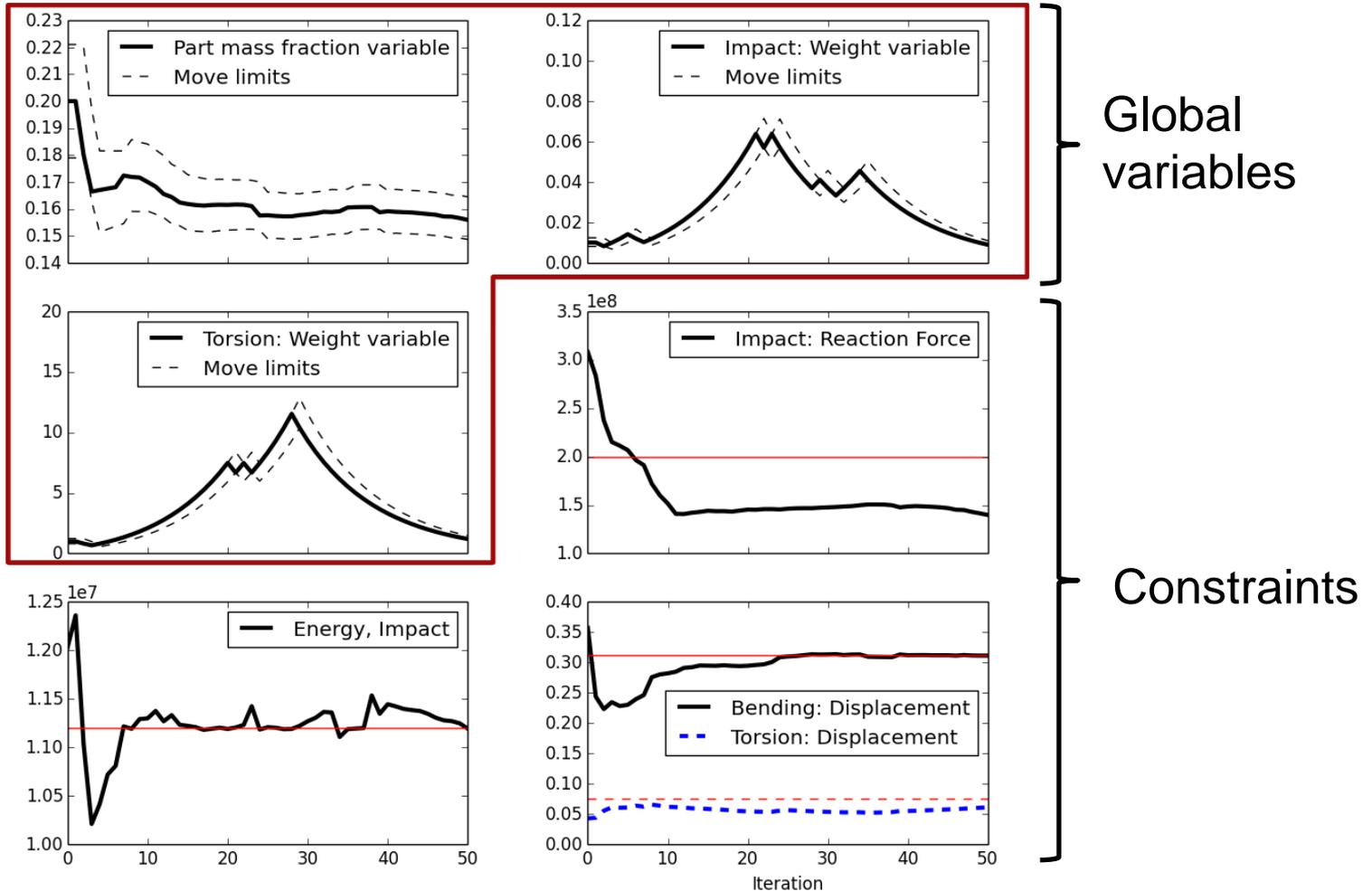
Multi-disciplinary Example

- Three load cases
 - Impact: Constraints
 - Energy absorption $> 11.2e6$
 - Reaction force $< 200e6$
 - Linear bending: Constraint
 - Displacement < 0.3125
 - Linear torsion: Constraint
 - Displacement < 0.075
- Global variables
 - Part mass fraction
 - Load case weights



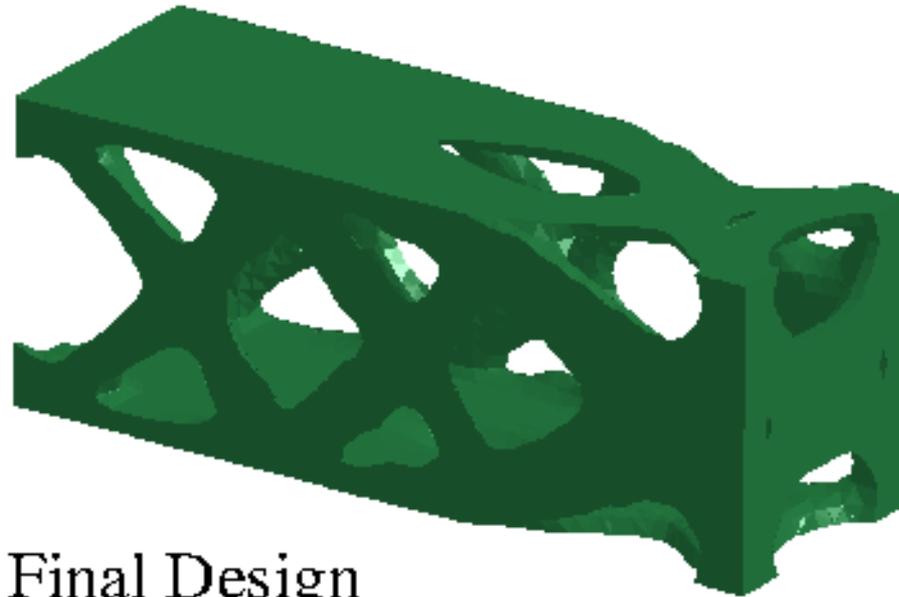
Multi-disciplinary Example

■ Design Histories



Multi-disciplinary Example

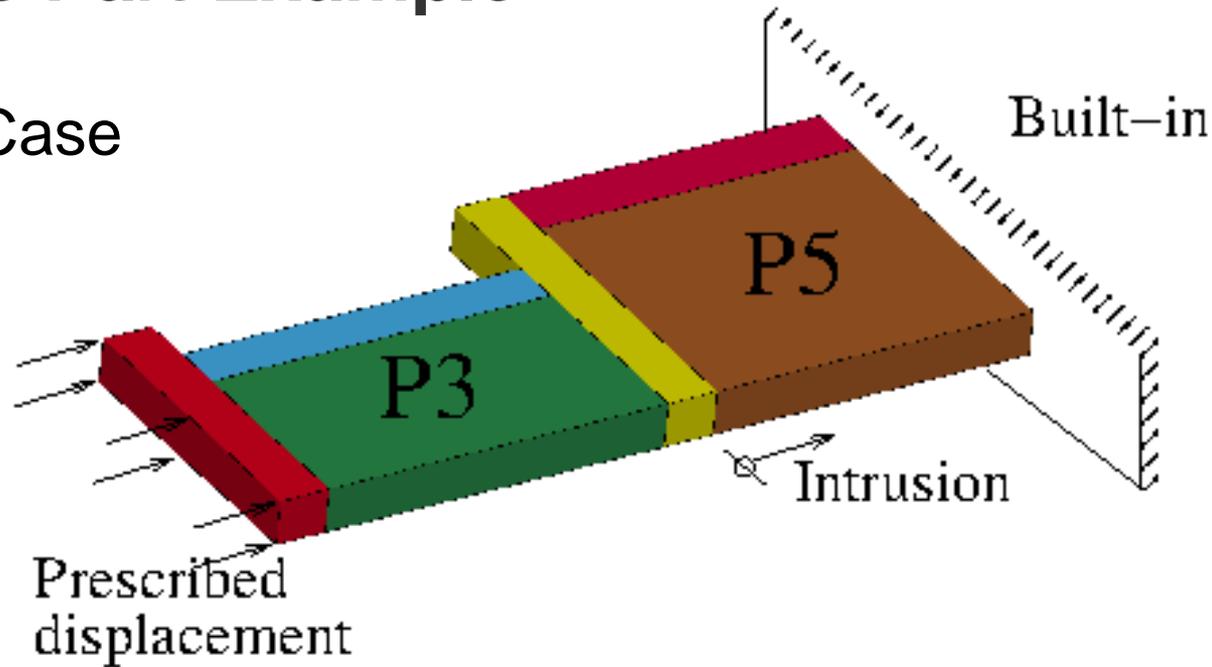
- Final topology



Final Design

Multiple Part Example

- Load Case



- Objective minimize mass

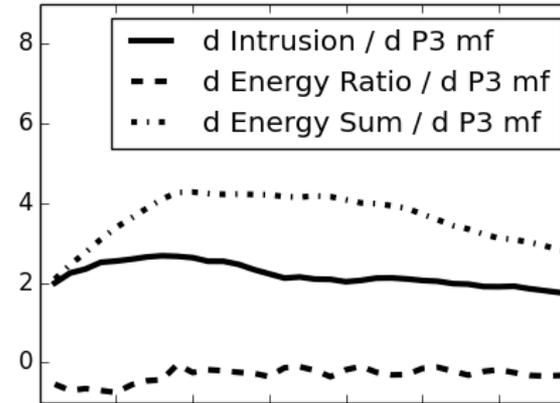
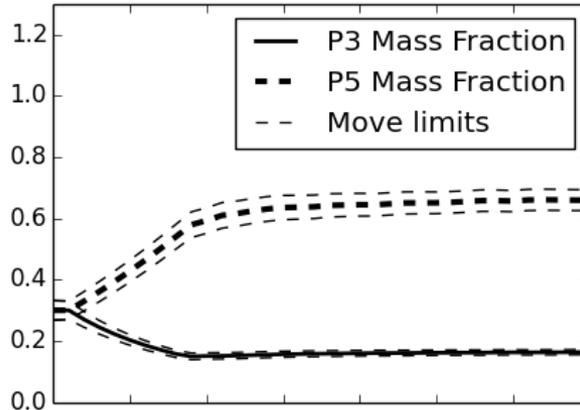
- Constraints

- Intrusion: $XDISP_N987 - XDISP_N1523 > 0.003$
- Energy ratio: $ENER_P3 / ENER_P5 > 1.5$
- Energy absorption: $ENER_P3 + ENER_P5 > 800.0$

Multiple Part Example

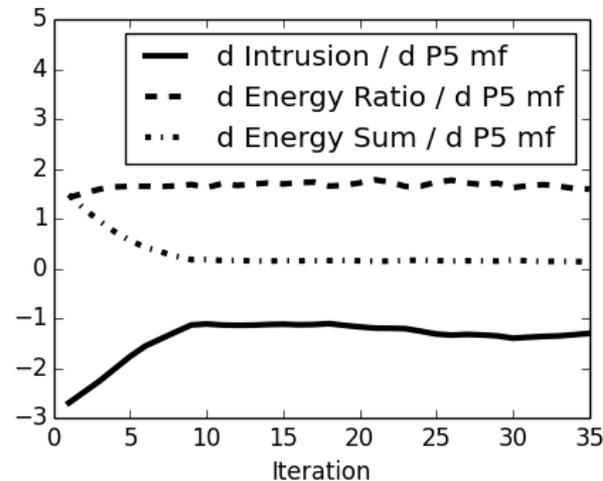
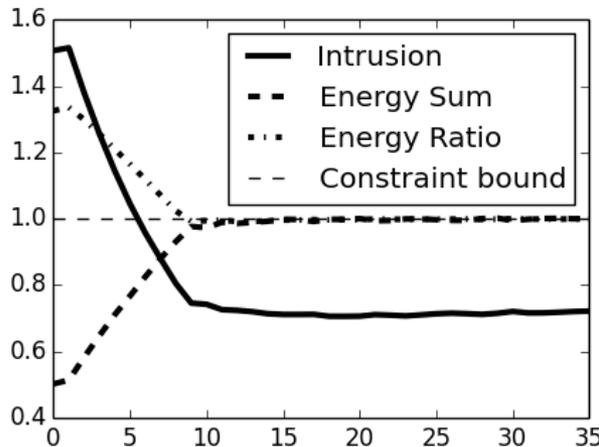
■ Design Histories

Global variables:
Part mass fractions



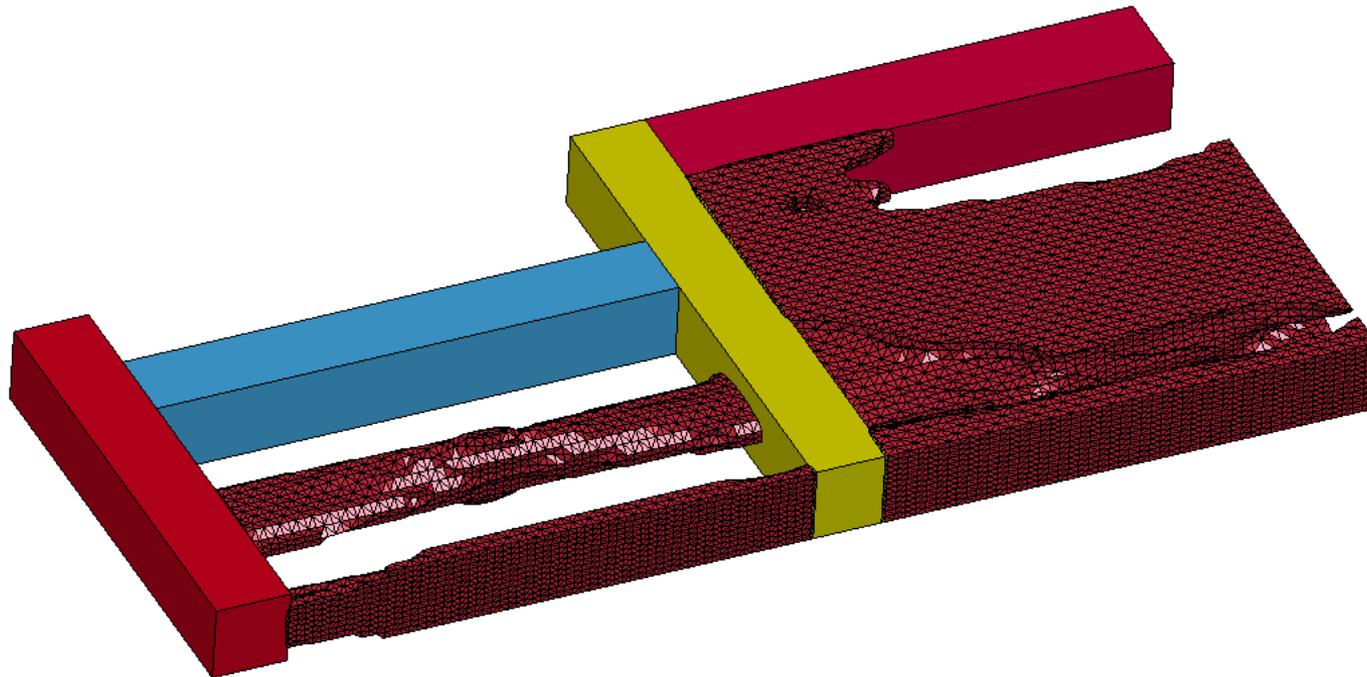
Constraints:
derivatives

Constraints



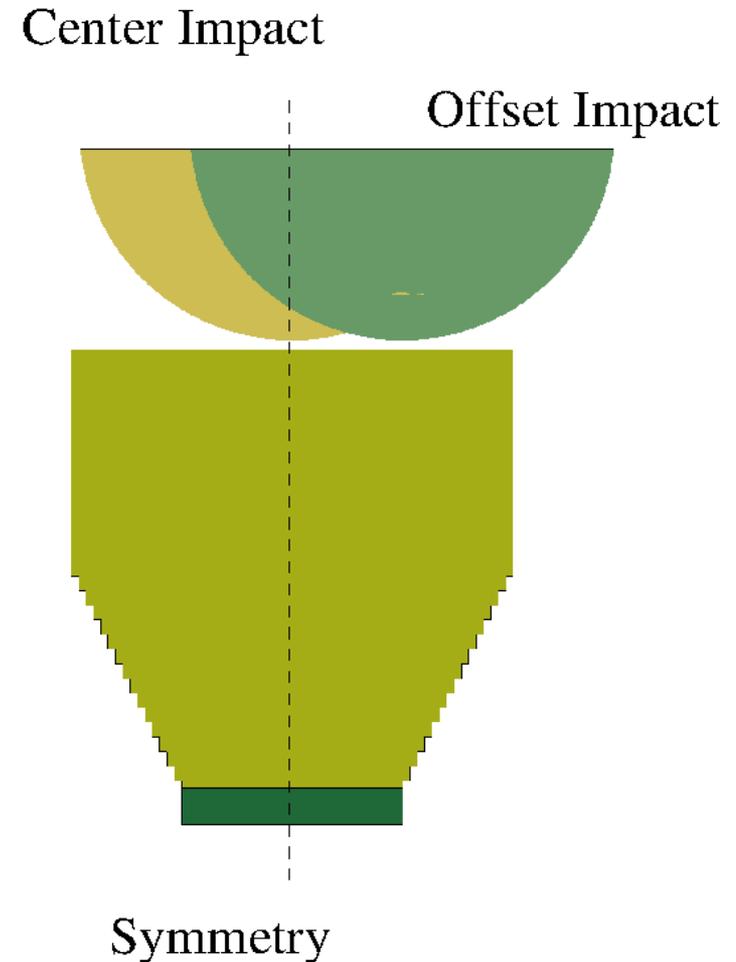
Multiple Part Example

- Final topology



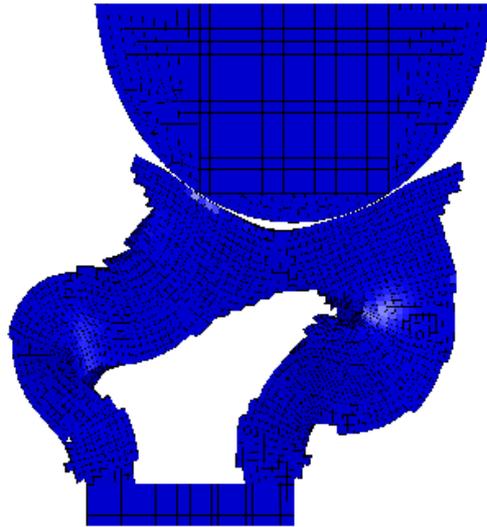
Highly nonlinear Example

- Load cases
- Objective
 - minimize difference in reaction force between load cases.
- Global variable
 - second load case weight
- The mass fraction is constant.

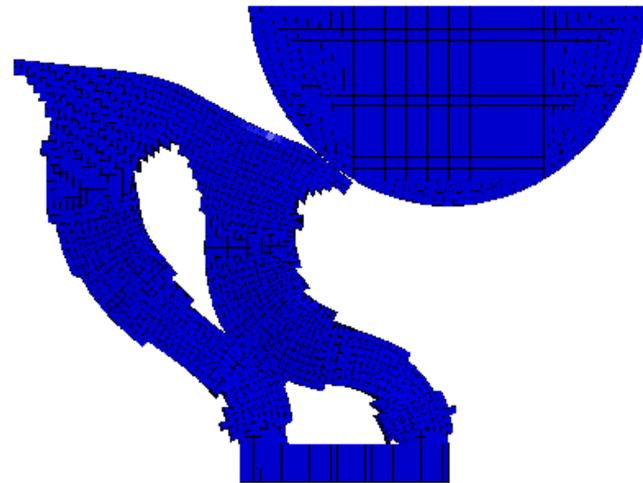


Highly nonlinear Example

- ***Buckling mode changes*** between iterations or within multipoint loop. Note the change of topology due to the ***contact closures***.



Iteration 20



Iteration 21

Offset impact load case

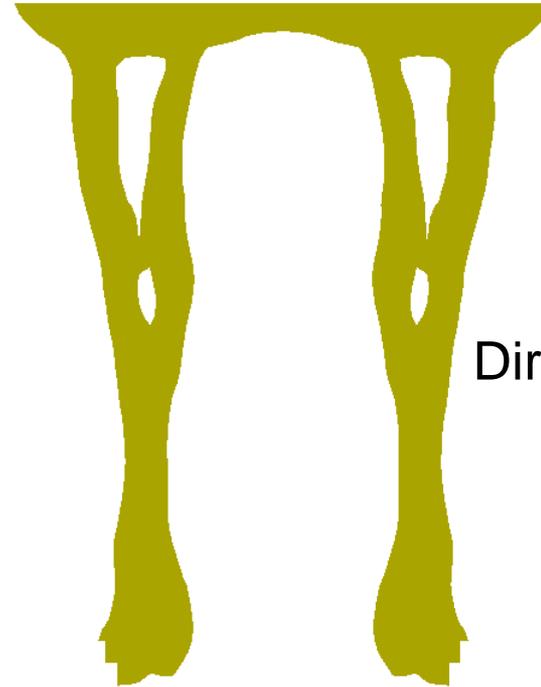
Highly nonlinear Example

- Final topologies
- Note the effect of the internal contact closure. The design algorithm must handle the load paths changes.

Response
Surface
Method



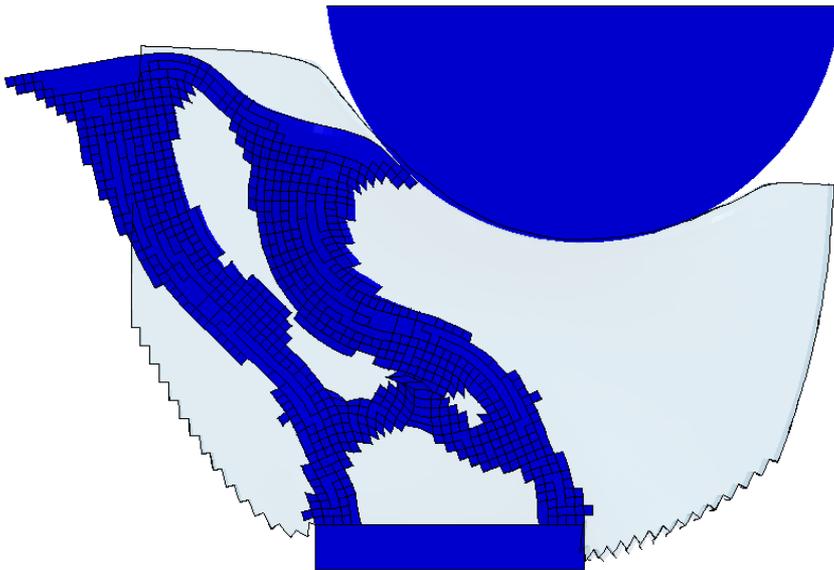
Direct Search



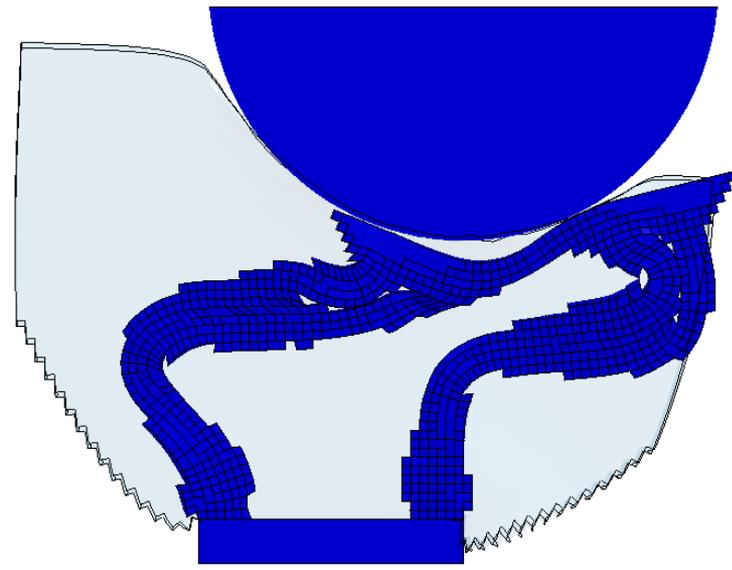
Highly nonlinear Example

- Deformed topologies

Response Surface Method



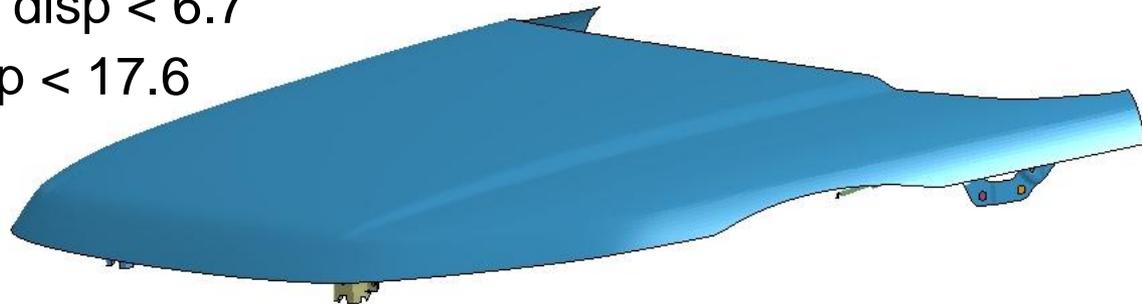
Direct Search



Bonnet Case Study

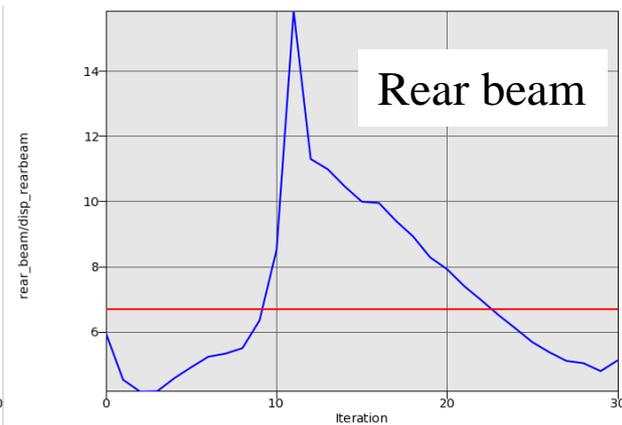
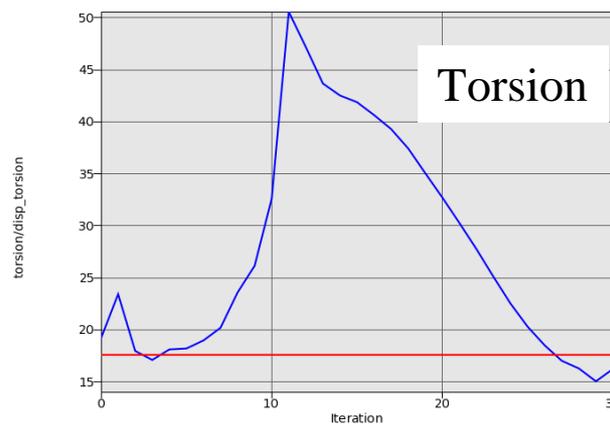
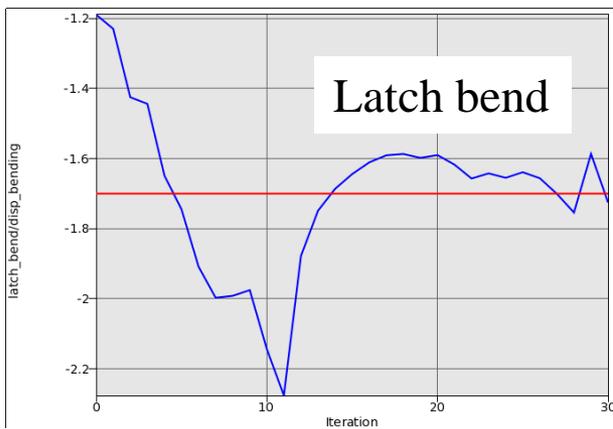
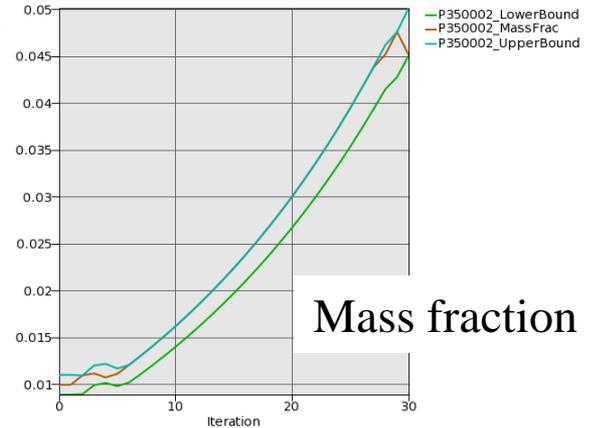
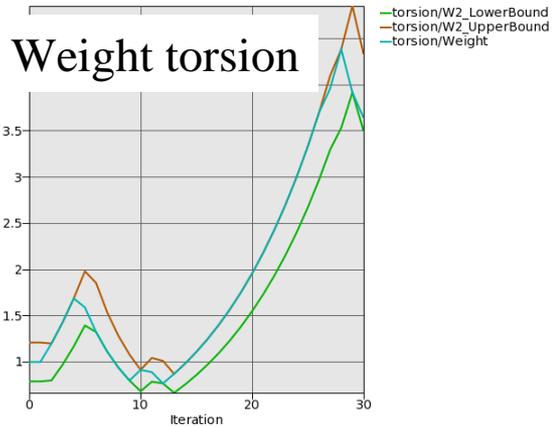
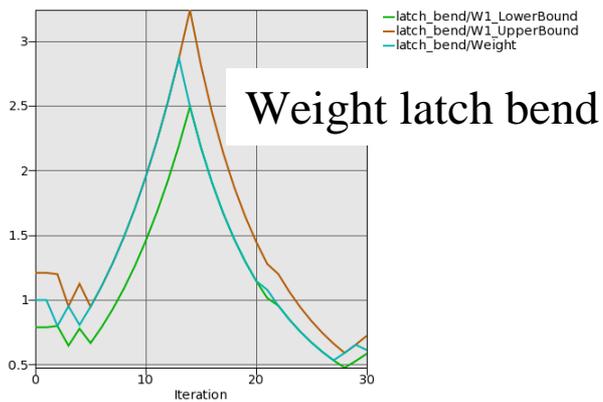
- Solid elements with layer of shell elements on top
 - Design part is solid part
- Three nonlinear load cases
 - Latch bend
 - Rear beam
 - Torsion
- Target mass fraction of 1%
- Constraints
 - Latch bend: $z \text{ disp} > -1.7$
 - Rear beam: resultant disp < 6.7
 - Torsion: resultant disp < 17.6

Example courtesy of JRL



Bonnet Case Study - Results

Global variables and constraints

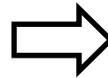


Bonnet Case Study - Results

- Final topology



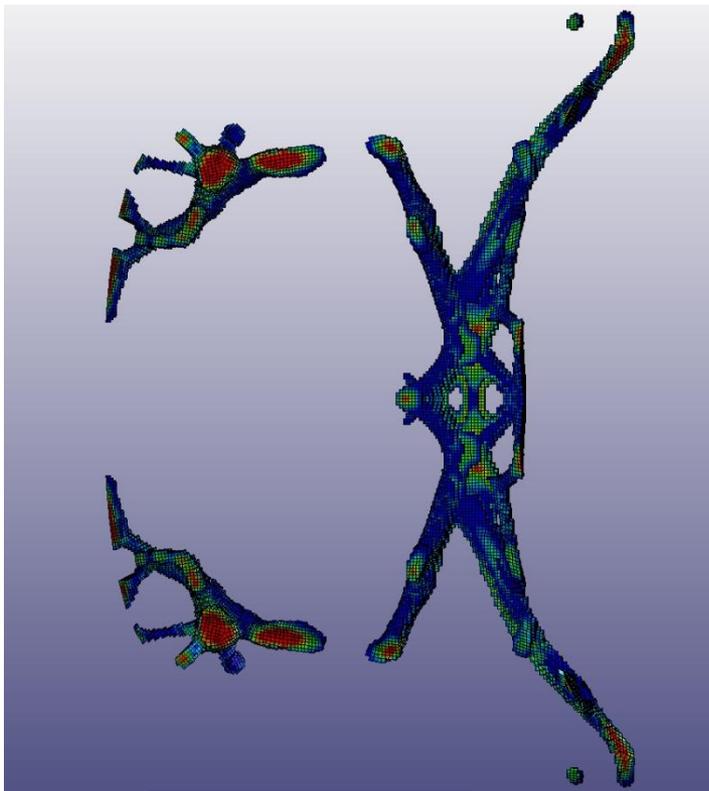
Initial design



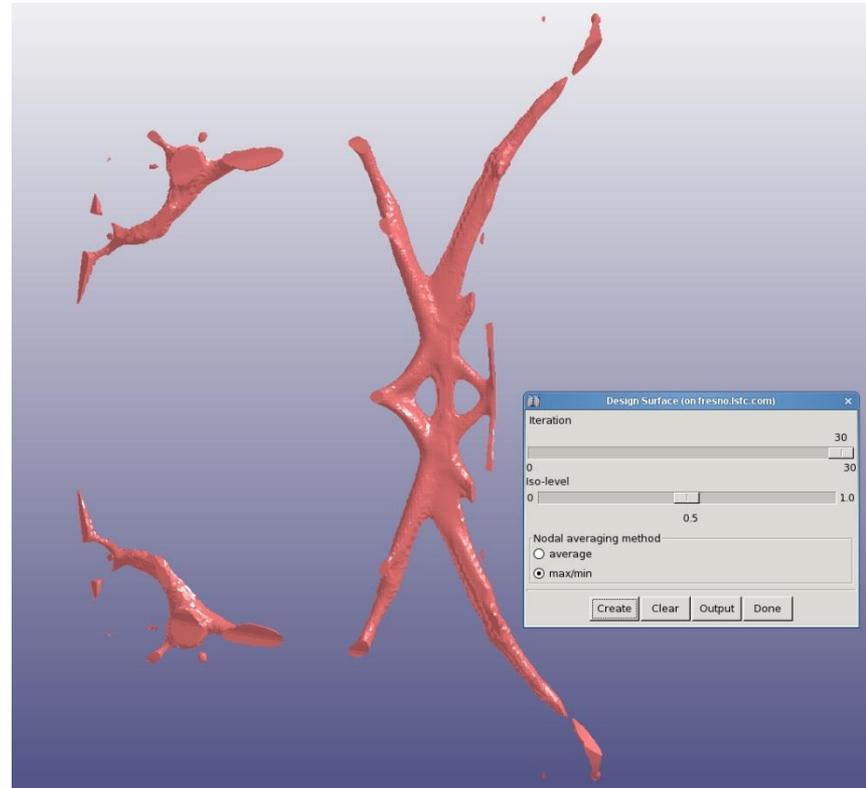
Optimal structure

Bonnet Case Study - Results

- Iso-surface



Final structure



Iso-surface

Other new Features

- Unconnected regions in a part can be identified and deleted
- The job submission system has been updated to match LS-OPT Version 5
- GUI improvements

Current Development

- Eigenfrequency Analysis
 - Topology optimization using Design Sensitivity Analysis in LS-DYNA
 - Objectives
 - Maximization of fundamental frequency

Thank you!