



# Technical Guide for Explicit Analyses using Ansys LS-DYNA



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# / Abstract

The purpose of this document is to provide a hands-on technical guide, by providing LS-DYNA models of typical applications, on how to make well-conditioned LS-DYNA models for explicit simulations. The idea is that the methods and settings described in this document can serve as a good starting point when building similar models for simulation using the explicit solver in Ansys LS-DYNA.

This document is under continuous development and future improved revisions are expected.

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# / Technical Guide

# Introduction

This document includes examples of typical applications for simulation using the explicit solver in Ansys LS-DYNA. The document is to be used in conjunction with running/studying the accompanied FE-models. The intention of the authors is to provide a hands-on technical guide, by providing LS-DYNA models of typical applications, on how to make well-conditioned LS-DYNA models for explicit simulations. The modeling is based on the experience from the authors and an effort has been made to use generally accepted best methods and settings. The idea is that the methods and settings described in this document can serve as a good starting point when building similar models for explicit simulation.

The models have been built and post processed using Ansys LS-PrePost. The examples are developed with MPP/ LS-DYNA in mind and tested with version R14.1 in single precision. To provide a relevant measurement of how computationally demanding each example is, the total elapsed time of each simulation/example is noted. For this purpose, all examples have been benchmarked on 4 cores on a Linux machine equipped with dual Xeon SP 6148 CPU`s (launch date Q3'17). Output files such as D3PLOT and BINOUT can be studied in any suitable LS-DYNA post-processor, such as for instance LS-PrePost.

# Support and training

For thorough details regarding LS-DYNA keywords and material models, the reader is referred to the LS-DYNA User's Manuals<sup>1</sup>. For help with using Ansys LS-DYNA and Ansys LS-PrePost contact your local Ansys LS-DYNA distributor. Useful web resources include <u>www.dynasupport.com</u> for general support and <u>www.dynalook.com</u> for conference papers. Example keyword files can be found at <u>www.dynaexamples.com</u>. Basically, the same information can be found at <u>Isdyna.ansys.com</u> under Resources and Knowledge base. If you find errors in this document, you are welcome to contact <u>support@dynamore.se</u>.

The document is under continuous development and future improved revisions are expected.

# System of units

The system of units used in the models is mm, ms, kg, kN, GPa

<sup>&</sup>lt;sup>1</sup> ANSYS, LS-DYNA® keyword user's manual Volume 1, 2 & 3. See also https://lsdyna.ansys.com/manuals.

# **Control cards**

Most of the control parameters are left to their default settings. The main approach from the authors have been to rather define parameters further down in the model than on a global level on \*CONTROL\_OPTION, if possible. The control cards, and database cards, can be found in the include file control\_database.k.

The control settings used in this technical guide are shown in the cards below. The parameters that differ from defaults are marked with a blue rectangle. Dashed rectangles indicate additional parameters that are discussed. The parameter settings may differ slightly in some examples to comply with the specific problem at hand.

#### \*CONTROL\_ACCURACY

1	<u>OSU</u>		<u>INN</u>		PIDOSU •	IACC	EXACC
	1	-	4	~	0	0	0.0
i							

Comment: Objective stress update and invariant node numbering is switched on. PIDOSU can be used to limit objective stress update to a part set, read the manual for details.

#### \*CONTROL\_BULK\_VISCOSITY

1 <u>01</u>	<u>Q2</u>	TYPE		BTYPE		TSTYPE	
1.5000000	0.0600000	-2	•	0	•	0	•

Comment: Bulk viscosity is activated for some shell element formulations in addition to the standard bulk viscosity for solids.



#### \*CONTROL\_CONTACT

Comment: Shell thickness changes are considered in single surface contact, see THKCHG and ISTUPD in \*CONTROL\_SHELL. Actual shell thickness is used in single surface contact, see SSTHK. IGNORE, SHLEDG and PSTIFF can be activated on a global level here or locally on \*CONTACT (recommended). SPOTDEL is activated so that the spot weld is deleted when an attached shell fails. SPOTSTP=2, to report deleted spot welds due to SURFB in the contact definition not found.



#### \*CONTROL\_DYNAMIC\_RELAXATION

1	NRCYCK	DRTOL	DRFCTR	DRTERM	TSSFDR	IRELAL		<u>EDTTL</u>	<u>IDRFLG</u>
	250	1.0E-03	9.95E-01		0.0	0	•	4.0E-02	0

Comment: DR is activated by SIDR≠0 on any \*DEFINE\_CURVE in the model. DR can be fully disabled in the simulation by IDRFLG=-999, see manual for details.

#### \*CONTROL\_ENERGY



Comment: Energy dissipation calculation is switched on for most parameters.



#### \*CONTROL\_OUTPUT

Comment: NPOPT and NEECHO limit the amount of data written to d3hsp and standard out. Update of the reference node coordinate is turned on for beams for visualization purposes, see NREFUP. This requires a unique third node to be defined for each beam element. Output of eroded internal and kinetic energy into the matsum file, see IERODE. Note that TETIOS8=1 can be used to output full nodal connectivity for higher order elements. ICRFILE=2 to visualize (in e.g. LS-PrePost) cross sections defined by \*DATABASE\_CROSS\_SECTION. SPC2BND to convert constraints defined on \*MAT\_RIGID and \*CONSTRAINED\_NODAL\_RIGID\_BODY\_SPC to equivalent \*BOUNDARY\_PRESCRIBED\_MOTION\_RIGID. Needed for force extraction from bndout. PENOUT≠0 to output information about contact penetrations for MORTAR contacts to d3plot and sleout. HISNOUT=1 activates output of more information for history variables in d3hsp. ENGOUT≠0 to output information about minimum sliding energy density for MORTAR contacts to d3plot.

#### \*CONTROL\_SHELL

1	WRPANG		ESORT		<u>IRNXX</u>		<u>ISTUPD</u>		<u>THEORY</u>		<u>BWC</u>	MITER	PROJ	
	30.000000		1	-	-1		4	~	1		2 👻	1 -	- 1	~
		_					·	A	ctive opti	onal	cards		_	1
							_		) None (	Ор	t1 () Opt12	🔾 Opt123 🧿	) Opt1234	
2	ROTASCL		INTGRD		LAMSHT		CSTYP6		THSHEL					
	1.0000000		0		0		1		0					
3	PSSTUPD		SIDT4TU	•	<u>CNTCO</u>		<u>ITSFLG</u>		IRQUAD		W-MODE	STRETCH	<u>ICRQ</u>	
	0		0		1	~	0	Ŧ	3	~	0.0	0.0	0	
4	NFAIL1		NFAIL4		PSNFAIL	•	KEEPCS		DELFR		DRCPSID	DRCPRM	INTPER	R
	0		1		0		0	~	3	-	0	1.0000000	0	~
5	DRCMTH		LISPSID	•	NLOCDT									
	0	-	0		0	-								

Comment: Automatic sorting of degenerated quadrilateral shell elements, see ESORT. Shell thickness change due to membrane straining is activated, see ISTUPD. THEORY=1 is recommended when having non-uniform-thickness shells in contacts. LAMSHT=5 is recommended when working with layered composites, see \*PART\_COMPOSITE and \*INTEGRATION\_SHELL. Shell thickness offset, for instance by NLOC, affects the contact reference plane, see CNTCO. NFAIL4=1 to delete highly distorted fully integrated elements and print a message. DRCPSID can be used to activate drilling stiffness of a shell part set. DELFR to delete isolated or badly connected shell elements.

#### \*CONTROL\_SOLID

1	SORT	<u>FMATRIX</u>	NIPTETS	<u>SWLOCL</u>	PSFAIL •	<u>Т10ЛОГ</u>	<u>ICOH</u>	<u>TET13K</u>		
	1 -	0	4	1 -	0	0.0	1	0 -		
			-		Acti	ve optional ca None 🔵 Opt1	opt12	1		
2	<u>M1</u>	<u>PM2</u>	<u>PM3</u>	<u>PM4</u>	<u>PM5</u>	<u>PM6</u>	<u>PM7</u>	<u>PM8</u>	<u>PM9</u>	<u>PM10</u>
	0	0	0	0	0	0	0	0	0	0
3	ET13V	RINRT	COHEQC							
	1 -	0 -	0 -							

Comment: Automatic sorting of degenerated tetrahedron and pentahedral solid elements is activated with ESORT. Note that optional card 2 is not supported in LS-DYNA versions before R11.2.2. Sometimes FMATRX=2 will improve accuracy for prestressed rubber solids using reference geometry or when explicit/implicit switching is performed. When using an ERODING contact PSFAIL will limit the check for negative volume to the solid elements in the part set. ICOH=1 to delete badly connected cohesive elements. The more accurate type 13 solid implementation with TET13V.

#### \*CONTROL\_SOLUTION

1 SOLN	NLQ	<u>ISN</u> /	IAN	LCINT	LCACC	NCDCF	NOCOP	
0	- 0	1	Ŧ	1001	0	1	0	-

Comment: ISNAN=1 for check of NaN in force and moment arrays (debugging purposes). LCINT is set to 1001, default is 100. An odd number is useful when having curves that are both in the negative and the positive region of the x-axis. NOCOPY=1 may speed-up element processing for certain materials.



### Additional control settings

Additional control cards are \*CONTROL\_MPP\_IO\_LSTC\_REDUCED, \*CONTROL\_MPP\_IO\_NODUMP and CONTROL\_MPP\_DECOMPOSITION\_AUTOMATIC. There are no input parameters for these cards.

The decomposition of a model can be studied in Ansys LS-PrePost by adding the keyword \*CONTROL\_MPP\_DECOMPOSITION\_OUTDECOMP. In the examples presented in this guide \*CONTROL\_MPP\_DECOMPOSITION\_AUTOMATIC has been used, which often makes a more efficient solution compared to default. When running simulations on many cores other settings might be more suitable through, e.g., \*CONTROL\_MPP\_DECOMPOSITION\_TRANSFORMATION or \*CONTROL\_MPP\_DECOMPOSITION\_ARRANGE\_PARTS.

\*CONTROL\_MPP\_IO\_NODUMP suppresses the output of all possible restart files. Remove this card if restart files are needed.

\*CONTROL\_HOURGLASS is omitted. In this guideline the hourglass control is referenced on \*PART and defined by \*HOURGLASS.

Note that \*CONTROL\_TERMINATION and \*CONTROL\_TIMESTEP can be found in the main file *run.key*.

#### \*CONTROL\_TERMINATION

1	<u>ENDTIM</u>	ENDCYC	DTMIN	ENDENG	ENDMAS •	NOSOL	
	10.0000000	0	0.0	0.0	1.000e+08	0	-

Comment: Termination time is set by ENDTIM.

#### \*CONTROL\_TIMESTEP



Comment: For most cases the default value of the time step scale factor TSSFAC is 0.9, but there are exceptions, see manual. Further reduction of TSSFAC can be needed to assure numerical stability. Mass scaling setting in DT2MS. IHDO=1 is used for a more consistent time step calculation for solid elements. A negative number on IMSCL references a part set for selective mass scaling, see manual for details.



## Database cards

Output files containing results information are requested and specified in the input file using \*DATABASE. There are many options available, and the reader is advised to the \*DATABASE section in the manual. In the comments below only a brief and limited explanation is given for the parameters. Please read the manual for the full description. The following, baseline, database settings are used in this guideline. The parameters that differ from defaults are marked with a blue rectangle. Dashed rectangles indicate additional parameters that are discussed.

#### \*DATABASE\_BINARY\_D3PLOT

1	<u>DT</u>	LCDT -	BEAM	NPLTC	PSETID -	
	0.5000000	0	0	0	0	
2	<u>IOOPT</u>	RATE	CUTOFF	WINDOW	TYPE	PSET -
	0 🔻	0.0	0.0	0.0	0 🗸	0

Comment: Time interval DT between d3plot output states. There are many binary output databases that can be requested and controlled using \*DATABASE\_BINARY\_OPTION1\_{OPTION2}. E.g. \*DATABASE\_BINARY\_D3DRLF, \*DATABASE\_BINARY\_D3PART and DATABASE\_BINARY\_ INTFOR are commonly used.

1	<u>NEIPH</u>		<u>NEIPS</u>		<u>MAXINT</u>	STRFLG		<u>SIGFLG</u>		EPSFLG		<u>RLTFLG</u>		ENGFLG	
	0		0		3	1		1	~	1	~	1	~	1	*
2	CMPFLG		<u>IEVERP</u>		<u>BEAMIP</u>	DCOMP		<u>SHGE</u>		<u>STSSZ</u>		N3THDT		IALEMAT	
	0	-	1	~	0	1	-	2	~	1	~	2	~	1	*
3	NINTSLD		PKP SEN		<u>SCLP</u>	<u>HYDRO</u>		MSSCL		THERM		INTOUT		NODOUT	
	0	•	0	~	1.0000000	0		1	*	0	*		~		*
4	DTDT		RESPLT		<u>NEIPB</u>	QUADSLD		CUBSLD		DELERES					
	0	-	0	-	0	0	Ŧ	0	Ŧ	0	Ŧ				

#### \*DATABASE\_EXTENT\_BINARY

Comment: Some options to control the content of the binary output databases (typically d3plot, d3thdt and d3part). NEIPH, NEIPS and NEIPB, number of additional history variables to output for solids, shells, tshells and beams. See also \*DEFINE\_MATERIAL\_HISTORIES for additional options to control the output of history variables. MAXINT, number of shell (and tshell) through-thickness integration points to output. NINTSLD, number of solid element integration points to output. BEAMIP, number of integration points to output for beams. STRFLG=1, write strain tensor data to d3plot, elout and dynain. IEVERP=1 to output each d3plot state to a separate plot file. SHGE=1 for including shell hourglass energy density. HYDRO to output extra data (history variables) for solids, e.g. pressure from bulk viscosity, hourglass energy per unit initial valume and internal energy per reference volume. MSSCL=1 to output incremental (added) nodal mass due to mass scaling. The content can be further specified for a part set using \*DATABASE\_BINARY\_D3PART and \*DATABASE\_EXTENT\_D3PART for output in d3part. The content of the contact interface database intfor is specified using \*DATABASE\_BINARY\_INTFOR and \*DATABASE\_EXTENT\_INTFOR.

#### \*DATABASE\_FORMAT

I IFORM	IBINARY
0 🗸	1 •

Comment: IBINARY=1, 32bit IEEE format for binary output files (e.g. d3plot) to reduce output volume for simulations made with double precision LS-DYNA by a factor of two.

#### \*DATABASE\_OPTION1\_{OPTION2}

ABASE_BN	DOUT						
dt	binary	lcur	ioopt	option1	option2	option3	option4
0.01	Õ	0	- 1	- 0	- 0	- 0	- 0
ABASE_EL	OUT						
dt	binary	lcur	ioopt	option1	option2	option3	option4
0.01	Õ	0	- 1	- 0	- 0	- 0	- 0
'ABASE_GL	STAT						
dĒ	binary	lcur	ioopt				
0.01	Õ	0	- 1				
'ABASE_MA	TSUM						
dĒ	binary	lcur	ioopt				
0.01	0	0	1				
'ABASE_NO	DOUT						
dt	binary	lcur	ioopt	option1	option2		
0.01	0	0	1	0.0	0		
'ABASE_RC	FORC						
dt	binary	lcur	ioopt				
0.01	0	0	1				
'ABASE_RW	FORC						
dt	binary	lcur	ioopt				
0.01	0	0	1				
'ABASE_SE	CFORC						
dt	binary	lcur	ioopt				
0.01	0	0	1				
'ABASE_SL	EOUT						
dt	binary	lcur	ioopt				
0.01	0	0	1				
	ABASE_BN dt 0.01 ABASE_EL 0.01 ABASE_GL 0.01 ABASE_MA dt 0.01 ABASE_NO 0.01 ABASE_RC 0.01 ABASE_RC 0.01 ABASE_SE dt 0.01 ABASE_SL 0.01	ABASE_BNDOUT dt binary 0.01 0 ABASE_ELOUT dt binary 0.01 0 ABASE_GLSTAT dt binary 0.01 0 ABASE_MATSUM dt binary 0.01 0 ABASE_NODOUT dt binary 0.01 0 ABASE_RCFORC dt binary 0.01 0 ABASE_RWFORC dt binary 0.01 0 ABASE_SECFORC dt binary 0.01 0 ABASE_SECFORC dt binary 0.01 0 ABASE_SECFORC dt binary 0.01 0 ABASE_SECFORC dt binary 0.01 0 ABASE_SECFORC dt binary 0.01 0 ABASE_SECFORC dt binary 0.01 0 ABASE_SLEOUT dt binary 0.01 0	ABASE_BNDOUTlcurdtbinarylcur0.0100ABASE_ELOUT00dtbinarylcur0.0100ABASE_GLSTAT00dtbinarylcur0.0100ABASE_MATSUM00dtbinarylcur0.0100ABASE_NODOUT00ABASE_ROFORC00dtbinarylcur0.0100ABASE_REFORC00dtbinarylcur0.0100ABASE_SECFORC00dtbinarylcur0.0100ABASE_SLEOUT00dtbinarylcur0.0100	ABASE_BNDOUTlcuriooptdtbinarylcurioopt0.01001ABASE_ELOUT001ABASE_GLSTAT001ABASE_GLSTAT001ABASE_MATSUM001ABASE_MATSUM001ABASE_NODOUT001ABASE_NODOUT001ABASE_ROFORC01ABASE_RCFORC01ABASE_SECFORC01ABASE_SECFORC01ABASE_SECFORC01ABASE_SECFORC01ABASE_SLEOUT01dtbinarylcurioopt0.01001ABASE_SLEOUT01dtbinarylcurioopt0.01001	ABASE_BNDOUT dt binary lcur ioopt option1 0.01 0 0 1 0 ABASE_ELOUT dt binary lcur ioopt option1 0.01 0 0 1 0 ABASE_GLSTAT dt binary lcur ioopt 0.01 0 0 1 ABASE_MATSUM dt binary lcur ioopt 0.01 0 0 1 ABASE_NODOUT dt binary lcur ioopt option1 0.01 0 0 1 0.0 ABASE_RCFORC dt binary lcur ioopt 0.01 0 0 1 ABASE_SECFORC dt binary lcur ioopt 0.01 0 0 1 ABASE_SLEOUT dt binary lcur ioopt 0.01 0 0 1	ABASE_BNDOUT       dt       binary       lcur       icopt       option1       option2         0.01       0       0       1       0       0         ABASE_ELOUT       dt       binary       lcur       icopt       option1       option2         0.01       0       0       1       0       0       0         ABASE_ELOUT       0       0       1       0       0       0         0.01       0       0       1       0       0       0         ABASE_GLSTAT       0       0       1       0       0         ABASE_MATSUM       0       0       1       0       0         ABASE_MATSUM       0       0       1       0       0         ABASE_MATSUM       0       0       1       0       0       0         ABASE_NODOUT       0       0       1       0       0       0       0         ABASE_RCFORC       0       0       1       0       0       1       0       0       1         ABASE_SCFORC       0       0       0       1       0       0       1         ABASE_SLEOUT       0	ABASE_BNDOUT       dt       binary       lcur       ioopt       option1       option2       option3         0.01       0       0       1       0       0       0         ABASE_ELOUT       dt       binary       lcur       ioopt       option1       option2       option3         0.01       0       0       1       0       0       0         ABASE_GLSTAT       oopt       oopt       0       0       0         0.01       0       0       1       0       0       0         ABASE_GLSTAT

Comment: Request and control output of different LS-DYNA ASCII databases using \*DATBASE\_OPTION1\_{OPTION2}. Time interval between outputs DT. BINARY=0, data written to the binary database binout is default for MPP. BNDOUT, boundary condition forces and energy. ELOUT, element data (see \*DATABASE\_HISTORY\_OPTION). GLSTAT, global statistics and energy. MATSUM, part energies. NODOUT, nodal motion (see \*DATABASE\_HISTORY\_NODE\_OPTION). RCFORC, resultant contact interface forces. RWFORC, rigidwall forces. SECFORC, cross section forces (see \*DATABASE\_CROSS\_SECTION\_OPTION), SLEOUT, contact interface energy.

#### Record maximum, and/or minimum, stress/strain

\*DATABASE\_BINARY\_D3MAX activates the output of binary plot database d3max. Elements to be included in d3max are chosen with \*DATABASE\_MAX\_OPTION. The parameter "OPTION" determines if maximum or minimum values are to be recorded.

\*DEFINE\_MATERIAL\_HISTORIES offer extensive possibilities to control output to history variables such as to record max/min values over time, stress ranges, points in time and to apply operators.



# **Contact cards**

The main contact types that are used in the examples are \*CONTACT\_AUTOMATIC\_SINGLE\_SURFACE with SOFT=1 or SOFT=2 for shells and solids and \*CONTACT\_GENERAL for beams.

\*CONTACT\_ERODING\_SINGLE\_SURFACE is used for the cases where erosion of solids is expected. For tricky contact situations, often with high contact pressures, a MORTAR contact variant can be useful.

The contact settings are mostly left to their defaults. Common for these contacts is that friction coefficients are set with FS/FD/DC and contact damping is added with VDC. The soft constraint option is chosen with SOFT. Using option "\_ID" is recommended for many reasons.

The contact settings are shown in the cards that follow. Parameters that differ from defaults are marked with a blue rectangle. Dashed rectangles indicate additional parameters that are discussed.



Comment: SOFT=1. IGNORE=2 so that initial penetrations are tracked, which means that nodes are not moved. Information about initial penetrations at t=0.0 can be found in message files. Set FTORQ=2 to account for moments across the contact interface. Typically, it is recommended to set DTSTIF to the initial timestep size for consistency or to a specific timestep size needed for a validated component.

*CONTACT AUTOMATIC SINGLE SURFACE ID								
\$#	cīd	-	-	-				title
	1Ge:	neral con	tact					
\$#	surfa	surfb	surfatyp	surfbtyp	saboxid	sbboxid	sapr	sbpr
	1	0	2	Ō	0	0	0	0
\$#	fs	fd	dc	vc	vdc	penchk	bt	dt
	0.2	0.2	0.0	0.0	20.0	0	0.01.	00000E20
\$#	sfsa	sfsb	sast	sbst	sfsat	sfsbt	fsf	vsf
	1.0	1.0	0.0	0.0	1.0	1.0	1.0	1.0
\$#	soft	sofscl	lcidab	maxpar	sbopt	depth	bsort	frcfrq
	2	0.1	0	1.025	3	35	0	1
\$#	penmax	thkopt	shlthk	snlog	isym	i2d3d	sldthk	sldstf
	0.0	0	0		0_	0	0.0	0.0
\$#	igap	ignore	dprfac	dtstif	edgek	unused	flangl	cid_rcf
•	1	2	0.0				0.0	0
Ş#	q2tri	dtpchk	sfnbr	fnlscl	dnlscl	tcso	tiedid	shledg
•	. 0	0.0	0.0	0.0	0.0	. 0	<u> </u>	1
Ş#	sharec	cparm8	ipback	srnde	fricsf	icor	ftorg	region
•			U	U	1.0	U	╘╼╤╝┛	U
\$#	pstiff	ignroff	-	fstol	2dbinr	ssftyp	swtpr	tetrac
	1	U		2.0	U	U	U	1.0

Comment: SOFT=2 with SBOPT=3 and DEPTH=35. SHLEDG=1 for square shell edges flush with the nodes. IGNORE=2. TETFAC can be used to make the contact stiffness for tetrahedral solid elements more comparable with hex meshes, see manual. Set FTORQ=2 to account for moments across the contact interface. Typically, it is recommended to set DTSTIF to the initial timestep size for consistency. EDGEK can be used to increase edge-to-edge penalty stiffness.

*CONTACT ERODING SINGLE SURFACE ID								
\$#	cid							title
	1Ge:	neral con	tact					
\$#	surfa	surfb	surfatyp	surfbtyp	saboxid	sbboxid	sapr	sbpr
	1	0	2	0	0	0	0	0
\$#	fs	fd	dc	vc	vdc	penchk	bt	dt
	0.2	0.2	0.0	0.0	20.0	0	0.01.	00000E20
\$#	sfsa	sfsb	sast	sbst	sfsat	sfsbt	fsf	vsf
	1.0	1.0	0.0	0.0	1.0	1.0	1.0	1.0
\$#	isym	erosop	iadj					
•	U	1	1					
Ş#	soft	sofscl	lcidab	maxpar	sbopt	depth	bsort	trctrq
<b>.</b>	2	0.1	U	1.025	3	35	U	1
Ş#	penmax	thkopt	shithk	snlog	ısym	12d3d	sidthk	sidstf
<b>.</b>	. U. U	2	, , 1		╺╾┯╼┥┹┓	U	0.0	. 0.0
\$#	ıgap	ignore	dprfac	dtstif	edgek	unused	flangl	cid_rcf
<b>.</b>	. <u>1</u>		0.0	آب س ⊻. ۲. ۳			0.0	
\$#	qztri	atpenk	stnbr	TRISCI	anisci	tcso	tiedid	shiedg
<u>а</u>	-1	0.0	U.U	0.0	0.0	. U	U .	<u>1</u>
\$#	snarec	cparms	прраск	srnae	rricsr	1cor	rtorq	region
<u>н</u> о	U	:	U		1.0	<del>.</del>		U
\$#	pstiff 1	ignrorf	-	ISTOL	2001nf	ssrcyp	swtpr	tetrac
		U		2.0	U	U	U	1.0

Comment: The eroding contact has the same settings as the ordinary SOFT=2 contact presented above except that one extra card has been added specific for the ERODING functionality. Both EROSOP and IADJ are hard coded 1 for MPP.

*C0 \$#	*CONTACT_AUTOMATIC_SINGLE_SURFACE_MORTAR_ID \$# cid title							
\$#	surfa 1	surfb	surfatyp	surfbtyp O	saboxid O	sbboxid O	sapr O	sbpr O
\$#	fs 0.2	fd 0.2	dc 0.0	vc 0.0	vdc 20.0	penchk O	bt 0.01	dt .00000E20
\$#	sfsa	sfsb	sast	sbst	sfsat	sfsbt	fsf	vsf
•	1.0	1.0	0.0	U. U	1.0	1.0	1.0	1.0
Ş#	soft	sofscl	lcidab	maxpar	sbopt	depth	bsort	trctrq
	0	0.1	0	1.025	2	2	0	1
\$#	penmax	thkopt	shlthk	snlog	isym	i2d3d	sldthk	sldstf
	0.0	O	0	Ő	0	0	0.0	0.0
\$ <b>#</b>	igap	ionore	mpar1	mpar2	edgek	unused	flangl	cid rcf
•	1	2	0.0	0.0	0.0		0.0	- 0

Comment: The contact settings for the MORTAR-contact are mostly left to defaults. The contact stiffness can be scaled by SFSA/IGAP if penetrations are unacceptable. In case of solids or tshells setting the release depth with PENMAX may also help. PENMAX should then be set to a typical element thickness found in the model. For MORTAR, IGNORE=2 by default, although it behaves a bit differently compared to other contacts, i.e.no tracking of initial penetrations. If a tracked surface is desired, then IGNORE=1 can be activated. If IGNORE is set to a negative value, then contact within a part is omitted. See manual for details.



Comment: \*CONTACT\_AUTOMATIC\_GENERAL is only used for beam-to-beam contact in this technical guide. Note: There is an option to exclude contact with beams in the same part (PID) using CPARM8 (1 or 2) in the additional \_MPP card. Set CPARM8=2 to consider also spotweld beams (type 9) in the contact. Typically, it is recommended to set DTSTIF to the initial timestep size for consistency or to a specific timestep size needed for a validated component. SOFT=1 and IGNORE=2.

### **Tied contacts**

There are many options in LS-DYNA for the modelling of tied contacts. See the LS-DYNA manual for the complete functionality that is available as well as for more details. In this section some commonly used contact variants are presented. The intention is not to cover all functionality but to highlight some tied contacts that are widely used and may serve as a good starting point when modelling tied connections. For best control it is recommended to use node and segment sets rather than part or part sets. Thereby preventing that, unintentionally, wrong nodes are being tied. It may sometimes be advantageous to use CONSTRAINED\_EXTRA\_NODE instead for tying deformable nodes to rigid bodies since the nodes can be an arbitrary distance away from the rigid body.

#### Constraint or penalty

There are two main categories of tied contacts, i.e., constraint based, and penalty based. The offset options on OPTION4 on \*CONTACT determine the category. If OPTION4 is left blank the contact will be constraint based. If "OFFSET" or "BEAM\_OFFSET" is used, then the contact will be penalty based. "CONSTRAINT\_OFFSET" is constraint based. Penalty based variants will work with rigid bodies but not constraint based. That is a fundamental difference. Another functionality that is affected by OPTION4 is the treatment of SURFA nodes. I.e., if they are moved to SURFB at initialization or not (i.e., allowing offsets). If OPTION4 is left blank, then SURFA nodes will be projected and moved to SURFB at initialization. For the other OPTION4 variants SURFA nodes will not be moved thus allowing offsets/play between SURFA and SURFB to remain. An example is given below.

Contact type	Allow	Allow
	rigid	offset/
	bodies	play
TIED_SHELL_EDGE_TO_SURFACE	No	No
TIED_SHELL_EDGE_TO_SURFACE_OFFSET	Yes	Yes
TIED_SHELL_EDGE_TO_SURFACE_BEAM_OFFSET	Yes	Yes
TIED_SHELL_EDGE_TO_SURFACE_CONSTRAINED_OFFSET	No	Yes

By IPBACK≠0 on Optional Card E a "backup" penalty contact will created for rigid nodes that fail to tie. This feature is available for some constraint based tied contacts, see below.

TIED\_NODES\_TO\_SURFACE TIED\_NODES\_TO\_SURFACE\_CONSTRAINED\_OFFSET TIED\_SURFACE\_TO\_SURFACE TIED\_SURFACE\_TO\_SURFACE\_CONSTRAINED\_OFFSET TIED\_SHELL\_EDGE\_TO\_SURFACE TIED\_SHELL\_EDGE\_TO\_SURFACE\_CONSTRAINED\_OFFSET



#### Tying tolerance

If the default tying tolerance is not appropriate, then it can be set by the user. I.e., with negative numbers on SAST and SBST. For numerical robustness aim to keep the tying distance low.

E.g., SAST = SBST = -1.0 will give a tolerance of 0.6\*(1.0+1.0) = 1.2 mm.

#### **Highlighted contacts**

The contact cards below are presented with OPTION4 (offset options) left blank but could equally been presented with other choices of OPTION4. The choice of contact typically depends on the type of elements being connected, which is primarily due to if the nodes have rotational DOF or not. The intended use of the contacts below is revealed in their titles. SAST/SBST may be used to set tying tolerance, see dashed rectangles.

*CON	*CONTACT TIED NODES TO SURFACE ID							
\$#	cīd							title
	1Tie	d solids <sup>,</sup>	-to-solids	and solids	-to-shells	;		
\$#	surfa	surfb	surfatyp	surfbtyp	saboxid	sbboxid	sapr	sbpr
	1	2	3	3	0	0	0	0
\$#	fs	fd	dc	vc	vdc	penchk	bt	dt
	0.0	0.0	0.0	0.0	0.0	· 0	0.01.	00000E20
\$#	sfsa	sfsb	sast	sbst	sfsat	sfsbt	fsf	vsf
	1.0	1.0	0.0	0.0	1.0	1.0	1.0	1.0

Comment: OPTION4 can be "blank", "OFFSET" or "CONSTRAINED\_OFFSET".

<pre>*CONTACT_TIED_SHELL_EDGE_TO_SURFACE_ID \$# cid title title</pre>								title
	ITTE	a snells	-to-shells	and beams-	to-shells			
Ş#	surfa	surfb	surfatyp	surfbtyp	saboxid	sbboxid	sapr	sbpr
	1	2	3	3	0	0	0	0
\$#	fs	fd	dc	vc	vdc	penchk	bt	dt
	0.0	0.0	0.0	0.0	0.0	- 0	0.01.0	0000E20
\$#	sfsa	sfsb	sast	sbst	sfsat	sfsbt	fsf	vsf
	1.0	1.0	0.0	0.0	1.0	1.0	1.0	1.0

Comment: OPTION4 can be "blank", "OFFSET", "BEAM\_OFFSET" or "CONSTRAINED\_OFFSET". "BEAM\_OFFSET" and "CONSTRAINED\_OFFSET" enforces equilibrium since transmitted forces and moments are coupled. The drilling rotation constraint method that is default in implicit can be activated for a part set also in explicit, see DRCPSID on \*CONTROL\_SHELL. Thereby tying also torsion for beams tied to shells.

*CON	*CONTACT TIED SHELL EDGE TO SOLID ID							
\$#	cīd			_				title
	1Tie	d shell e	edges-to-s	olids and b	beam ends-t	to-solids		
\$#	surfa	surfb	surfatyp	surfbtyp	saboxid	sbboxid	sapr	sbpr
	1	2	13	13	0	0	10	10
\$#	fs	fd	dc	vc	vdc	penchk	bt	dt
	0.0	0.0	0.0	0.0	0.0	° 0	0.01.	00000E20
\$#	sfsa	sfsb	sast	sbst	sfsat	sfsbt	fsf	vsf
	1.0	1.0	0.0	0.0	1.0	1.0	1.0	1.0

Comment: OPTION4 can be "blank", "OFFSET", "BEAM\_OFFSET" or "CONSTRAINED\_OFFSET". The OFFSET variants do only tie rotational DOF for solid elements with rotational DOF, i.e., ELFORMs 3, 4, 20 or 22, or when the solid part is rigid.



### Additional contact information

Other contact types are occasionally used in the examples, such as for instance FORMING contacts and \*FORCE\_TRANSDUCER. Settings used for these contacts can be found in the models.

Resultant contact interface force is output in rcforc, see \*DATABASE\_RCFORC.

SPR/MPR=1 is needed when a contact interface force file (intfor) is requested, see also \*DATABASE\_BINARY\_INTFOR. SPR/MPR=1 is also needed for output of nodal interface force in ncforc, see \*DATABASE\_NCFORC.

#### Some comments about penalty stiffness

The recommendation is to begin with default settings for penalty stiffness. The penalty stiffness can be altered, if needed, with several different scaling parameters. Sometimes a reduction is needed to assure numerical stability. In other cases, an increase is needed to reduce the size of the penetrations.

For SOFT=1 and SOFT=2 contacts, the (global) penalty stiffness for the contact at hand can be scaled using SFSA/SFSB directly on the contact definition. The base level of the penalty stiffness for this contact can be set using the DTSTIF parameter. If used, DTSTIF is often set to the initial time step dt of the simulation at hand. Furthermore, for individual parts the penalty stiffness can be scaled (locally) using SSF on \*PART\_CONTACT.

For SOFT=2 contact the edge-to-edge penalty stiffness can be scaled using EDGEK. The penalty stiffness for tetrahedron elements can be scaled using TETFAC. Also, if sliding is prevalent for SOFT=2, then DPRFAC=0.001 on Optional Card C is recommended. Note: for SOFT=2 when using the scaling parameters SFSA/SFSB then they should both be set, and to the same value.

For MORTAR there are some special parameters. The base level of penalty stiffness can be scaled with SFSA. For solids and tshells the PENMAX parameter also affects the base level of penalty stiffness. Though, PENMAX is typically used to set a consistent release depth for solid parts. If locally penetrations become too large then IGAP can be increased, see figure below.



For tied contacts using penalty formulation (Option 4: OFFSET/BEAM\_OFFSET) it has been observed that reducing the penalty stiffness, e.g. SFSA=SFSB=0.1, can increase the numerical stability. This type of stability issues may be observed through strange contact energies in sleout/glstat.

Remember, the solution timestep size will determine how often the contact situation is evaluated and will influence the penalty stiffness, robustness, and accuracy.

# **Element formulations**

The element formulation for a part (PID) is set on the \*SECTION card, e.g., the ELFORM on \*SECTION\_SHELL, \*SECTION\_SOLID or \*SECTION\_BEAM.

#### General

The application at hand, as well as previous experiences, often contribute to the choices that are made. In explicit simulation, linear element formulations are typically favoured due to the higher cost of using higher order elements. The first choice is often whether to use an under-integrated element formulation or a more expensive fully integrated variant. There are no strict rules on how to make this choice and it is often application based. Generally, under-integrated elements tend to be a bit more robust in terms of surviving harsh conditions but also a bit too soft compared to an equivalent fully integrated element. Given a good mesh quality and conditions that are not too harsh a fully integrated element is typically the more accurate element. Other circumstances that may indicate a favour for the fully integrated element may be the application of point loads or the build-up of excessive hourglass energy. For solids, the ability to capture bending modes accurately may also favour the fully integrated element. Keep in mind that the choice that is made is not only the element formulation, i.e., the material, the element formulation and the hourglass control should be seen as a unit.

For the under-integrated element, the choice of hourglass formulation and hourglass coefficient may not be trivial. Though, some guidance is given in the LS-DYNA manual. Often in automotive crash the hourglass coefficient for elform 2 shells (IHQ = 4, stiffness form) is decreased from the default 0.1 to 0.05 or 0.03. The hourglass coefficient for elform 1 solids may in some cases be adjusted as well, e.g., QM=1.0 for elastic materials.

A general recommendation is to set ESORT=1 on \*CONTROL\_SHELL and \*CONTROL\_SOLID so that degenerated quadrilateral shells are automatically switched to a more suitable triangular formulation and degenerated solids are automatically switched to a tetrahedron or pentahedron formulation. It is recommended to minimize the use of pentahedron elements when creating structured meshes.

Some valuable sources for information regarding elements in LS-DYNA are, other than the LS-DYNA manual, "Properties & Limits: Review of Shell Element Formulations" by André Haufe, Karl Schweizerhof and Paul DuBois, "Solid element formulations in LS-DYNA" by Christoph Schmied and "Review of Solid Element Formulations in LS-DYNA" by Christoph Schmied and "Review of Solid Element Formulations in LS-DYNA" by Tobias Erhart.

### **Elements by selection**

There are many element formulations in the LS-DYNA library, and it is not the intention to give an overview of all of them in this document. The table below is to highlight some elements that are often used in explicit simulations and that can serve as a good starting point in many cases.

Element type	Hourglass control	Comment
and formulation		
Shell, elform = 2	IHQ = 4 (or 2 for viscous)	Under-integrated, very effective
Shell, elform = -16	IHQ = 8	Fully integrated,
		enhanced version of elform 16
Solid, elform = 1	IHQ = 6 (or 3 for viscous)	Under-integrated, very effective
Solid, elform = -1	NA	Fully integrated
		more efficient version of elform -2
Solid, elform = 13	NA	1-point tetrahedron element,
		TETI3V = 1 on *CONTROL_SOLID
Solid, elform = -2	NA	Fully integrated,
		Like elform 2, but can handle poor
		aspect ratio better.
Solid, elform = 16	NA	4 or 5 point 10-noded tetrahedron
Beam, elform 1	NA	Hughes-Liu with cross section
		integration (plastic deformation)
Beam, elform 2	NA	Belytschko-Schwer resultant beam
		(elastic deformation)
Beam, elform 3	NA	truss (resultant)
Beam, elform 6	NA	Discrete beam/cable
		(spring/damper)
Beam, elform 9	NA	Spot weld beam

Worth mentioning is that there are cases where the tetrahedron elform 10 element may be preferred to elform 13, e.g., when the material model requires it (typically foams), when problem arises with negative volumes (severe deformations) or when two tetrahedron parts share nodes. Note: the more accurate type 13 solid implementation is activated on \*CONTROL\_SOLID with parameter TETI3V (recommended).

In addition, hourglass control IHQ=10 in combination with solid elform 1 or 16 activates the Cosserat Point Element, which has demonstrated good performance when modelling for large deformations, e.g., foams and hyperelastic materials such as rubber.

### Benchmark example: Crash box

The examples in this guide can serve as a platform for quick benchmarks or testing of different settings. In this section the "Crash box" (see Example 2 for more details) was used to do comparative testing of shell elform 2 and 16/-16. Five simulations were performed. As shown in the table below the runtime was about 2.4 times larger for elform 16/-16 compared to elform 2, which was expected. Adding hourglass control 8 to elform 16/-16 did, in this example, not increase the runtime significantly. As expected, a difference in response can be seen comparing elform 2 with elform 16/-16. In this example the results using 16/-16 were practically identical. The effect of adding hourglass form 8 was only of significance for the tube with the finest mesh (2.5 mm).

Simulation	Shell	Hourglass	Runtime	Runtime
	elform	control	(minutes)	(normalized)
1	2	4 (QM = 0.05)	43	1.00
2	16	None	101	2.32
3	16	8	103	2.37
4	-16	None	102	2.35
5	-16	8	105	2.42



Comment: Tube with mesh size 5.0 mm.



Comment: Tube with mesh size 2.5 mm.

# Example 1: Drop test

This is an example of a drop test simulation using Ansys LS-DYNA. A ceramic mug is dropped from a height of 1.5 m, allowing it to freefall until it hits a hard surface. The model exemplifies a parameter definition, an accelerometer, a strain gauge load cell and the contact treatment of eroding elements.



#### Simulation data

# nodes	184k
# elements	146k
Timestep size (ms)	0.1e-3 (constant)
Termination time (ms)	10.0
Solution time (minutes)	114

#### **Elements and material**

The mug and the surface are modelled with solid, hexahedral ELFORM -1, elements. Four elements through the thickness. The element size is 1 mm. Material type \*MAT\_110<sup>2</sup> is used for the mug. Elements erode when the negative pressure reaches 0.04 GPa. This can be confirmed by, for instance, plotting the pressure for solid EID 817 or 19649 (depending on which element erode) in ELOUT.

<sup>&</sup>lt;sup>2</sup> Parameters collected from the paper "Implementation and Validation of the Johnson-Holmquist Ceramic Material Model in LS-DYNA", by D. S. Cronin et al, and modified to fit the purpose of the example.



### **Contact definitions**

Contact type \*CONTACT\_ERODING\_SINGLE\_SURFACE with SOFT=2 is used for the model. An eroding contact is needed so that contact is treated correctly. This means an updated contact surface after elements have been eroded. An ERODING contact automatically invokes a negative volume failure criterion for all solid elements in the model. The use of PSFAIL on \*CONTROL\_SOLID limits this criterion to a partset, which is recommended. The handle is attached to the mug with a tied contact, i.e. \*CONTACT\_TIED\_NODES\_TO\_SURFACE.

#### **Control cards**

No mass-scaling or very little mass scaling is recommended when performing drop test simulations so that the kinetics is not significantly affected due to additional artificial added mass. In this case mass scaling is applied, see DT2MS on \*CONTROL\_TIMESTEP, it adds 4 extra grams to the mug. This is an increase of 2% of the total weight of the mug. This may be acceptable given that the time stepsize was increased by a factor 2 and therefore shortening the simulation time considerably. Since the parameter MSSCL is set to 1 on \*DATABASE\_EXTENT\_BINARY fringe plot of the added mass is possible.

#### Loads and constraints

Gravity is applied with \*LOAD\_BODY\_Z. The bottom of the impacted surface is fully constrained with SPC.

#### Initial conditions

The impact velocity is set with a combination of \*INITIAL\_VELOCITY, \*PARAMETER and \*PARAMETER\_EXPRESSION so that only the drop height is required as input. Thereby preparing the model for quick configuration when performing, for example, parameter studies.

#### Accelerometer

An accelerometer is installed on the handle to output accelerations in a local coordinate system in NODOUT (NID 185243). An additional node (NID 185244) on the same accelerometer outputs accelerations in the global coordinate system.



Comment: An accelerometer is mounted on the handle.



The global node can be used to check that the output frequency of the accelerometer is sufficiently high, which is by integrating each acceleration component (x, y, z) and compare the results to the corresponding measured velocities. These curves should match when offsets are removed, see figure below.



*Comment: The integrated accelerations and the velocities match (nodout, global node).* A Butterworth filter with a cut-off frequency 1000 Hz is applied.

#### Strain gauge load cells

Two strain gauge load cells (shell EID 200001 and EID 200002) are installed on the inner and outer surface of the mug, respectively. The strain gauge results, given in the local coordinate system of the shell element, can be plotted from ELOUT. A Butterworth filter with a cut-off frequency 3000 Hz may be appropriate. The results show (initially in the simulation) tension on the outer surface (positive x-strain) and compression on the inner surface (negative x-strain) as expected (looking at the deformation). Note that STRFLG=1 must be set on \*DATABASE\_EXTENT\_BINARY to get strain data in ELOUT and D3PLOT.



Comment: Strain gauge load cell (yellow shell element).

#### Results

Well, the purpose of a drop test is often to see if something breaks or not and the answer here is obvious. It can also be said that this crash event is rapid and brittle, which is shown by the acceleration curves. Evaluation of acceleration levels would therefore depend highly on the chosen filter and frequency.



Comment: Drop test at 10 milliseconds.



# Example 2: Crash box

This is an example of a crash box buckling analysis using Ansys LS-DYNA. Three identical, except for the mesh element size, crash boxes are impacted by rigid cylinders. Each cylinder has a weight of 700 kg. The impact speed is 30 km/h. This case exemplifies \*MAT\_024 with strain rate effects, history variables, contact force transducer and the concept of mesh convergence.



#### Simulation data

# nodes	81k
# elements	72k
Timestep size (ms)	0.3e-3 (constant)
Termination time (ms)	100.0
Solution time (minutes)	109

#### **Elements and material**

The three crash boxes (100x100x500 mm, thickness 2.0 mm) are modelled with shell elements ELFORM -16, which is a fully integrated element type. Hourglass control type 8 is applied. The mesh size is for the three boxes 10 mm, 5 mm and 2.5 mm, respectively. The material model used for the steel is \*MAT\_024 where strain rate effects are accounted for by adding a strain rate table definition. The table includes two hardening curves corresponding to static load (strain rate=0) and strain rate 1000/s. Intermediate strain rate values in LS-DYNA are interpolated between the curves. By setting VP=1 on the material card, the numerical noise caused by strain rate effects can be minimized.

#### **Contact definitions**

The contact, for the complete model, is treated by one \*CONTACT\_AUTOMATIC\_SINGLE\_SURFACE contact with SOFT=1. A key parameter when evaluating the performance of a crash box is the force vs.



displacement curve. By adding \*FORCE\_TRANSDUCER\_PENALTY, one for each impactor, the force between each impactor and its corresponding crash box can be measured.

#### **Control cards**

By setting NEIPS=1 on \*DATABASE\_EXTENT\_BINARY the effective plastic strain rate (VP=1) for \*MAT\_024 is output to the D3PLOT as history variable #1, which can be fringe plotted in a post-processor. In Ansys LS-PrePost this can be done under "FComp-Misc-History var#1". Note that with HISNOUT=1 on \*CONTROL\_OUTPUT the content of each history variable is printed in d3hsp, search for "history var i a bles listing".

#### Loads and constraints

The rear end of the crash box is fixed with \*BOUNDARY\_SPC\_SET.

#### **Initial conditions**

The initial velocity of the impactors is applied with \*INITIAL\_VELOCITY\_RIGID\_BODY.

#### Results

By specifying the same out frequency for RCFORC and NODOUT the force versus displacement curve can easily and accurately be plotted in a post-processor. The force curves, and stop displacements, for the tube with 5 mm mesh and 2.5 mm are similar whereas the 10 mm mesh produce a significantly stiffer response.



Comment: Force verses displacement curves for the three tubes.



# **Example 3: Deep drawing**

This is an example of a deep drawing simulation using Ansys LS-DYNA. A rectangular shape with curved top is formed with a single action die process. This example shows a typical setup for a deep drawing simulation with blank and tool definition, process boundary conditions, material model and some recommended settings for metal forming simulations.

The model uses PARAMETER and PARAMETER\_EXPRESSION to define the process. The user needs to define the parts IDs for the blank and tools, distances between the tools, blank thickness, tool velocities and binder force. The load curves for the motions of the tools and the end time are calculated from the user input.



#### Simulation data

# nodes	11k (initial)
# elements	11k (initial)
Timestep size (ms)	0.54e-3 (constant)
Termination time (ms)	8.8
Solution time (minutes)	4

#### **Elements and material**

The blank and tools are modelled with shell elements. The tools are rigid and meshed with deviation mesh mode which allows the elements to follow the correct tool geometry. The blank is meshed with an initial element length of 4 mm, but also has adaptive mesh applied which enables the elements to be refined in two steps. A stiffness based hourglass control type 4 is applied (QH=0.1). MAT\_BARLAT\_YLD2000 is used for the blank with material parameters defining the anisotropic behavior of the material. The hardening is defined with a load curve and the material direction is defined by the AOPT flag.



### **Contact definitions**

Contact type FORMING\_ONE\_WAY\_SURFACE\_TO\_SURFACE is used between the blank and all the tools. This contact type is recommended for sheet metal forming simulations with some parameters setting defined for this purpose, e.g. the tool thickness is not considered. Note that this a non-symmetric contact type and the blank should be set as tracked (SURFA) surface. The contact is also sensitive for orientation of the mesh, and the normal of the tool elements should be directed toward the blank.

#### **Control cards**

The mesh adaptivity of the blank is defined in CONTROL\_ADAPTIVE. Here the adaptivity frequency, tolerance, refinement levels etc. are defined. Also set ADPOPT to 1 for the blank in PART to activate the adaptivity for the part. In CONTROL\_CONTACT it is recommended to set PENOPT=4 for sheet metal forming simulations. The shell thickness change due to membrane stretching is one of the main results from this type of simulation so the ISTUPD in CONTROL\_SHELL must be set to 1 or 4.

Deep drawing simulations when the blank is clamped between the die and binder usually have low dynamic effects and can therefore be mass-scaled to quite a high ratio. The timestep used for this simulation is 6e-4 ms, see DT2MS in CONTROL\_TIMESTEP, which increases the blank mass by approximately 360 %.

#### Loads and constraints

The die is translated with a trapezoidal load curve defining the velocity with a ramping time of 1 ms. The die is translated with the forming velocity (BOUNDARY\_PRESCRIBED\_MOTION\_RIGID), and the binder has a load applied to it with LOAD\_RIGID\_BODY. The binder also has a vertical constraint (CONSTRAINED\_RIGID\_BODY\_STOPPERS) applied to it during the forming. The punch is fixed during the whole process.

#### Output

The ASCII files usually needed for a sheet metal forming simulation is bndout, glstat, matsum, rcforc and sleout. In the DATABASE\_EXTENT\_BINARY the STRFLG option must be set to 1 to get the strain data. The strain output is needed when doing post-processing with Forming Limit Diagram (FLD).

INTERFACE\_SPRINGBACK\_LSDYNA is used to extract a dynain file at the end of the simulation containing the thickness, strain, and stress in the blank.

#### Results

A search in the d3hsp and message files shows no errors or significant warnings. There are no hot spots of plastic strain in the blank that could indicate tool geometry errors. The contact forces show some oscillations due to the mesh adaptivity, but apart from that the curves looks smooth.





Comment: Thickness distribution of blank at end of forming.



Comment: Tool forces during forming.

#### Springback

A springback analysis gives the resultant deformation of the blank when it is removed from the forming tools. The analysis is performed using static implicit. The element formulation is switched from ELFORM 2 to ELFORM 16. The number of integration points (NIP) is increased from five to seven to improve the accuracy. Some constraint points need to be added to prevent the rigid body motions. When adaptivity is used for the blank it is convenient to use \*CONSTRAINED\_COORDINATE rather than SPCs as this keyword is not dependent on the node numbering.

# Example 4: Post-buckling strength

This is an example of a quasi-static buckling analysis using Ansys LS-DYNA. The target of the simulation is not only to find the ultimate strength of the profile but also to find the residual post-buckling strength. The model demonstrates a testing rig with an applied uniform pressure load, the use of a PI-regulator as well as the inclusion of geometry perturbations. The use of selective mass-scaling is briefly discussed.

![](_page_28_Picture_2.jpeg)

### Simulation data

# nodes	136k
# elements	136k
Timestep size (ms)	0.664e-3 (constant)
Termination time (ms)	150.0
Solution time (minutes)	176 <sup>3</sup>

### **Elements and material**

The steel profile (750x45x2000 mm, thickness 0.5 mm) is modelled with shell elements ELFORM -16, which is a fully integrated element type. Hourglass control type 8 is applied. A uniform mesh with element size 4-5 mm is used. Element thickness perturbation (±0.003 mm) is introduced with \*PERTURBATION\_SHELL\_THICKNESS. The material model used for the steel is \*MAT\_024. Strain-rate effects are not considered since the simulation is to be quasi-static. It is important to have a realistic description of the hardening curve, also in the necking region, when simulating post-buckling behaviour.

<sup>&</sup>lt;sup>3</sup> The solution time may be reduced by about 40% with Selective Mass Scaling (SMS). The main file run.key is prepared for such a test, see lines that are commented out.

### **Contact definitions**

The contact between the profile and the supports is handled by an \*CONTACT\_AUTOMATIC\_SINGLE\_SURFACE contact with SOFT=1.

#### **Control cards**

By adding \*DATABASE\_CURVOUT, the curve defined by \*DEFINE\_CURVE\_FUNCTION is output. The curve is in this case the steer signal for the pressure as applied by the PI-controller.

#### Loads and constraints

Boundary symmetry conditions are set in the transverse direction by \*BOUNDARY\_SPC\_SET. Gravity is accounted for by \*LOAD\_BODY\_Z. The profile is loaded with a uniform pressure. The pressure can be applied in different ways where a relatively slow ramp load is the most straight forward approach. This works fine up till ultimate strength is reached. After that the kinetic energy will increase rapidly as the profile collapses due to buckling. A ramp load cannot be used to evaluate the post buckling strength of this profile. For this reason, a PI-controller is used as defined by \*DEFINE\_CURVE\_FUNCTION with function PIDCTL. The controller is set to keep a steady load velocity of 0.4 m/s by controlling the applied pressure. As with all PI controllers adjusting the parameters can be a bit tricky. In this case the initial settings of parameters, i.e., the proportional gain kp and the integral gain ki, gave a fast prediction of ultimate load as well as post-buckling strength. However, the elastic response before buckling was oscillating too much. This was improved by decreasing kp and increasing ki.

#### Results

By scaling CURVOUT by a factor of 1.0e-2 the applied pressure (MPa) can be plotted. BNDOUT gives the reaction forces in the supports. In addition, the z-displacement can be plotted from NODOUT. The node at hand, NID 108355, is used by the PI-regulator to measure the velocity of the applied load. This node is the dependent node in a \*CONSTRAINED\_INTERPOLATION definition and follows the motion of the profile. By this arrangement, a more robust set point (in this case the velocity) can be monitored.

The kinetic energy is held to a low level throughout the simulation. The peak kinetic energy was about 7% of the total energy just after buckling and then rapidly decreased to a level <1%. The figure below shows the kinetic energy in the simulations with a ramp load compared to a PI-controller. It is evident that a ramp cannot be used to evaluate post-buckling strength. Note that in the figure below the time for the RAMP-curve has been scaled down a factor 10 for a clearer comparison.

![](_page_29_Picture_9.jpeg)

![](_page_30_Figure_0.jpeg)

Comment: Kinetic energy with a ramp load (red curve) and with a PI-controller (green curve).

It is possible to decrease the simulation time about 40% by using selective mass scaling. This can be an efficient way to speed up for instance an optimization loop. Though, care must be taken so that the mass scaling does not significantly affect the simulation results. To activate selective mass scaling set IMSCL=1, or preferably refer to a \*SET\_PART, on \*CONTROL\_TIMESTEP. DT2MS=-1.4756e-3 was found to give a good speed-up without compromising the results too much (x2 the timestep size).

![](_page_30_Figure_3.jpeg)

Comment: Pressure vs. displacement. Conventional mass scaling (red curve) and selective mass scaling (green curve).

![](_page_30_Picture_5.jpeg)

# **Example 5: Bolted connection**

Bolted connections can be modelled in many ways depending on the purpose. This connection include friction between bolt head/washer and underlying surface, it allows slip due to play between the bolt shank and the hole edge, it allows tear-out of the hole, it allows the bolt to deform/bend to some degree, it allows the bolt to break, and it includes pretension. The bolt shank is in this example modelled in two ways, which is with beam elements and with solid elements, respectively.

This example demonstrates, among others, the use of \*INITIAL\_AXIAL\_FORCE\_BEAM, \*INITIAL\_STRESS\_SECTION, \*CONTACT\_AUTOMATIC\_GENERAL and \*CONTROL-\_DYNAMIC\_RELAXATION.

![](_page_31_Picture_3.jpeg)

### Simulation data

# nodes	19k
# elements	18k
Timestep size (ms)	0.4e-3 (constant)
Termination time (ms)	40.0
Solution time (minutes)	14

### **Elements and material**

The bolts in the first connection are modelled with beam elements (ELFORM 1) for the shank and shell elements ELFORM -16 for the bolt head/washer. The bolts in the second connection are modelled completely with solid elements ELFORM 1. Hourglass control type 6 and type 8 is applied to the solids and shells, respectively. The bolt shanks are discretized with 3 elements along the length. The same material model, \*MAT\_024 with a bilinear plastic hardening curve, is used for all bolts. The yield strength is 240 MPa and the ultimate strength is 400 MPa. The elements are set to fail at 20% effective plastic strain. By defaults the beam element is deleted when all 4 integration points fail. To make the comparison more interesting \*MAT\_ADD\_EROSION was added with NUMFIP=1.0. This way the beam element is deleted when the first integration point fails. As shown in the figure a simulated tension tests give differences in results, as expected due to the difference in element types.

![](_page_31_Picture_8.jpeg)

![](_page_32_Figure_0.jpeg)

Comment: Tension test results for the beam bolt and the solid bolt, respectively.

#### **Contact definitions**

\*CONTACT\_ERODING\_SINGLE\_SURFACE with SOFT=2 is used for shells and solids. \*CONTACT\_AUTOMATIC\_GENERAL is used for the beam bolt shank and for the null beams that are added for the hole edges.

#### **Control cards**

DRCPSID on \*CONTROL\_SHELL activates a drilling rotation constraint for shell elements. Without this the beams of the bolt shank would be unconstrained in torsion/drilling. On \*DATABASE\_EXTENT\_BINARY, BEAMIP=4 for beam integration point output.

#### **Prestress of bolts**

The bolts are prestressed during dynamic relaxation which is activated by SIDR=1 on the loading curves. \*INITIAL\_AXIAL\_FORCE\_BEAM and \*INITIAL\_STRESS\_SECTION with ramped load curves prestress the beams and solids, respectively. IZSHEAR=2 on \*INITIAL\_STRESS\_SECTION allows bending/shear stresses to develop during the prestress of the solid bolts. \*CONTROL\_DYNAMIC\_REALAXTION, DRTERM=10.0 limits the relaxation phase to a maximum of 10 milliseconds. Since DRTOL is set to a very low number (1.e-10) 10.0 ms for the prestress is expected. The prestress process can be studied in detail through D3DRLF (see \*DATABASE\_BINARY\_D3DRLF) and the convergence from ASCII output file RELAX. The end of the stress initialization is confirmed in d3hsp. The resulting prestress load level can be checked in elout and/or secforc.

![](_page_32_Picture_8.jpeg)

#### Loads and constraints

The tubes are fully constrained at one end and pulled in the other end using rigid grips in combination with \*BOUNDARY\_PRESCRIBED\_MOTION\_RIGID.

#### Results

A fringe plot of the prestressed connections show that the resulting stress distributions are the same. Looking at stresses and axial force in ELOUT and SECFORC confirm that the desired prestress level (3 kN, 120 MPa) has been achieved.

![](_page_33_Picture_4.jpeg)

Comment: Prestressed connections.

The ultimate load of the beam element connection is about a factor 2 larger than for the solid bolt variant. For complex loading situations such as this calibration of the model at hand is needed. Especially in situations when a very detailed model is not convenient, which the modelling approach presented here is an example of.

![](_page_33_Figure_7.jpeg)

Comment: Ultimate load (x-force) for the loading case where the connections are pulled apart. Beam element variant in red and solid element variant in green. BW 3000Hz filter applied.

**Ansys** 

# **Example 6: Interference fit**

This is an example of an interference fit analysis in Ansys LS-DYNA. The model consists of the inner shaft and the outer hub. The hub is thought to have been heated and is set to be cooled 250°C in the simulation, thereby introducing the interference pressure. The applied temperature corresponds to an allowance of about 0.1 mm (0.25% of the diameter). Firstly, the shaft is loaded with a prescribed rotation and, secondly, the shaft is pulled thus testing the performance of the friction fit for torque and axial load. In this case the simulation results can be compared to an analytical solution.

This example demonstrates, among others, the use of \*LOAD\_THERMAL\_VARIABLE, \*CONTACT\_AUTOMATIC\_SURFACE\_TO\_SURFACE\_SMOOTH and the INTFOR file.

![](_page_34_Picture_3.jpeg)

#### Simulation data

# nodes	7k
# elements	17k
Timestep size (ms)	0.3e-3 (constant)
Termination time (ms)	100.0
Solution time (minutes)	18

#### **Elements and material**

The shaft is modelled with ELFORM 1 hexahedron (Hourglass control type 6) and ELFORM 15 pentahedron elements (through automatic sorting). The hub is modelled with ELFORM 13 tetrahedron elements. Both the shaft and the hub are made of steel, \*MAT\_024, where the yield stress is set high to ensure only elastic deformation. The thermal property of the hub is applied with \*MAT\_ADD\_THERMAL\_EXPANSION.

![](_page_34_Picture_8.jpeg)

### **Contact definitions**

The contact between shaft and hub is treated by

\*CONTACT\_AUTOMATIC\_SURFACE\_TO\_SURFACE\_SMOOTH. The SMOOTH option is necessary here to transfer the torque correctly, i.e., the faceted surface that would be the case for a non-SMOOTH contact would yield an incorrect torque. The interface force file is requested by setting SAPR=SBPR=1 on contact card 1 in combination with using S=iff on the execution command line. The output frequency is set with \*DATABASE\_BINARY\_INTFOR. The static and dynamic friction coefficient is 0.1 in the model.

### **Control cards**

\*CONTROL\_DYNAMIC\_RELAXATION is added for the temperature load, which is applied during the DRphase.

### Loads and constraints

The outer surface of the hub is fixed through \*CONSTRAINED\_INTERPOLATION, which allows only radial movement. The beam element is added for the purpose of fixing the dependent node. This boundary condition is chosen to make comparing of the results with a known analytical solution applicable. The loading, which are x-moment and x-force, is applied to the rigid end of the shaft with \*BOUNDARY\_PRESCRIBED\_MOTION\_RIGID.

#### Results

The main results are the interference pressure, which can be fringe plotted from the interface force file, and the axial friction force and transmission torque. The latter two can be plotted from BNDOUT. The measured allowance, which is after compensation for the contact penetration, is about 0.088 mm. Analytically this would yield an interference pressure of 194 MPa. The maximum/ultimate analytical transmission torque is 2438 Nm. The analytical transmission axial force is 121.9 kN. The simulation is within 5% of these values. The analytical estimate is sensitive to the input allowance. Hence, the accuracy in the comparison is estimated to be within a few percent.

![](_page_35_Picture_9.jpeg)

Comment: Interference pressure. Red colour is range 190-200 MPa.

![](_page_36_Figure_0.jpeg)

Comment: Transmission torque (red curve) and transmission axial force (green curve). BW 1000Hz filter applied.

# **Example 7: Rubber seal**

This example exemplifies the modelling of a rubber seal. In this case a seal between the door frame and the side window of a car. The purpose of this 3D quasi-static simulation is to test the ability of the seal to prevent water and air leakage. The model consists of the deformable rubber seal as well as a rigid door frame and a window. This example demonstrates, among others, the use of \*MAT\_077 (\*MAT\_HYPERELASTIC\_RUBBER) in combination with a MORTAR-contact to handle the complicated contact situation.

![](_page_37_Picture_2.jpeg)

#### Simulation data

# nodes	30k
# elements	23k
Timestep size (ms)	0.311e-3 (constant)
Termination time (ms)	540.0
Solution time (minutes)	171

#### **Elements and material**

The rubber is modelled with the solid hexahedron (mostly) ELFORM 1 element in combination with hourglass formulation IHQ=6 and QM=1.0. The material model is \*MAT\_077 with a poison's ratio of 0.495. The hyperelastic constants (C10, C01) are set directly. Frequency independent damping is accounted for by G and SIGF. It was needed to decrease TSSFAC from default 0.9 to 0.7 to achieve numerical stability of the material.

![](_page_37_Picture_7.jpeg)

![](_page_38_Figure_0.jpeg)

Comment: The stress-strain curve was produced through axial tension and compression of a oneelement model.

![](_page_38_Figure_2.jpeg)

Comment: The damping curve was produced by simulating the damped oscillations of a solid element beam.

#### **Contact definitions**

The contact is treated by a \*CONTACT\_AUTOMATIC\_SINGLE\_SURFACE\_MORTAR contact. The only parameters that are changed from defaults are contact damping, VDC=20, and friction behaviour. A \*DEFINE\_FRICTION is referenced by setting FS=-2.

![](_page_38_Picture_6.jpeg)

### **Control cards**

ISYM on \*CONTROL\_CONTACT references to a node set on the symmetry plane. The MORTAR-contact picks this up and uses it in the contact algorithm.

#### Loads and constraints

The movements of the rigid parts, which are defined by smooth velocity curves, are controlled by \*BOUNDARY\_PRESCRIBED\_MOTION\_RIGID. The symmetry constraint (in z-direction) is defined with \*BOUNDARY\_SPC\_SET.

#### Output

\*CONTROL\_OUTPUT, PENOUT=1 for output of contact (absolute) penetrations in D3PLOT. This feature is handy when evaluating if contact is established or not. The maximum (absolute) penetration is also output to SLEOUT.

#### Results

The contact performs well, which is crucial when evaluating potential leakage. It needs to be accurate.

![](_page_39_Figure_8.jpeg)

Comment: Fringe plot of contact penetration in the MORTAR-contact.

Note: For demonstrative purposes, this simulation is performed in 3D. However, this example can with advantage be run in 2D. A test simulation in 2D (not documented in this document) was 10 times faster than the 3D simulation.

![](_page_39_Picture_11.jpeg)

# **Example 8: Vehicle initialization for crash**

This example shows a procedure to initialize a vehicle for crash analysis. The aim is to first set the vehicle in balance with gravity and then to apply initial velocity, which also includes rotational velocity of the wheels. This example demonstrates, among others, the use of \*INITIAL\_VELOCITY\_GENERATION for a situation that includes \*PART\_INERTIA, and \*INITIAL\_VELOCITY\_GENERATION\_START\_TIME.

![](_page_40_Picture_2.jpeg)

### Simulation data

# nodes	20k
# elements	18k
Timestep size (ms)	1.0e-3 (constant)
Termination time (ms)	200.0
Solution time (minutes)	4

### **Barrier model**

The original barrier model was developed by Livermore Software Technology Corporation (LSTC)<sup>4</sup>. The model has been modified by the author to fit the purpose of this example e.g., the bumper parts have been removed and the initial velocity cards have been updated. Worth mentioning is that the mass and inertia properties of the carriage is defined through \*PART\_INERTIA and that the wheels are inflated by airbags with \*AIRBAG\_SIMPLE\_PRESSURE\_VOLUME. The barrier has been positioned, in the vertical direction, where it should be when static equilibrium is attained. This means that the tires are initially sunk below ground level, see the figure below. Note that for a general suspension model the springs are assumed to be prestressed, e.g., with initial offset, according to the vehicle weight acting upon them.

<sup>&</sup>lt;sup>4</sup> D. Bhalsod, R. Chivukula, H. Devaraj; "214 Shell Barrier", LSTC, 2018.

![](_page_41_Picture_0.jpeg)

Comment: The initial position of the vehicle/tire showing ground level and floor plate.

#### Vehicle initialization procedure

The procedure used by the author is as follows:

- 1. The tires are inflated during a span of 5 milliseconds.
- 2. A rigid floor is pushing up the tires to ground level between time 5 ms and 10 ms. The vertical motion of the floor is prescribed with \*BOUNDARY\_PRESCRIBED\_MOTION\_RIGID. Gravity, see \*LOAD\_BODY\_Z, is ramped up during the same time-period. In addition, mass weighted damping is applied to the tires. The damping is applied until time 20 ms and then ramped down to zero. This means no damping after time 20 ms. Note that no damping is applied by the airbag model itself, see parameter MWD in \*AIRBAG\_SIMPLE\_PRESSURE\_VOLUME. The reason for this is that such airbag damping would prevent the wheels from rotating correctly.
- 3. The contact between the floor plate and the tire is then removed and replaced by \*RIGIDWALL\_PLANAR. This step is not necessary if using the floor plate through the entire analysis.
- 4. Initial velocity is applied at time 30 ms with \*INITIAL\_VELOCITY\_GENERATION and \*INITIAL\_VELOCITY\_GENERATION\_START\_TIME. The procedure of point 1 to 3 may in some cases be completed faster, and if so the start time for the initial velocity can be set earlier. The \*INITIAL\_VELOCITY\_GENERATION card is defined twice, which means once for PHASE=0 and once again for PHASE=1. Finally, IRIGID=1 for the carriage to overrule initial velocity set on \*PART\_INERTIA.

An alternative to this method would be to use \*CONTROL\_DYNAMIC\_RELAXATION. This would be easier to set up and it would probably work, but it would be more time consuming (i.e. longer simulation time) compared to this relatively fast procedure. This is even more true in the case of a more complicated car. A second alternative would be to do it with the implicit solver, but it would potentially be time consuming to do the necessary changes to make the model run efficiently also using the implicit solver.

![](_page_41_Picture_9.jpeg)

#### Preparations

The preparations needed when setting up a model as shown in this procedure is a) to determine the damping coefficient of \*DAMPING\_PART\_MASS\_SET for the tires and b) to determine the vertical position of the barrier when in static equilibrium, and thereby determine the z-direction motion of the rigid plate to reach ground level.

Point a is done by inflating a free wheel and measure the period of the tire oscillations. Point b can be determined by leaving the vehicle on ground under the influence of gravity and measure the motion of the vehicle in z-direction. Note that in this model there is no wheel suspension. If wheel suspension is present, then the suspension springs must be prestressed accordingly. This means given that the vehicle is modelled as if static equilibrium is fulfilled. The vertical motion of the rigid floor described earlier is mainly to initiate the deformation of the tires, i.e., not to prestress the suspension springs.

#### **Contact definitions**

The contact between tires and ground is treated by \*CONTACT\_AUTOMATIC\_SURFACE\_TO\_SURFACE with SOFT=1. The contact is later replaced by \*RIGIDWALL\_PLANAR using the BIRTH/DEATH functionality.

#### **Control cards**

No special cards added.

#### Loads and constraints

\*LOAD\_BODY\_Z, \*BOUNDARY\_PRESCRIBED\_MOTION\_RIGID and \*INITIAL\_VELOCITY\_GENERATION as described earlier. The use of \*PARAMATER and \*PARAMETER\_EXPRESSION makes it easy to change the velocity.

#### Output

\*DATABASE\_ABSTAT has been added for monitoring tire pressure.

![](_page_42_Picture_11.jpeg)

### Simulation check

A "steady state" analyses shows that the initiated vehicle/carriage drops <1 mm under the influence of gravity, which proof that the vehicle initialization procedure has been successful. The initial velocity at time 30 ms is confirmed by a vector plot of the velocity from D3PLOT. Checking the kinetic energy in GLSTAT is also a good measure.

![](_page_43_Figure_2.jpeg)

Comment: A "steady state" analysis (no velocity applied) to check static equilibrium. A switch of any contact or rigid wall will introduce noise, see spike in green curve.

![](_page_43_Figure_4.jpeg)

Comment: Initial velocity applied.

![](_page_43_Picture_6.jpeg)

# Example 9: Short Fiber reinforced plastic bracket

This is an example of a Short Fiber Reinforced Plastic (SFRP) bracket. The mechanical properties of SFRP are greatly dependent on the fiber orientations with respect to the loading direction. Within a component, the orientation of fibers differs due to the manufacturing process. The manufacturing process, often Injection Molding, of the component at hand must therefore be considered when simulating short fiber thermoplastics. A process of how to find out the fiber directions and make a LS-DYNA material card is described here. For further information about the simulation of plastics, please contact your local distributor of Ansys LS-DYNA.

![](_page_44_Picture_2.jpeg)

### Simulation data

# nodes	80k
# elements	54k
Timestep size (s)	0.2e-6 (constant)
Termination time (s)	0.02
Solution time (minutes)	34

Note that the unit system in this example is mm, s, tonne, N, MPa.

### **Elements and material**

The plastic bracket is modelled with 10-noded tetrahedron elements type 16. The number of integration points is set by NIPTETS=4 on CONTROL\_SOLID. Three suitable LS-DYNA material model alternatives for SFRP are the composite model \*MAT\_ANISOTROPIC\_ELASTIC\_PLASTIC (\*MAT\_157), the micromechanical model \*MAT\_4A\_MICROMEC (MAT\_215) and the machine learning-based multiscale material model \*MAT\_DMN\_COMPOSITE\_FRC. In this example \*MAT\_157 is chosen. A brittle failure can be considered in \*MAT\_157 by either the Tsai-Wu or the Tsai-Hill criterion, although it is not included in this example.

![](_page_44_Picture_8.jpeg)

#### Procedure

The procedure to create the input data for \*MAT\_157 to LS-DYNA is in short:

- 1. Perform an Injection Molding simulation of the component.
- 2. Map the results to LS-DYNA format (\*INITIAL\_STRESS\_SOLID).
- 3. Characterize the hardening properties of the material (LCSS).

#### Fiber orientation and anisotropic stiffness

To consider the fiber orientations in the LS-DYNA model, one must first perform the Injection Molding simulation. There are different software on the market that can be used for this purpose. In this example, the Ansys Technology Partner Moldex3D, by CoreTech System Co., Ltd, was used. The setup was straightforward since the material, including fiber content, was chosen from a database of material data from many suppliers. The component was read in as a 3D-CAD, in this case a STEP-file. A gate and runner were defined where the material was injected. The software suggested a mesh size, that was generated automatically. Process parameters such as flow rate profile and temperatures can be adjusted before starting the filling analysis.

The results from the Injection Molding analysis must then be mapped to a format that can be read by LS-DYNA and the mesh one intends to use for the structural evaluation, which probably differs from the mesh used for the Injection Molding analysis. There is a limited mapping functionality within Moldex3D. In this case, Envyo, by DYNAmore GmbH an Ansys Company, was used for the mapping. The results from the filling simulation were exported, including weld lines information and fiber orientations. This means that a total of three files were exported, i.e. nodes and elements of the injection molding mesh, fiber orientations and location of weld lines. There is a functionality in Envyo to make separate parts on areas around weld lines, where the material properties typically can be set lower if desired.

In Envyo, the target material model was set to \*MAT\_157. The parameter IHIS was set to 3, which means that both material directions and an anisotropic elastic stiffness matrix is output to \*INITIAL\_STRESS\_SOLID for each integration point of the element. The elastic properties of the polymer matrix and fibers are in the mapping combined to give the complete elastic stiffness matrix in what is referred to as a homogenization process. To perform this homogenization with Envyo, the elastic material properties, density etc, of the polymer matrix and the individual fibers are part of the input to Envyo. Also note that the local element coordinate systems, in which the fiber orientations are given, are calculated differently depending on the value of the INN parameter on \*CONTROL\_ACCURACY. The chosen setting in the simulation model, in this case INN=1, must therefore also be set in Envyo.

Currently, the mapped material directions cannot be shown directly from the keyword file in LS-PrePost. The model must first be initialized by LS-DYNA and then the first vector (the main direction) can be displayed from the d3plot results. For LS-PrePost to know which value of the INN parameter that has been used, one must also open the keyword file. After that, the visualization is done through "Post" - "Vector Plot". For solids choose "Hist. Var. cosine" and set "H.VAR X" = -4, "H.VAR Y" = -5 and "H.VAR Z" = - 6, which is where you find the appropriate history variables for \*MAT\_157. The corresponding settings for shells are "H.VAR X" = -1, "H.VAR Y" = -2.

![](_page_45_Picture_10.jpeg)

Note: Even though the elastic stiffness tensors are input on an integration point level, which overwrites the defined stiffness tensor on the material card, a reasonable value of a typical stiffness must be given on the material card. LS-DYNA uses this information to e.g. calculate an initial time step size.

![](_page_46_Picture_1.jpeg)

Comment: The weld line is given a separate PID by Envyo (green in figure).

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S#         SIGXX         SIGYY         SIGZZ         SIGXY         SIGY         SIGZ         SIGY         SIGY         SIGZ         SIGY         <		1	1	27	1				
\$#         SIG2X         EPS         HISV1         HISV2         HISV2           \$#         HISV4         HISV5         HISV6         HISV7         HISV2           \$#         HISV9         HISV10         HISV11         HISV12         HISV1           \$#         HISV9         HISV10         HISV11         HISV12         HISV1           \$\$         HISV14         HISV15         HISV16         HISV17         HISV1           \$\$         HISV19         HISV20         HISV21         HISV2         HISV2           \$\$         HISV19         HISV20         HISV21         HISV2         HISV2           \$\$         HISV24         HISV25         HISV26         HISV27         HISV2           \$\$         HISV24         HISV25         HISV26         HISV27         HISV2	Ş#	SI	IGXX	SIGYY		SIGZZ	SIG	XY	SIGYZ
\$#         HISV4         HISV5         HISV6         HISV7         HISV7           \$#         HISV9         HISV10         HISV11         HISV12         HISV1           \$#         HISV14         HISV10         HISV11         HISV12         HISV1           \$\$         HISV14         HISV15         HISV16         HISV17         HISV1           \$\$         HISV19         HISV20         HISV21         HISV22         HISV2           \$\$         HISV19         HISV20         HISV21         HISV22         HISV2           \$\$         HISV24         HISV25         HISV26         HISV27         HISV2           \$\$         HISV24         HISV25         HISV26         HISV27         HISV27           \$\$         HISV26         HISV27         HISV27         HISV27         HISV27	\$#	SI	IGZX	EPS		HISV1	HIS	V2	HISV3
S#         HISV4         HISV5         HISV6         HISV7         HISV7           q4         q5         q6         c11         c1           s#         HISV9         HISV10         HISV11         HISV12         HISV1           c13         c14         c15         c16         c2           s#         HISV14         HISV15         HISV16         HISV17         HISV1           c23         c24         c25         c26         c33           s#         HISV19         HISV20         HISV21         HISV22         HISV2           c34         c35         c36         c44         c4           s#         HISV24         HISV25         HISV26         HISV27         HISV27           c46         c55         c56         c56         c56         c56						ql		q2	q3
q4         q5         q6         cl1         cl2         cl2 <thcl2< th=""> <thcl3< th=""> <thcl3< th=""></thcl3<></thcl3<></thcl2<>	\$#	HI	tsv4	HISV5		HISV6	HIS	V7	HISV8
S#         HISV10         HISV11         HISV12         HISV12         HISV12           c13         c14         c15         c16         c2           \$#         HISV14         HISV15         HISV16         HISV17         HISV1           c23         c24         c25         c26         c3           \$#         HISV19         HISV20         HISV21         HISV2           c34         c35         c36         c44         c4           \$#         HISV24         HISV25         HISV26         HISV27         HISV2           c46         c55         c56         c66         c46         c55         c56			q4	q5		q6	c	11	c12
c13         c14         c15         c16         c2           \$#         HISV14         HISV15         HISV16         HISV17         HISV1           c23         c24         c25         c26         c3           \$#         HISV19         HISV20         HISV21         HISV22         HISV2           c34         c35         c36         c44         c4           \$#         HISV24         HISV25         HISV26         HISV27         HISV27           c46         c55         c56         c66         c66         c56         c66	\$#	HI	LSV9	HISV10		HISV11	HISV	12	HISV13
S#         HISV14         HISV15         HISV16         HISV17         HISV17           c23         c24         c25         c26         c3           \$#         HISV19         HISV20         HISV21         HISV22         HISV2           c34         c35         c36         c44         c4           \$#         HISV24         HISV25         HISV26         HISV27           c46         c55         c56         c66			c13	c14		c15	c	16	c22
c23         c24         c25         c26         c3           \$#         HISV19         HISV20         HISV21         HISV22         HISV2           c34         c35         c36         c44         c4           \$#         HISV24         HISV25         HISV26         HISV27         HISV27           c46         c55         c56         c66         c66         c66	\$#	HIS	SV14	HISV15		HISV16	HISV	17	HISV18
S#         HISV19         HISV20         HISV21         HISV22         HISV2           c34         c35         c36         c44         c4           s#         HISV24         HISV25         HISV26         HISV27         HISV27           c46         c55         c56         c66         c66         c66			c23	c24		c25	c	26	c33
\$# (C34 C35 C36 C44 C4 HISV24 HISV25 HISV26 HISV27 HISV2 C46 C55 C56 C66	\$#	HIS	5V19	HISV20		HISV21	HISV	22	HISV23
\$# HISV24 HISV25 HISV26 HISV27 HISV2 c46 c55 c56 c66			c34	c35		c36	c	44	c45
C46 C55 C56 C66	Şŧ	( HIS	5V24	HISV25		HISV26	HISV	21	HISV28
			C46	c55		c56	c	66	

Comment: Orientation vectors (red) and stiffness matrix (blue) for IHIS=3.

![](_page_47_Figure_0.jpeg)

Comment: First vector of material direction plotted in Ansys LS-PrePost.

### Material characterization

The hardening parameters on the material card are based on physical tension tests of specimens, which means 0° specimen (fibers aligned with specimen), 45° specimen and 90° specimen (fibers oriented transverse specimen). A CT-scan can be used to measure the detailed distribution of fibers to form an average orientation degree. The testing can also be done at different strain rates to calibrate strain rate dependency, but only static data is used in this example. If CT-scan results of the fiber orientation degree are to be used in the calibration, they need to be either mapped from threedimensional scan data using a mapping tool, such as Envyo, or assumed as non-varying through the thickness of the specimen. Both options require a homogenization step of the elastic properties though. Fiber orientation distribution through the thickness can also be accomplished by performing an Injection Molding simulation of the specimen. For the homogenization in this example, in which the fiber orientations were based on measurements, the Homogenization Tool in Ansa by BETA CAE, another Ansys Technology Partner, was used. Apart from the measured (and assumed constant over the specimen) orientation degree, remaining input parameters were the same as mentioned earlier for Envyo, i.e. the elastic properties of the individual fibers and the polymer matrix and density. Ansa outputs a homogenized stiffness tensor, which can be used directly in \*MAT\_157 when performing the tensile test simulations. The material hardening curve LCSS, which includes contributions from both matrix and fibers, can be calibrated against results from tensile tests in three directions. Since the material model is based on Hill's yield criteria, in the case of shells, the R-values may also be calibrated and for solids the parameters F/G/H/L/M/N may be calibrated. This part of the procedure can be viewed as a homogenization of the plastic properties. The software chosen for doing the optimization of the hardening curve was Ansys LS-OPT.

### **Control cards**

\*CONTROL\_ACCURACY, INN=1, which complies with the setting that was chosen in Envyo when doing the mapping. \*CONTROL\_OUTPUT, TETIOS8=1 so that all 10 nodes of the tetrahedral solids are output to d3plot. \*DATABASE\_EXTENT\_BINARY, NEIPH=6 for the possibility to plot the material directions in LS-PrePost.

![](_page_47_Picture_6.jpeg)

### Results

The produced input cards for LS-DYNA are shown below together with a fringe plot of Von Mises stress.

			*	MAT_ANISOTR	OPIC_ELASTIC_	PLASTIC_(TITL	E)(157) (1)	
	TITLE							
	SFRP							
1	MID	RO	<u>SIGY</u>	LCSS -	QR1	<u>CR1</u>	<u>QR2</u>	<u>CR2</u>
	1	1.500e-09	0.0	500000	0.0	0.0	0.0	0.0
2	<u>C11</u>	<u>C12</u>	<u>C13</u>	<u>C14</u>	<u>C15</u>	<u>C16</u>	<u>C22</u>	<u>C23</u>
	9847.0000	4830.0000	2611.0000	213.50000	0.7072000	-38.080002	6425.0000	1367.0000
3	<u>C24</u>	<u>C25</u>	<u>C26</u>	<u>C33</u>	<u>C34</u>	<u>C35</u>	<u>C36</u>	<u>C44</u>
	160.39999	-1.7980000	-27.820000	3973.0000	79.290001	-1.2020000	-15.780000	1537.0000
4	<u>C45</u>	<u>C46</u>	<u>C55</u>	<u>C56</u>	<u>C66</u>	<u>R00/F</u>	<u>R45/G</u>	<u>R90/H</u>
	2.0430000	-0.6930000	1794.0000	-14.320000	1663.0000	0.0	0.0	0.0
5	<u>S11/L</u>	<u>S22/M</u>	<u>S33/N</u>	<u>S12</u>	AOPT -	VP	UNUSED	MACE
	0.0	0.0	0.0	0.0	2.000000	0.0 👻	0.0	1 🔹
6	<u>XP</u>	YP	ZP	<u>A1</u>	<u>A2</u>	<u>A3</u>	UNUSED	EXTRA
	0.0	0.0	0.0	1.0000000	0.0	0.0	0.0	0.0
7	<u>V1</u>	<u>V2</u>	<u>V3</u>	<u>D1</u>	<u>D2</u>	<u>D3</u>	BETA	IHIS
	0.0	0.0	0.0	0.0	1.0000000	0.0	0.0	3.0000000

#### Comment: Material card (\*MAT\_157) for the SFRP.

9195	1	27	1		0 0	0	0
0.0		0.0		0.0	0.0	1	0.0
0.0		0.0	0.5	698915	-0.7453539	I –	-0.3459353
-0.086142		-0.47286	0.8	3769168	6649.625		2055.398
1870.337		0.0		0.0	0.0	1	4068.352
1430.028		0.0		0.0	0.0	1	2983.448
0.0		0.0		0.0	1651.795		0.0
0.0		1047.936		0.0	1473.1		0.0

Comment: \*INITIAL\_STRESS\_SOLID for EID 9195, as an example.

![](_page_48_Figure_6.jpeg)

Comment: Fringe plot of Von Mises stress.

![](_page_48_Picture_8.jpeg)

# Example 10: Plastic bracket

This is an example of a thermoplastic bracket. Thermoplastics can be modelled on different levels in LS-DYNA depending on the complexity needed for the simulation at hand. Typically, the start point is \*MAT\_024 where only tensile test data is needed to calibrate the material model and from there the complexity can be increased. In this example \*MAT\_SAMP\_LIGHT (\*MAT\_187L) is used, which can be considered a compromise choice between \*MAT\_024 and \*MAT\_SAMP-1 (\*MAT\_187) in terms of complexity and cost. For further information about simulation of plastics please contact your local distributor of Ansys LS-DYNA.

![](_page_49_Picture_2.jpeg)

### Simulation data

# nodes	80k
# elements	54k
Timestep size (ms)	0.2e-3 (constant)
Termination time (ms)	20.0
Solution time (minutes)	20

### **Elements and material**

The plastic bracket is modelled with 10-noded tetrahedron elements type 16. The number of integration points is set by NIPTETS=4 on CONTROL\_SOLID. Commonly used material models for plastics are \*MAT\_024, \*MAT\_124, \*MAT\_187 and \*MAT\_187L. \*MAT\_024 can account for strain rate dependent plastic hardening (viscoplasticity) with symmetry in terms of tension/compression. \*MAT\_124 is like \*MAT\_024 but also offers the choice to have asymmetric viscoplasticity in compression and tension. Both these material models use a constant plastic Poisson's ratio of 0.5, which is typically not true for plastics. \*MAT\_187 and \*MAT\_187L can also handle asymmetric viscoplasticity, but in this case with a constant or variable plastic Poisson's ratio to allow for the material to be compressible during plastic deformation. Both these models also include functionality for viscoelasticity. To utilize the full capacity of \*MAT\_187 test data for tension, compression, shear and biaxial tests is needed for the calibration. \*MAT\_187L is a simplified and faster version of \*MAT\_187 in terms of functionality and CPU cost. For instance, only test data for tension and compression is needed for the calibration. It is worth mentioning that \*MAT\_187

![](_page_49_Picture_7.jpeg)

does include options for damage and failure, which is not the case with \*MAT\_187L. However, it can be added through \*MAT\_ADD\_EROSION or \*MAT\_ADD\_DAMAGE. Due to differences between \*MAT\_187 and \*MAT\_187L material curves can typically not be directly transferred between the material models, which means a recalibration is needed.

The material card used for \*MAT\_187L is shown below. LCID-T/LCID are the plastic hardening curves in tension and compression, respectively. LCID-P is the plastic Poisson's ratio as function of plastic strain, and it covers both compression and tension.

	TITLE							
	Plastic							
1	MID	<u>RO</u>	=	=	EMOD	NUE		BETA
	1	1.000e-06	0.0	0.0	2.4000001	0.3000000	0	0.0
2		LCID C	CTFLG	RATEOP	NUEP	LCID-P	RFILTE	
	100	101	0 🗸	0	0.3000000	102	0.9500000	

Comment: The material card for \*MAT\_SAMP\_LIGHT used in this example.

![](_page_50_Figure_4.jpeg)

Comment: Poisson's ratio curve, LCID-P.

### \*MAT\_187L versus \*MAT\_024

The plastic Poisson's ratio is constant for \*MAT\_024, whereas it may vary for \*MAT\_187L if given by a load curve. This is a significant difference, which is illustrated by comparing the results from a tension test. When the test specimen reaches the necking point for \*MAT\_024 the necking is local, whereas for \*MAT\_187L the deformation is stable and extended over the gauge length. When local necking happens for \*MAT\_024 the load bearing capacity decreases rapidly, whereas \*MAT\_187L may continue to carry load. By using a steep hardening curve, the behavior of \*MAT\_024 can be made more like \*MAT\_187L. The steep curve is then, artificially, compensating for the constant plastic Poisson's ratio.

![](_page_50_Picture_8.jpeg)

![](_page_51_Picture_0.jpeg)

Comment: Fringe plot of Von Mises stress. Upper: \*MAT\_024, Lower: MAT\_187L.

#### **Control cards**

\*CONTROL\_OUTPUT, TETIOS8=1 so that all 10 nodes of the tetrahedral solids are output to d3plot. \*DATABASE\_EXTENT\_BINARY, NEIPH=7 for the possibility to plot additional data for \*MAT\_187L.

#### Results

A fringe plot of the Von Mises stress is shown below.

![](_page_51_Figure_6.jpeg)

Comment: Fringe plot of Von Mises stress.

![](_page_51_Picture_8.jpeg)

# **Record of revisions**

Rev. no	Release date	Author	Comment
1.0	2019-03-21	Klas Engstrand	Release of document.
		Anders Bernhardsson	
1.1	2019-10-21	Klas Engstrand	Example 8 added.
1.2	2021-08-20	Klas Engstrand	General update.
		David Aspenberg	Example 9 added.
			Example 10 added.
1.3	2023-02-01	Klas Engstrand	"Element formulations" added.
1.4	2023-02-15	Klas Engstrand	Tied contacts added
			in section "Contact cards"
1.5	2024-03-27	Klas Engstrand	General update.
		Jimmy Forsberg	"Database cards" added.

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![](_page_52_Picture_4.jpeg)

Keywords: Ansys LS-DYNA; explicit;

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![](_page_53_Picture_5.jpeg)

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